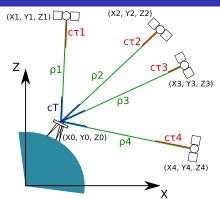
# Apparent Geocenter Oscillations in GNSS Solutions Caused by the Ionospheric Effect of Second Order

Kirill Palamartchouk

COMET, University of Oxford

15 September 2009

## GNSS trilateration: finding the receiver's coordinates



#### Where

- $\rho_i$  geometric distance between the satellite i and receiver
- $\tau_i$  clock error of the satellite *i*
- T − reciever clock error
- $R_i = \rho_i + c\tau_i + cT$  pseudorange

Known parameters:  $X_i, Y_i, Z_i, \tau_i$ 

Measured parameters:  $R_i$ 

Unknown parameters:  $T, X_0, Y_0, Z_0$ 

#### Realistic observation equation and error budget

$$\sqrt{(x_{i} - x_{0})^{2} + (y_{i} - y_{0})^{2} + (z_{i} - z_{0})^{2}} = c\tau_{i} - cT$$

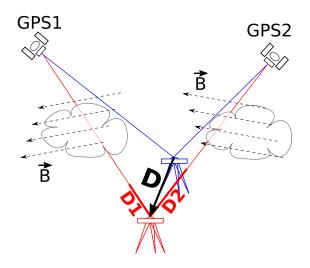
$$\delta_{i} + R_{i} + G_{i} + \epsilon_{i} + \Psi_{i} + R_{i} - I_{i} + S_{i} - \Phi - \Delta - M - \varepsilon - V - \Phi$$

#### Where

- Satellite-specific
  - $\tau_i$  Clock bias
  - $\delta_i$  Differential bias
  - $R_i$  Orbital error (2 cm-...)
  - G<sub>i</sub> Relativistic effect
  - $\epsilon_i$  Transmitter noise (1 cm)
  - Ψ<sub>i</sub> Antenna phase centre variation (0–20 cm)
- Propagation errors
  - *I<sub>i</sub>* Ionospheric refraction (up to 200 m)
  - $S_i$  Tropospheric refraction ( $\sim 2$  m)

- Receiver-specific
  - Φ Antenna phase centre variation (1–2 cm)
  - T − Clock bias
  - ullet  $\Delta$  Differential bias
  - M Multipath (up to 5 cm)
  - $\varepsilon$  Receiver noise (1 cm)
  - V − Phase windup (5 cm)

#### What happens if we fail to account for some contribution?



## Refraction index of plasma: Appleton-Hartree equation

$$n^2 = 1 - \frac{X}{1 - iZ - \frac{Y_{\rm T}^2}{2(1 - X - iZ)} \pm \sqrt{\frac{Y_{\rm T}^4}{4(1 - X - iZ)^2 + Y_{\rm L}^2}}}$$

where

$$X = \left(rac{f_{
m N}}{f}
ight)^2, \qquad Z = rac{
u}{2\pi f}$$
  $Y = rac{f_{
m H}}{f} = rac{eB_0}{2\pi f m_e}, \qquad Y_{
m T} = Y \sin heta, \qquad Y_{
m L} = Y \cos heta,$   $f_{
m N} = rac{1}{2\pi} \sqrt{rac{Ne^2}{m arepsilon_0}} \qquad f_{
m H} = rac{eB}{2\pi m_e}$ 

## Refraction index of plasma: Appleton-Hartree equation

$$n^{2} = 1 - \frac{X}{1 - iZ - \frac{Y_{T}^{2}}{2(1 - X - iZ)} \pm \sqrt{\frac{Y_{T}^{4}}{4(1 - X - iZ)^{2} + Y_{L}^{2}}}}$$

where

$$X = \left(rac{f_{
m N}}{f}
ight)^2, \qquad Z = rac{
u}{2\pi f}$$
  $Y = rac{f_{
m H}}{f} = rac{eB_0}{2\pi f m_e}, \qquad Y_{
m T} = Y \sin heta, \qquad Y_{
m L} = Y \cos heta,$   $f_{
m N} = rac{1}{2\pi} \sqrt{rac{Ne^2}{m arepsilon_0}} \qquad f_{
m H} = rac{eB}{2\pi m_e}$ 

