





chambres photogoniométriques placées de manière qu'on puisse photogra-

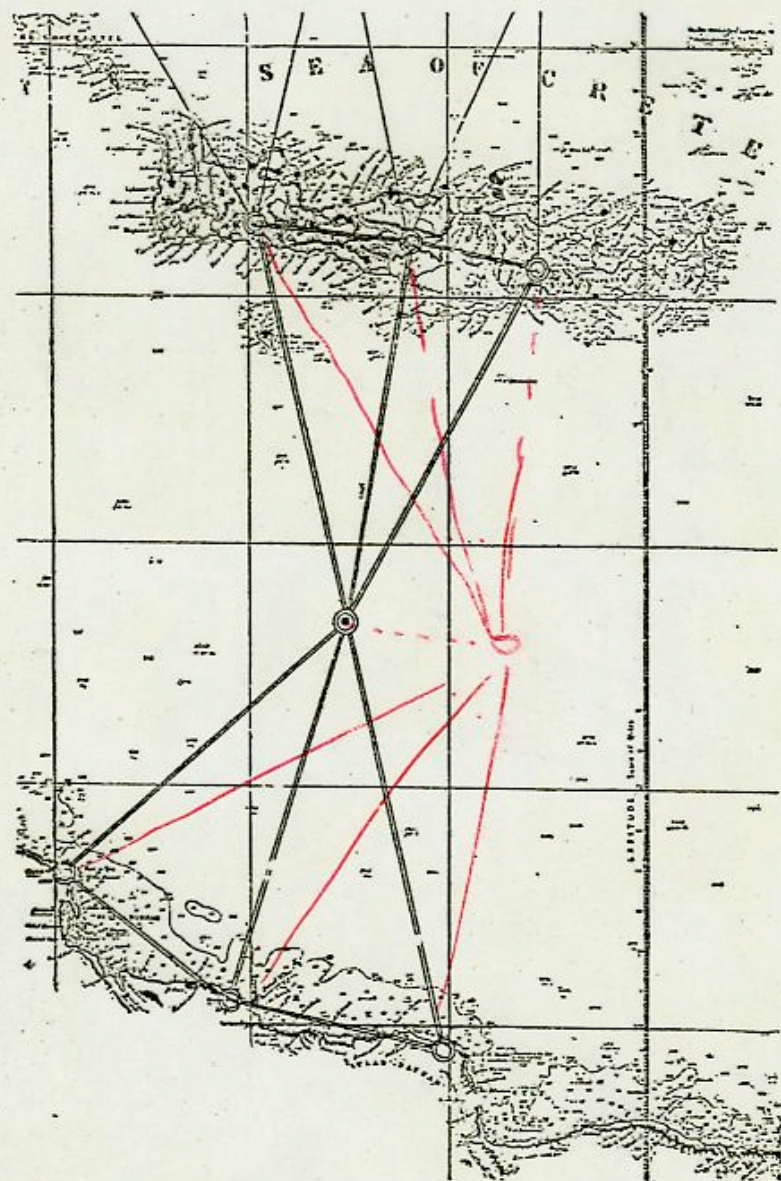
