

Australian Government

Geoscience Australia

An Assessment of the Value of SLR Observations to GNSS

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Inventory of GNSS Data Processed 9910 -- 0908





Assessment Measures

- Orbit Quality
 - WRMS of Fit
 - SRP Scale -- consistency
- SLR Products -- Benchmark solution
 - GA ITRF2008 submission + 2009 Weekly solutions
 - Geocentre from degree one coefficients
 - XPOLE, YPOLE, LOD
 - Minimally Constrained
- Single Number Description of Assessment

Number of Observations per satellite – per 7-day arc





Median = 185







Median = 130

Median = 117

Number of Observations per satellite – per 7-day arc



GLONASS-95

Median = 142







Median = 152

Median = 160

Number of Observations per satellite – per 7-day arc





Median = 163

Median = 170





Number of Stations observing per 7-day arc

	Minimum	Median	Maximum
GLNSS-80	5	11	21
GLNSS-84	3	9	15
GLNSS-86	3	8	15
GLNSS-87	3	9	16
GLNSS-89	4	12	18
GLNSS-95	4	10	18
GLNSS-99	5	10	16
GLNSS-102	3	9	16
GLNSS-109	3	9	16
GLNSS-115	б	10	17

POD Results: RMS of Orbit Fit (cm)



GLONASS-84

Mean WRMS = 1.75





MJD (Weekly Arcs)

52625

52750

Mean WRMS =2.00

Mean WRMS = 2.74



52500

7.5

5.0

2.5

0.0

52375

POD Results: RMS of Orbit Fit



GLONASS-95

Mean WRMS = 2.04

Mean WRMS = 1.68





Mean WRMS = 1.55

Mean WRMS = 2.26

POD Results: RMS of Orbit Fit

GLONASS-109

Mean WRMS = 1.81



Mean WRMS = 2.17





POD Results: SRP Scale Factor



GLONASS-84

54250



GLONASS-86



POD Results: SRP Scale Factor



POD Results: SRP Scale Factor



SRP Scale Factor

1.3 for GLONASS 80 – 89; 1.8 for GLONASS 95-115
Two Different Spacecraft – Mass, Surface Area, SRP Model ?
Slow Attitude change, Changing Orientation of Solar Panels ?
180-day Jumps ?

Results: COM Benchmark Solution





X-Geocentre (mm)

Y-Geocentre (mm)



Blue = GA gravimetric solution Red = GA ITRF2008 + Weekly

Z-Geocentre (mm)

LAGET + GLNSS Geocentre





X-Geocentre (mm)

Y-Geocentre (mm)



Z-Geocentre (mm)

Assessment of CoM

LAGET+GLNSS

LAGET

	Mean of Uncertainty (mm)	RMS of Uncertainty (mm)			Mean of Uncertainty (mm)	RMS of Uncertainty (mm)
X-com	2.01	9.92	X	-com	2.02	9.20
Y-com	2.06	11.36	Y	-com	2.01	8.84
Z-com	3.97	14.46	Z	-com	3.54	14.00

XPOLE (mas)





Mean uncertainty per arc



+VE = improvement



RMS of uncertainty per arc

% difference in uncertainty

YPOLE (mas)





Mean uncertainty per arc







RMS of uncertainty per arc

% difference in uncertainty

LOD: LARET vs. LARET+GLNSS (ms)





Mean uncertainty per arc



+VE = improvement



% difference in uncertainty

RMS of uncertainty per arc

Assessment: For 10% improvement

Arc 021027	#Obs	#Stns	Arc 060924	#Obs	#Stns
GLNSS-84	118	8	GLNSS-87	240	13
GLNSS-86	131	9	GLNSS-89	142	8
GLNSS-87	119	8	GLNSS-95	143	11

Arc 080727	#Obs	#Stns
GLNSS-99	150	10
GLNSS- 102	106	8
GLNSS- 109	79	5

Tables show the observation configuration of typical arcs where an improvement of ~10% was achieved in the uncertainties in XPOLE and YPOLE components.

- The main aim of this study was to assess the potential of SLR observations to GNSS satellites to contribute to the improvement of, specifically, SLR products – Geocentre (for the Terrestrial Reference Frame definition) and Earth Orientation Parameters.
- For the purposes of this assessment, all available GLONASS SLR data for the period October 1999 (Start of GLONASS-80 data) to end August 2009 – comprising 10 satellites was processed for POD and combined with the standard Lageos and Etalon data over this period.

• The combined solutions were compared with a benchmark solution – which was the Geoscience Australia contribution to the ILRSA combination for ITRF2008 – the parameters being the gravimetric (geocentre), XPOLE, YPOLE and LOD. The gravimetric geocentre was compared with the ILRSA determined translation components for consistency of the two methods.

- The POD results showed a typical RMS fit for the orbit of 2 cm for each 7-day arc for all the satellites.
- The estimated Solar Radiation Pressure (SRP) scale factor gave mean values of 1.38 (for GLONASS 80, 84, 86, 87, 89) and 1.85 (for GLONASS 95, 99, 102 and 115). This identified them as different spacecraft GLONASS and GLONASS-M in their dimensions and construction. This information is not readily available apriori.

• The estimated time series for the SRP scale factor showed a slowly changing attitude or orientation of the satellite or the solar panels with an "abrupt" adjustment every six months. This is consistent in the POD results for all 10 GLONASS satellites in this study. An attitude model to address this effect is also not available.

• The addition of the GLONASS data in determining the gravimetric geocentre has little impact – the mean and RMS of the uncertainties in the z-component deteriorated by 0.4 mm. Although there was no significant change in the mean uncertainty of the ycomponent the RMS did deteriorate by a significant 2.5 mm.

• The addition of the GLONASS data had a large impact on uncertainties of the EOP estimates. For both the x and y pole coordinate uncertainties, in each case there was an improvement in approximately 50% of the computed arcs and a detrioration in approximately 50% of the computed arcs. The uncertainty in the LOD estimates deteriorated for the majority of the computed arcs.

• Although the study implies that there is potential for GNSS SLR observations to positively impact the quality of SLR products, some major and important issues of space segment information has been identified and needs to be addressed if GNSS ranging is to be a contributor to the enhancement of SLR TRF products.

- Immediate Issue #1: Satellite Centre of Mass Offset.
 - The "instantaneous" offset between the satellite centre of mass and the LRA has to be provided for each (identified) satellite in an unambiguous form. This should comprise the nominal CoM offset and the "instantaneous" location of the CoM with respect to the origin of satellite's the body fixed frame. This value has to be updated at least after every manoeuvre.

- Immediate Issue #2: Solar Radiation Pressure
 - The physical dimensions ("instantaneous" mass and surface area) need to be provided for each satellite for SRP computations. The "instantaneous" mass should be provided with the location of the satellite CoM; at least after every manoeuvre.
 - If available, a SRP maromodel that may have been developed by GLONASS for each model of the constellation could be provided.

Immediate Issue #3: Attitude and Manoeuvres

 An attitude model is required and the six-monthly re-orientation of the satellite explained.