# Total Electron Content (TEC) and path delay: the "textbook approach"

$$\Delta 
ho = \int (n-1) ds = \underbrace{rac{e^2}{8\pi^2 arepsilon_0 m_e}}_{40.308 \,\, \mathrm{m}^3 \mathrm{s}^{-2}} f^{-2} \int \mathit{Nds}$$

#### Path delay for L1

$$\Delta \rho = 0.162 \; \frac{\mathrm{m}}{\mathrm{TECU}}$$

#### Path delay for L2

$$\Delta \rho = 0.267 \; \frac{\mathrm{m}}{\mathrm{TECU}}$$

#### TEC from the measurements

$$TEC = \frac{1}{40.308} \frac{f_1^2 f_2^2}{f_1^2 - f_2^2} (R_2 - R_1)$$

 $TECU = TEC Unit = 10^{16} electrons/m^2$ .

#### Geometric linear combination, LC

$$LC = \frac{1}{f_1^2 - f_2^2} \left( f_1^2 L_1 - f_2^2 L_2 \right) \qquad PC = \frac{1}{f_1^2 - f_2^2} \left( f_1^2 P_1 - f_2^2 P_2 \right)$$
Geometric distance lonospheric path delay Noise

#### Properties of LC

LC is not influenced by the 'classical ionosphere'. Therefore, still unaccounted remain other propagation terms:

• tropospheric refraction;

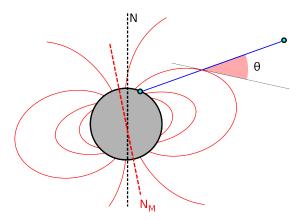
I C

• higher-order effects.

Noise amplitude of LC is greater by factor of 3, compared to the same of  $L_1$ .

#### Second order effect

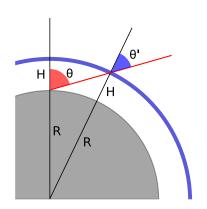
$$\Delta \rho = \int \frac{f_{\rm H} f_{\rm N}^2}{f^3} \cos \theta \, dl = \frac{7527c}{2} \int \frac{\textit{NB}_0}{f^3} \cos \theta \, dl$$

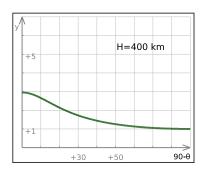


International Geomagnetic Reference Field (IGRF) model is used further.

## Thin layer geometry

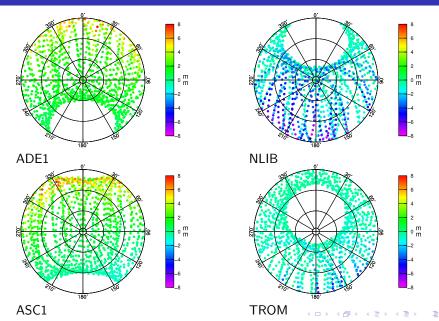
$$TEC = \frac{VTEC}{\cos \theta'}, \quad \sin \theta' = \frac{R}{R+H} \sin \theta$$



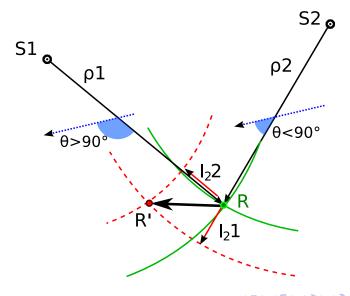


International Reference Ionosphere (IRI2007) model is used further.

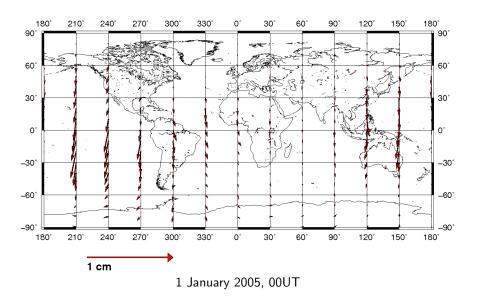
# Sky map of 2-nd order delay



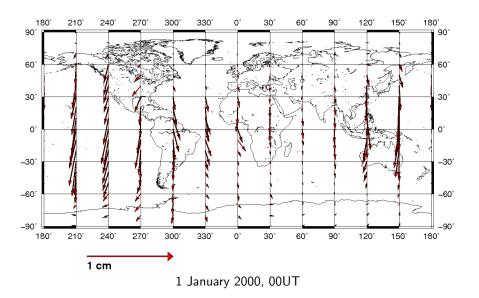
#### Geometry of the apparent displacement