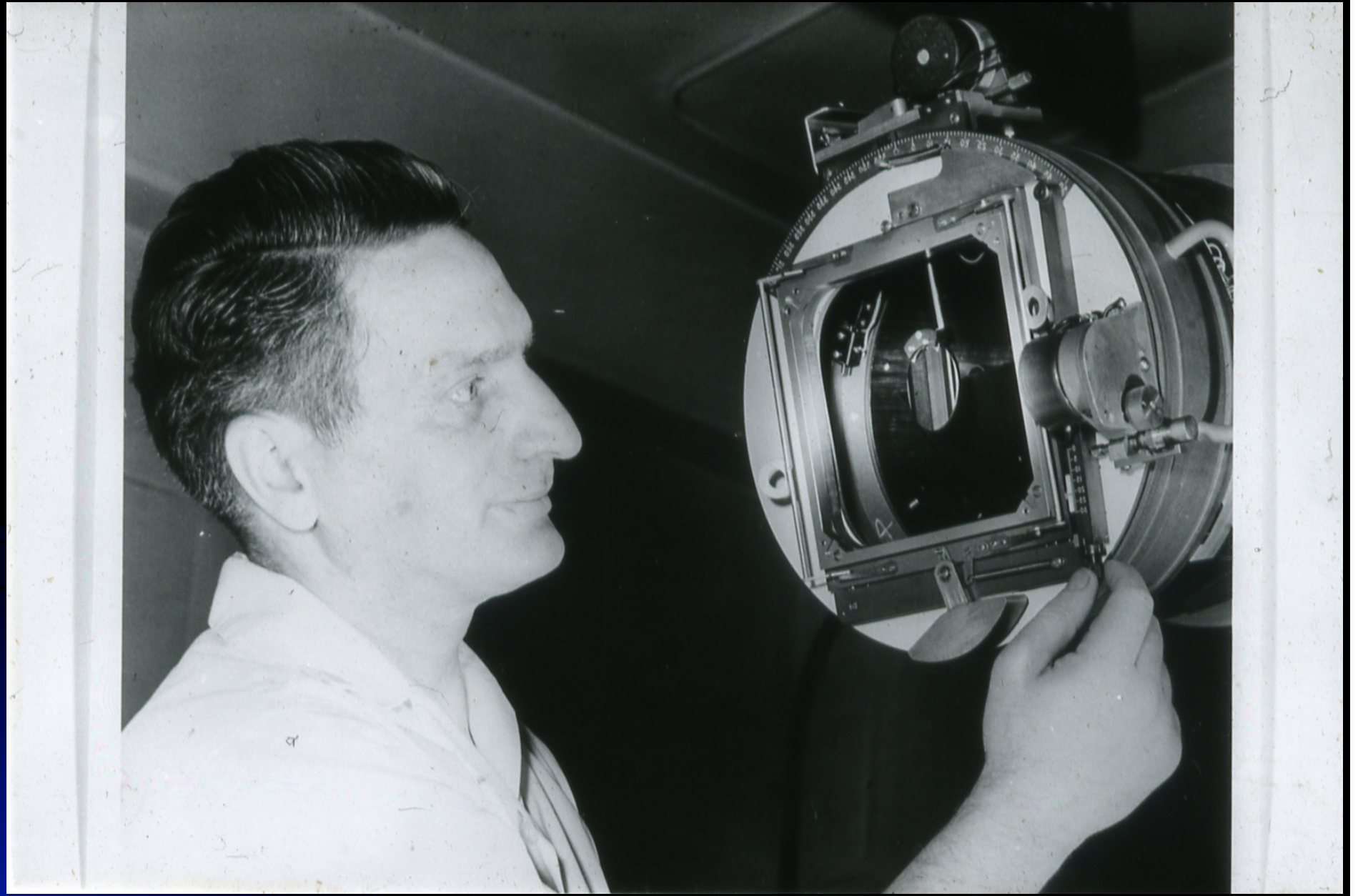
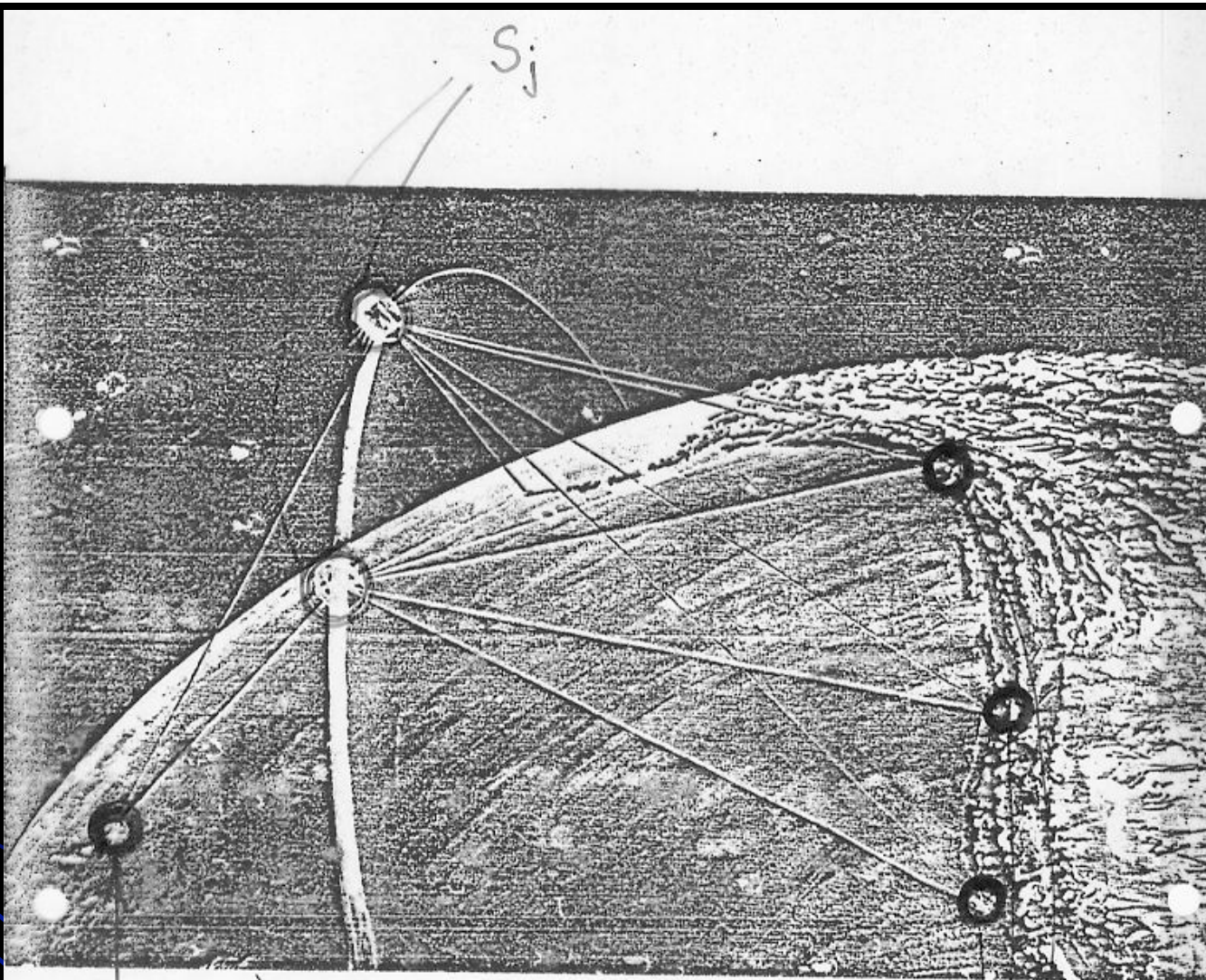


Fig. 2

phier simultanément sur la même plaque. Cette plaque aura une épaisseur



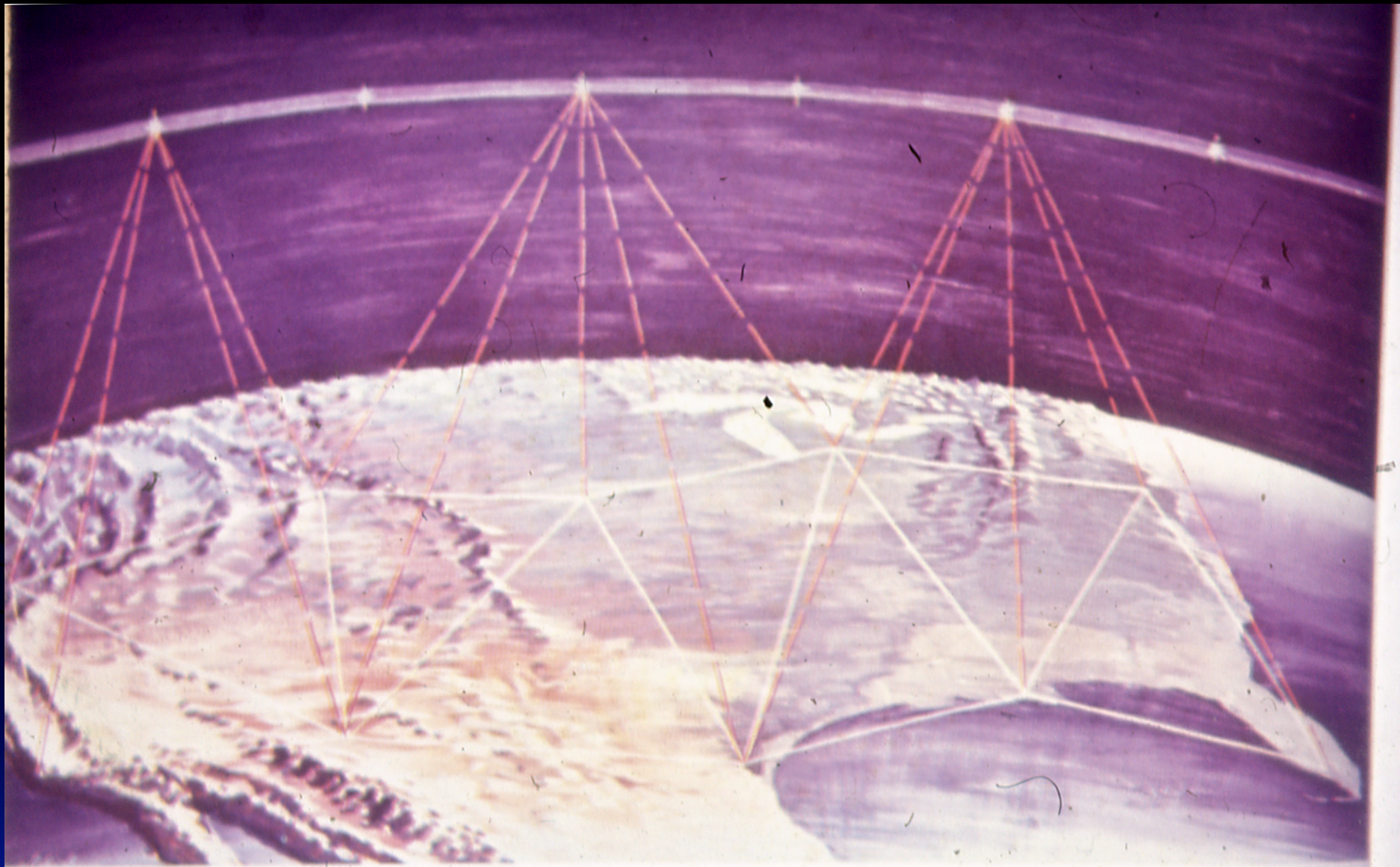




Continental networks can be extended to locate isolated islands, such as Hawaii.