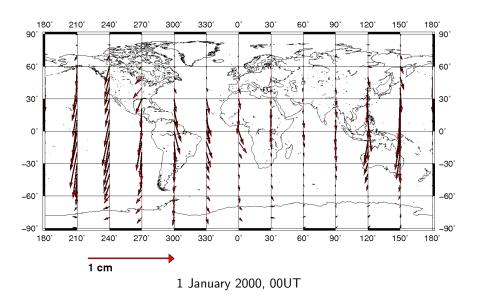


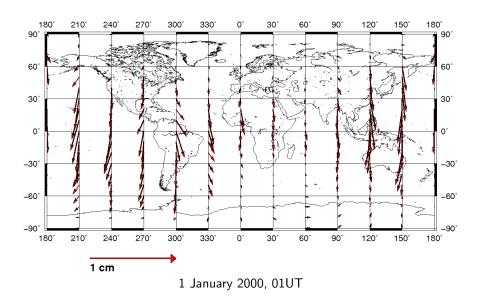
#### Apparent error: dependence on the Sun

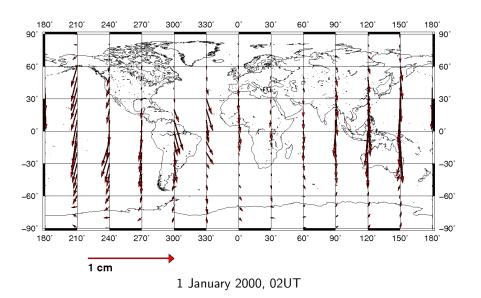


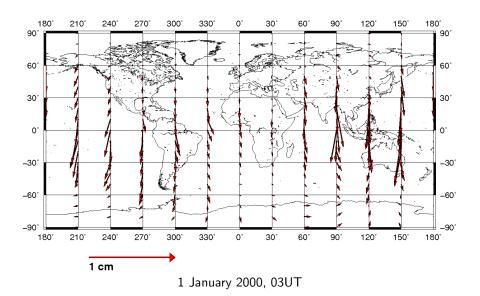
# Apparent error: dependence on the Sun

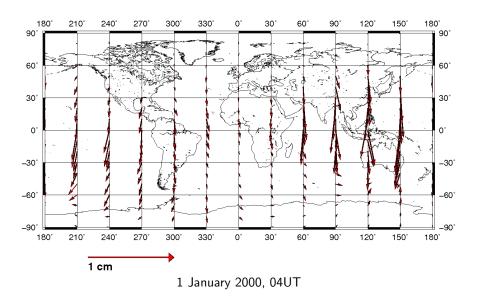


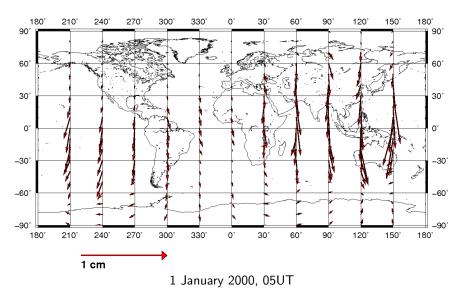


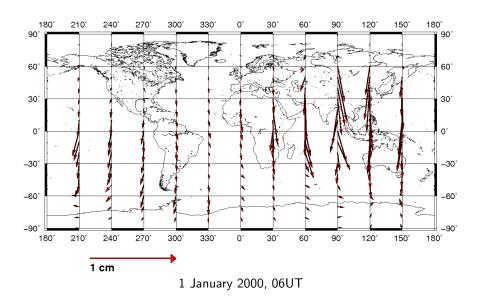


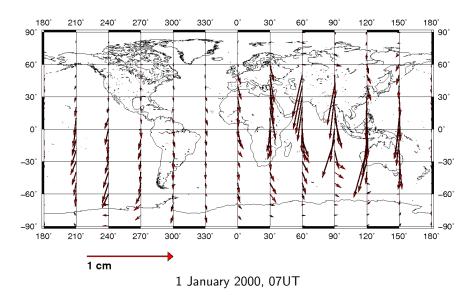