Q?

P_i



By photographing the successive positions of a satellite in its orbit from strategically placed camera sites a network of relatively few triangles can be established covering a whole continent.

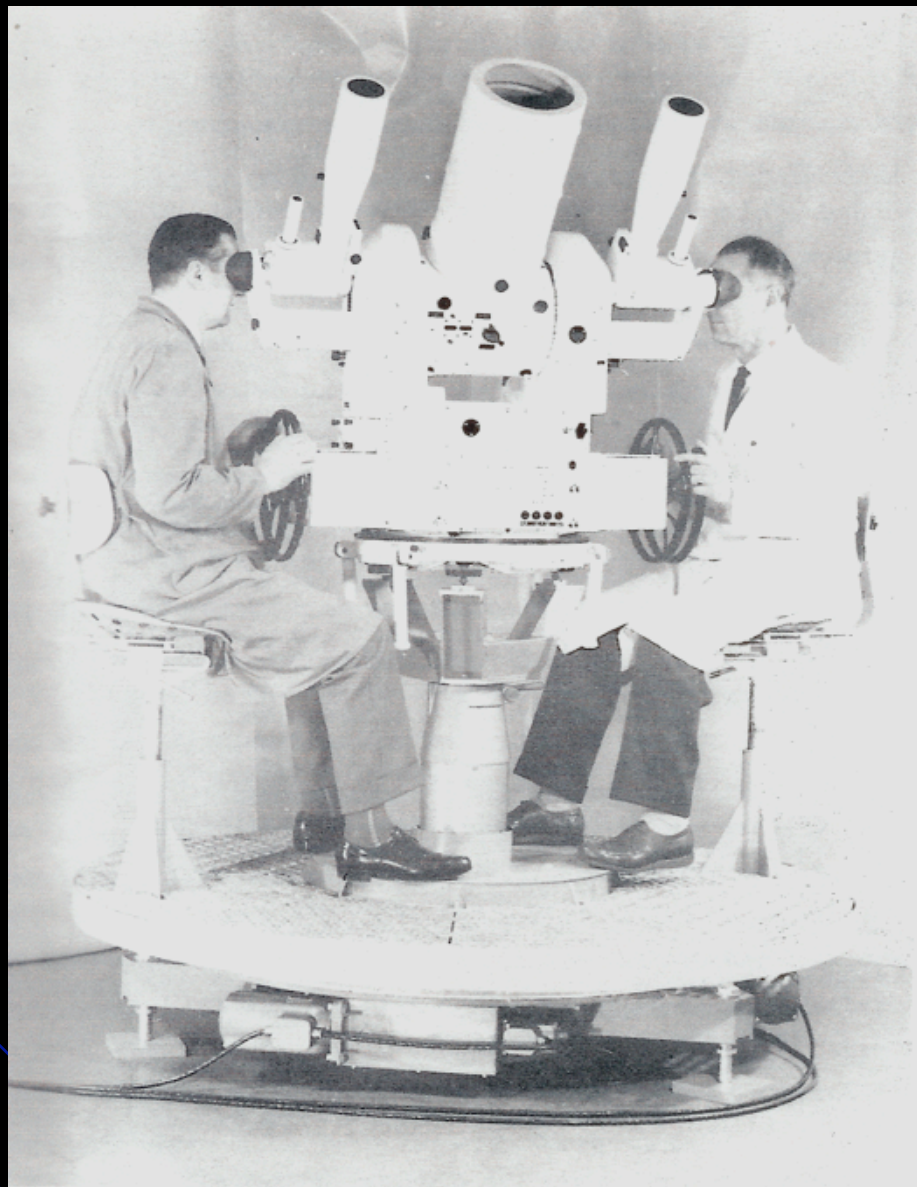
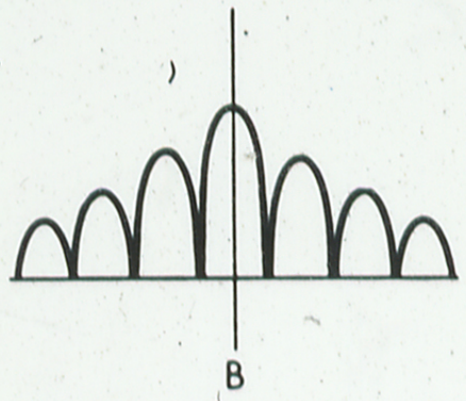
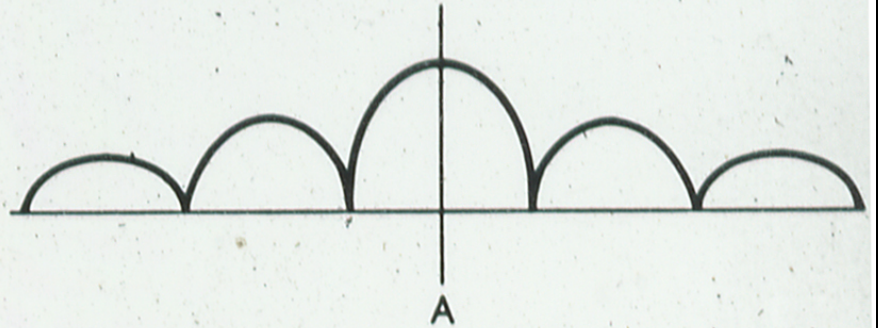
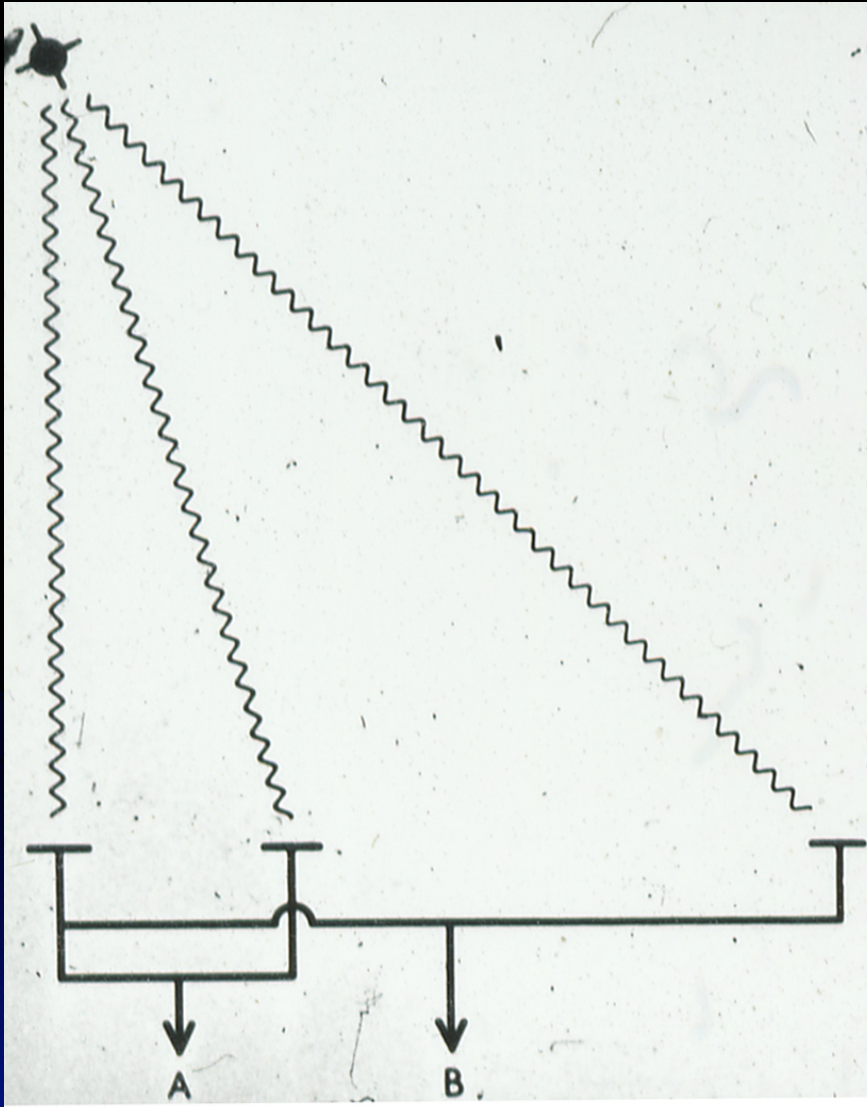
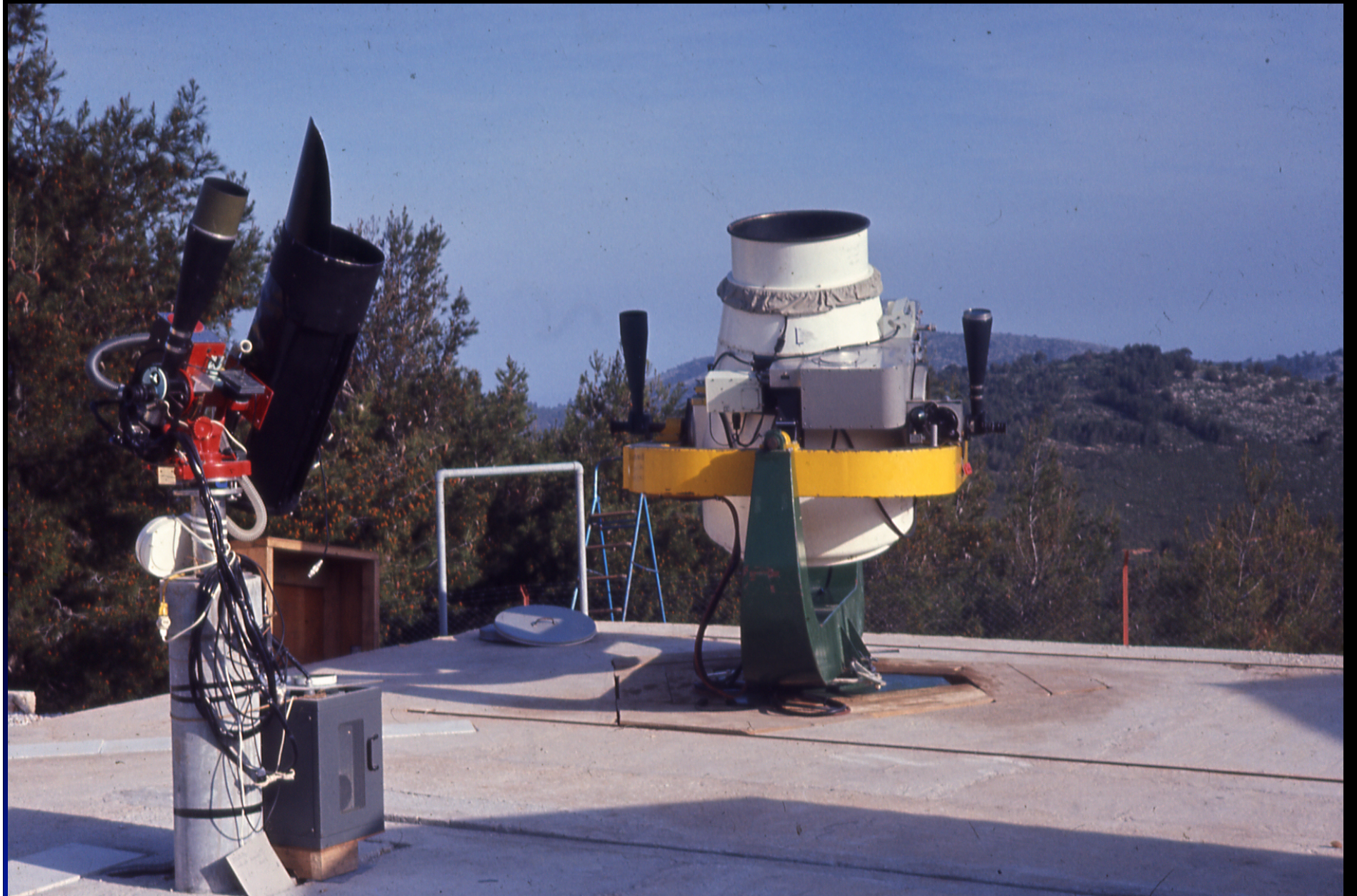
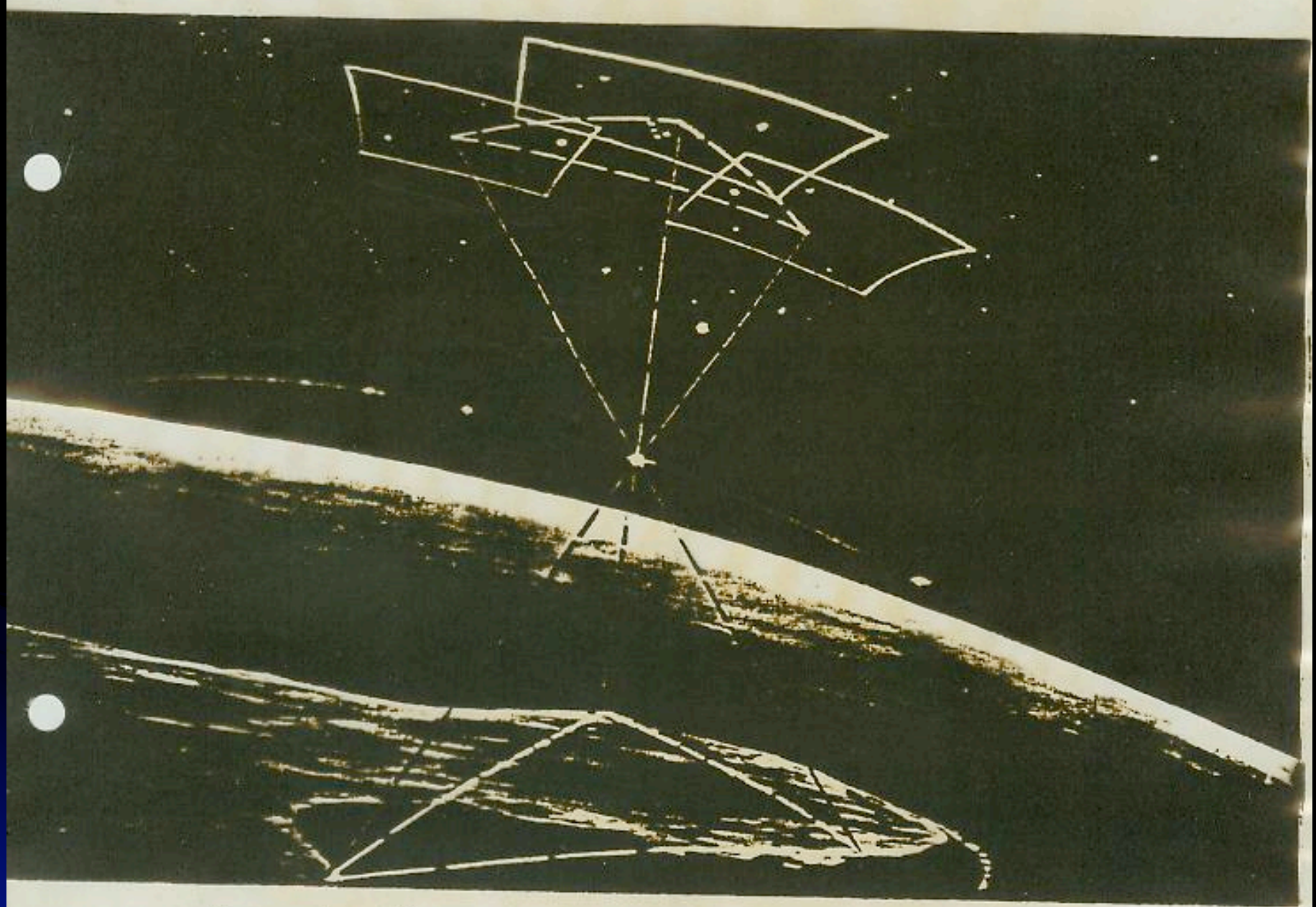