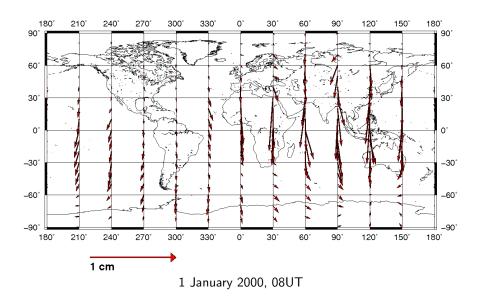


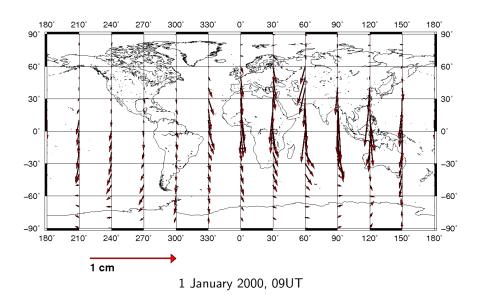


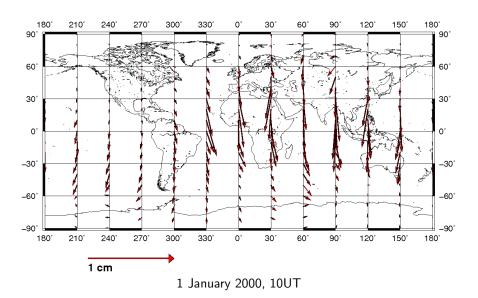


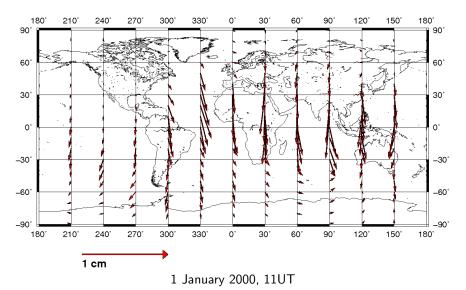


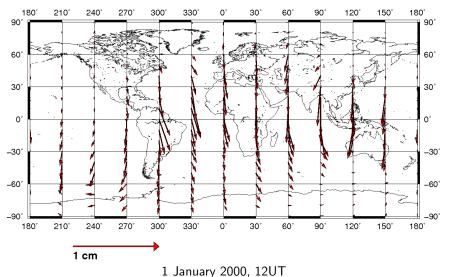


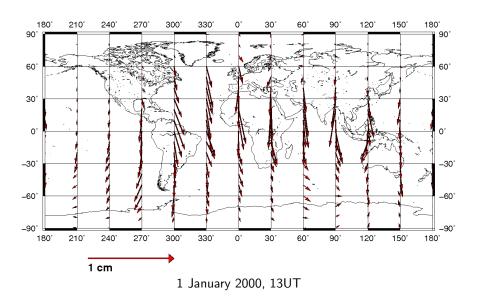


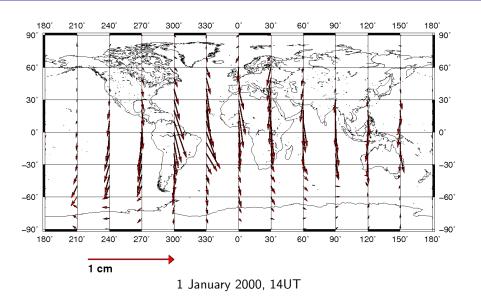


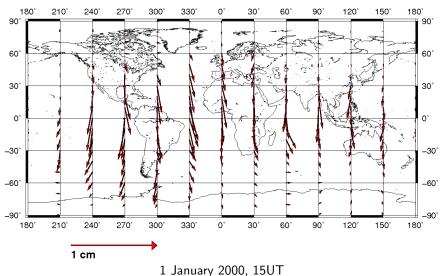




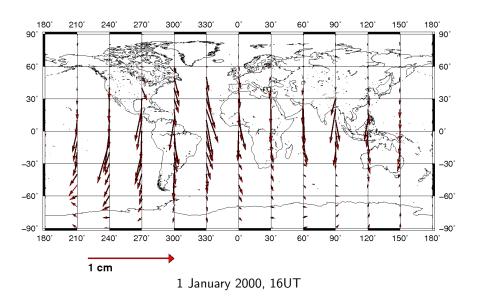