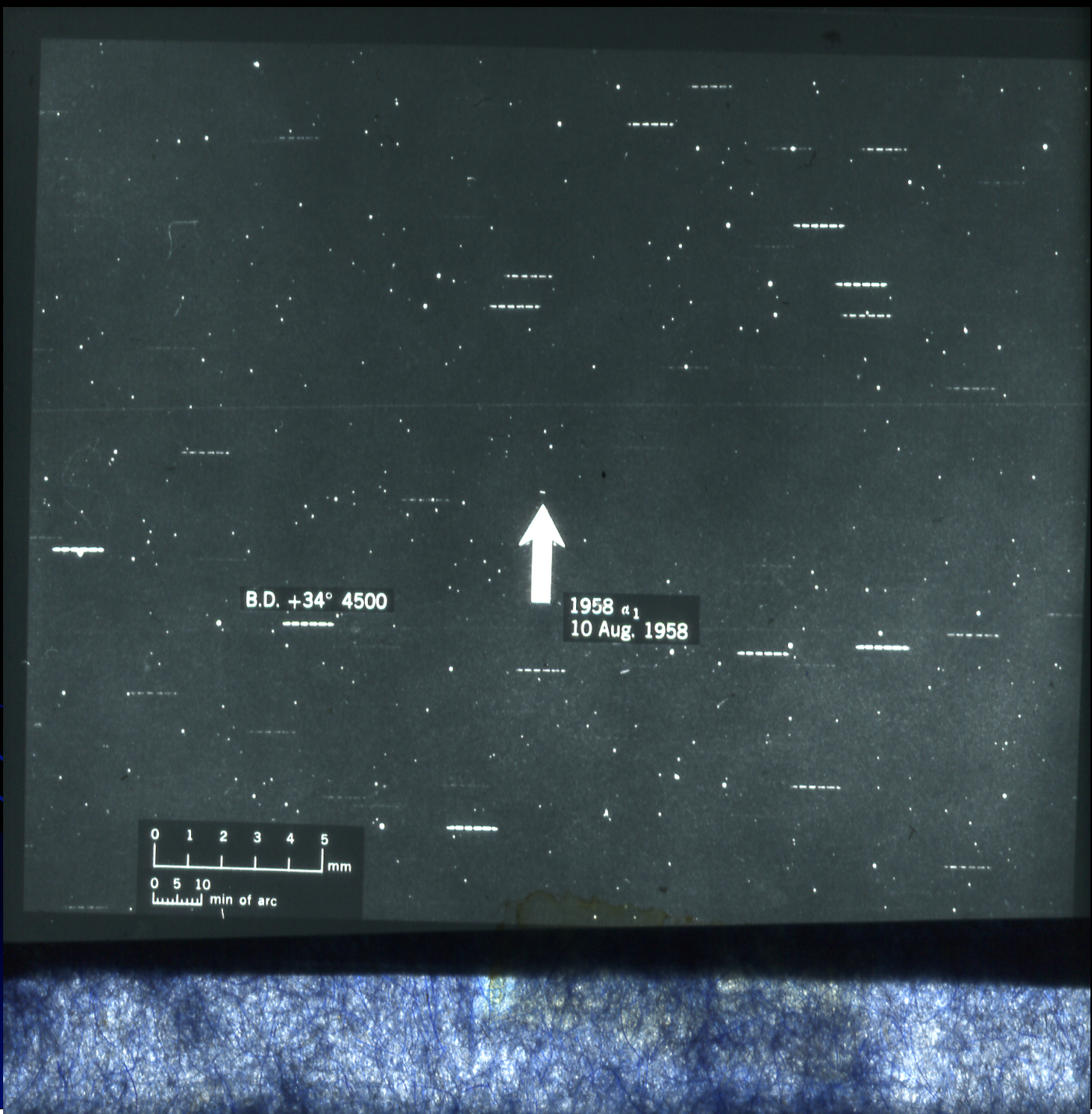
Fig. 9 Cinetheodolite. (Courtesy Askania Werke.)