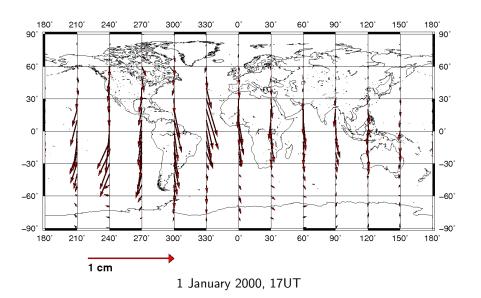


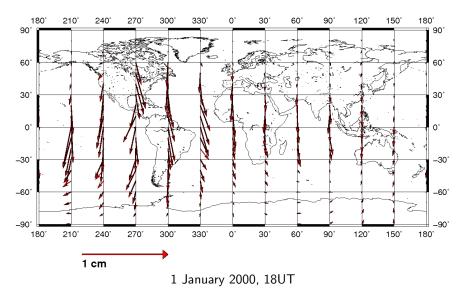


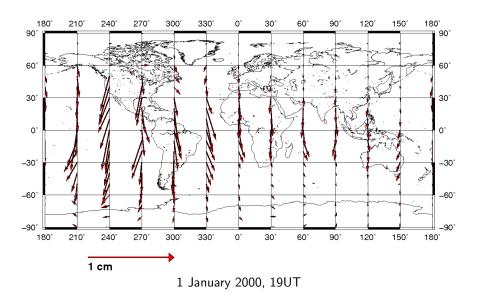


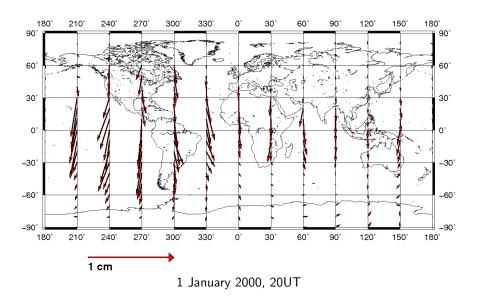
1301

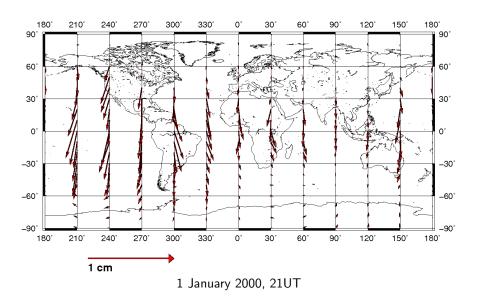




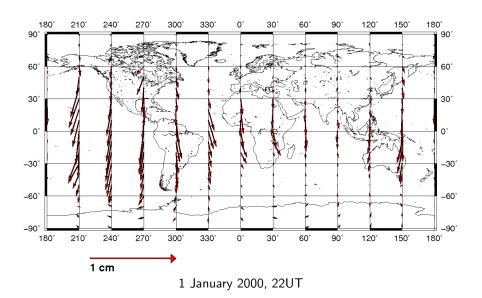




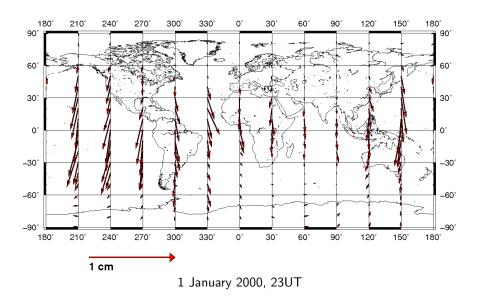


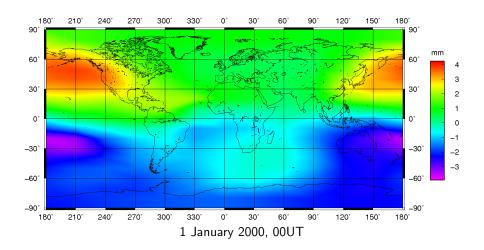


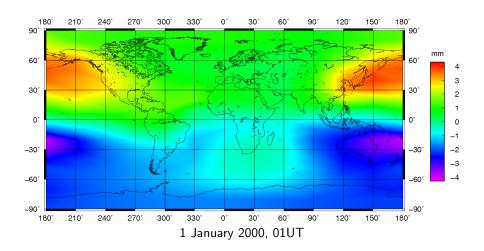
#### Using IRI: apparent horizontal errors

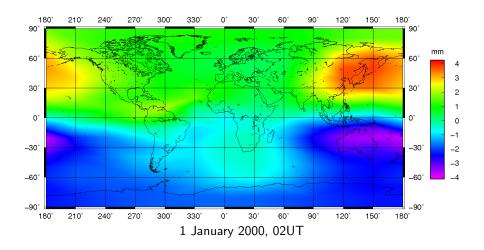


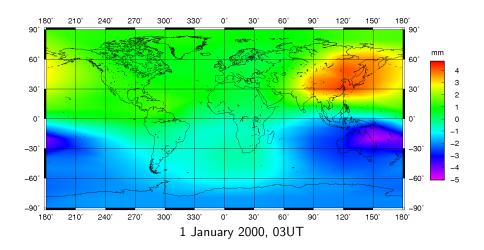
## Using IRI: apparent horizontal errors

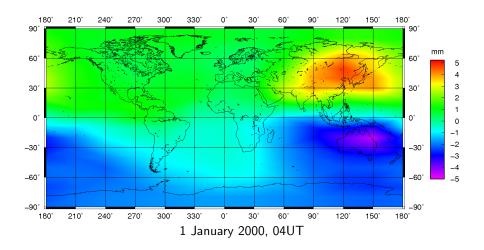


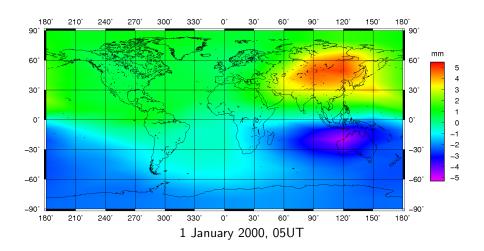


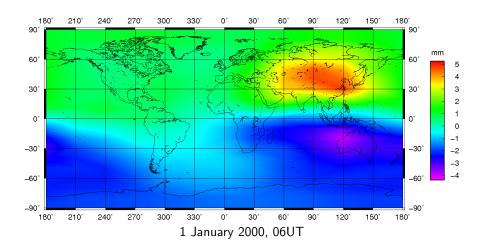


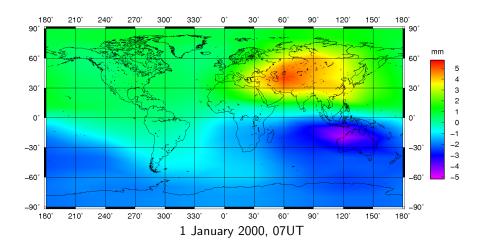


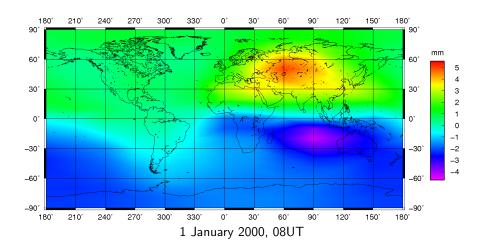


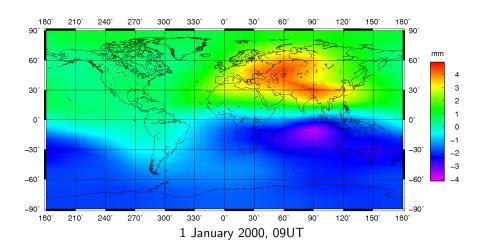


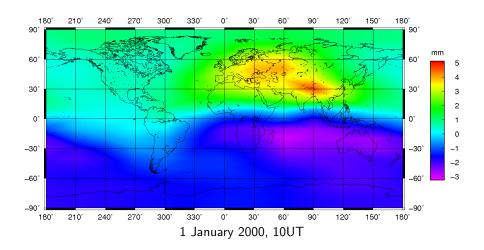


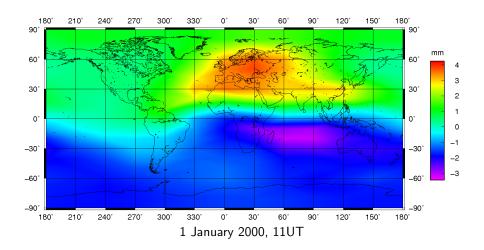


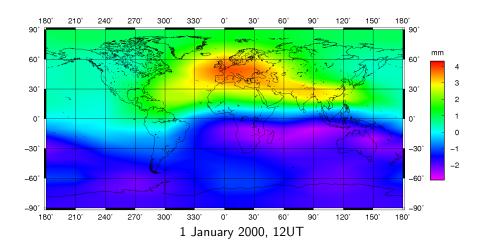


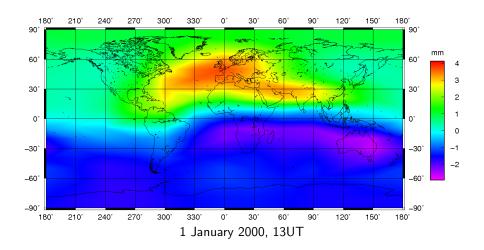


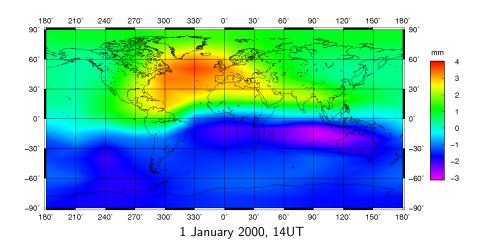


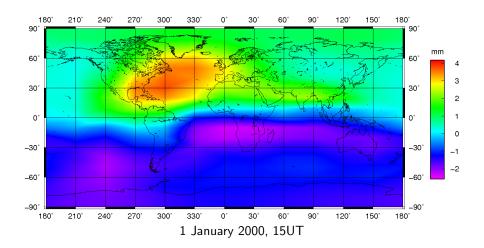


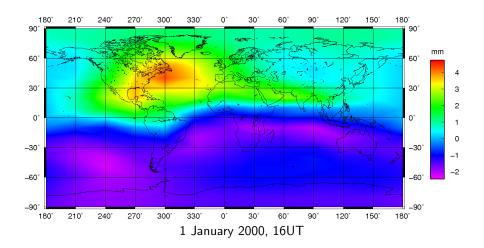


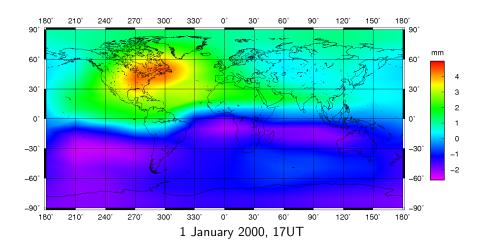


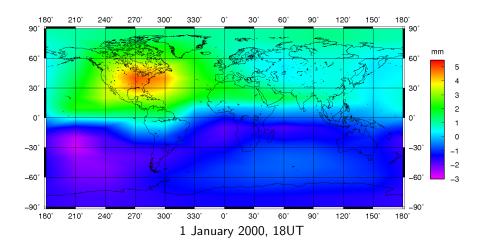


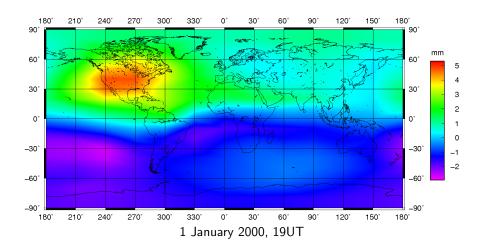


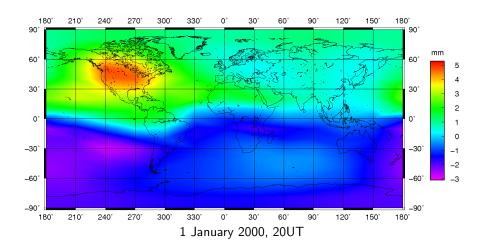


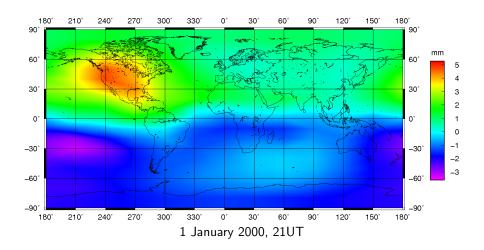


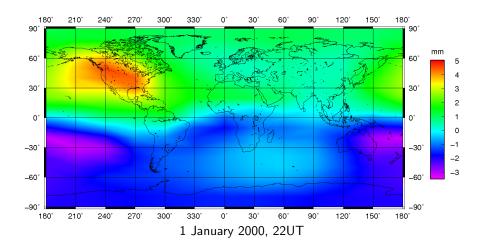


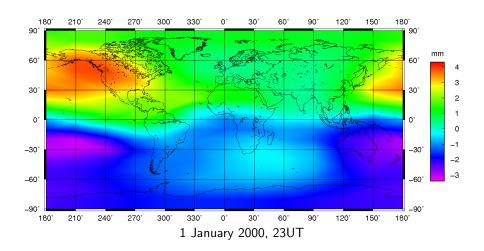




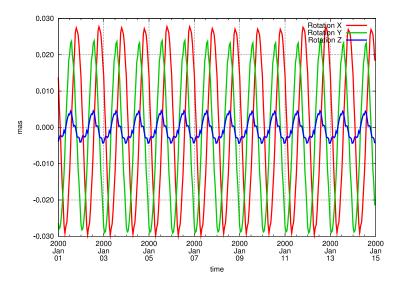




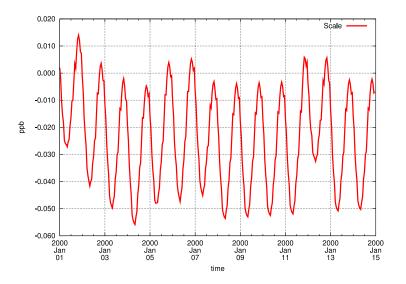




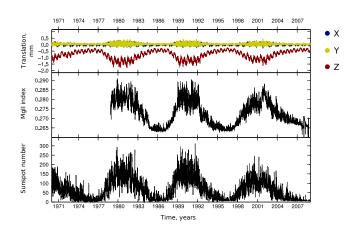
## Apparent Reference Frame Instabilities: subdaily rotations



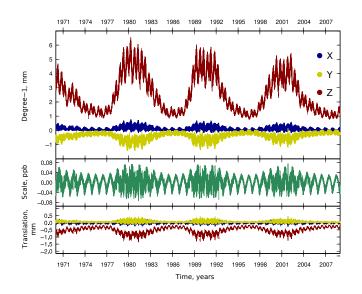
#### Apparent Reference Frame Instabilities: subdaily scale



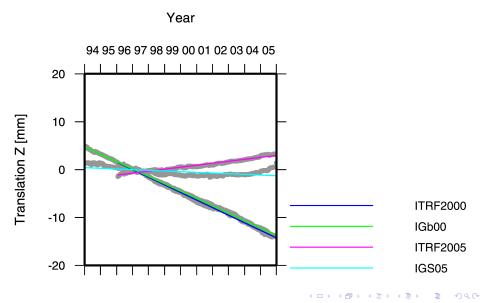
# Apparent Geocentre oscillations and solar activity: Helmert Transform



## Apparent Geocentre oscillations and solar activity: Degree-1



#### Z-translation: Rülke et al 2008



#### Conclusions

- GNSS is reliable and precise geodetic gool even in the presence of geomagnetic disturbances;
- But the accuracy and precision of space geodetic techniques has reached so high levels, that accounting for the 2-nd order ionospheric effect becomes necessary.
- Ionospheric effect of 2-nd order causes systematic errors in determination of coordinates of sites, satellite orbits, geocenter position, other parameters.
   This error has seasonal and 11-year variability and can propagate into the systematic error in satellite oceanic and ice altimetry, if the orbits of the altimeters are being determined using GNSS.
- Fortunately, satellite altimetry is not influenced by to 2-nd order error because of much higher sounding frequency. The same is true for SAR, but SAR is even less prone because of local character of measurements.

#### To be done

- Satellite orbits as determined from an analysis lacking 2-nd order model!!!
- Studies of spatial and temporal variability of 2-nd order effect using real TEC data from GNSS.
- More detailed comparison of geodetic solutions using radiotechnical data (subject to 2-nd order effect) and alternative technics, namely SLR;
- Real GNSS data processing with and without accounting for the 2nd order term: comparison.