Cameras at three widely separated points in Florida photograph simultaneously an artificial satellite against their respective star backgrounds.



B.D. +34° 4500

1958 α₁
10 Aug. 1958

0 1 2 3 4 5 mm
0 5 10 min of arc



tional radius $m = GM/c^2$, eqs. (37) to (40) become

$$y'_{11} = -\frac{4m}{r} \left(\frac{\Omega R}{c}\right)^2$$

$$\sum_{n=0}^{\infty} \left[\frac{I_n}{2} \left(\frac{R}{r}\right)^n P_n(\sin \phi) \mp \frac{(n-2)! L_n}{(n+2)!} \frac{L_n}{2} \left(\frac{R}{r}\right)^n P_n^2(\sin \phi) \cos 2\theta \right], \quad (41)$$

$$y'_{44} = \frac{4m}{r} \left[1 - \sum_{n=2}^{\infty} J_n \left(\frac{R}{r}\right)^n P_n(\sin \phi) \right], \quad (42)$$

$$g_{12} = y'_{12} = \frac{4m}{r} \left(\frac{\Omega R}{c}\right)^2 \sum_{n=1}^{\infty} \left[\frac{(n-2)! L_n}{(n+2)!} \frac{L_n}{2} \left(\frac{R}{r}\right)^n P_n^2(\sin \phi) \sin 2\theta \right], \quad (43)$$

$$g_{24} = y'_{24} = \pm i \frac{4m}{r} \left(\frac{\Omega R}{c}\right)^2 \sum_{n=1}^{\infty} \left[\frac{(n-1)!}{(n+1)!} K_n \left(\frac{R}{r}\right)^n P_n^1(\sin \phi) \frac{\sin \theta}{\cos \theta} \right], \quad (44)$$

and therefore

$$\sum_{\alpha=1}^4 y'_{\alpha\alpha} = \frac{4m}{r} \left[1 - \sum_{n=2}^{\infty} J_n \left(\frac{R}{r}\right)^n P_n(\sin \phi) - \left(\frac{\Omega R}{c}\right)^2 \sum_{n=0}^{\infty} I_n \left(\frac{R}{r}\right)^n P_n(\sin \phi) \right].$$

Because

$$g_{kk} = -1 + y'_{kk} - \frac{1}{2} \sum_{\alpha=1}^4 y'_{\alpha\alpha},$$

it follows that

$$g_{11} = -1 - \frac{2m}{r} \left[1 - \sum_{n=2}^{\infty} J_n \left(\frac{R}{r}\right)^n P_n(\sin \phi) \mp \left(\frac{\Omega R}{c}\right)^2 \frac{(n-2)! L_n}{(n+2)!} \frac{L_n}{2} \left(\frac{R}{r}\right)^n P_n^2(\sin \phi) \cos 2\theta \right], \quad (45)$$

$$g_{33} = -1 - \frac{2m}{r} \left[1 - \sum_{n=2}^{\infty} J_n \left(\frac{R}{r}\right)^n P_n(\sin \phi) - \left(\frac{\Omega R}{c}\right)^2 \sum_{n=0}^{\infty} I_n \left(\frac{R}{r}\right)^n P_n(\sin \phi) \right], \quad (46)$$

$$g_{44} = -1 + \frac{2m}{r} \left[1 - \sum_{n=2}^{\infty} J_n \left(\frac{R}{r}\right)^n P_n(\sin \phi) + \left(\frac{\Omega R}{c}\right)^2 \sum_{n=0}^{\infty} I_n \left(\frac{R}{r}\right)^n P_n(\sin \phi) \right]. \quad (47)$$

Taking only the terms up to $n=2$, the components of the fundamental tensor are

$$g_{11} = -1 - \frac{2m}{r} \left[1 - J_2 \left(\frac{R}{r}\right)^2 P_2(\sin \phi) \mp \left(\frac{\Omega R}{c}\right)^2 \frac{L_2}{8} \left(\frac{R}{r}\right)^2 \cos^2 \phi \cos 2\theta \right], \quad (48)$$

$$g_{33} = -1 - \frac{2m}{r} \left\{ 1 - J_2 \left(\frac{R}{r}\right)^2 P_2(\sin \phi) - \left(\frac{\Omega R}{c}\right)^2 \left[\Gamma + d \left(\frac{R}{r}\right)^2 P_2(\sin \phi) \right] \right\}, \quad (49)$$

$$g_{44} = -1 + \frac{2m}{r} \left\{ 1 - J_2 \left(\frac{R}{r}\right)^2 P_2(\sin \phi) + \left(\frac{\Omega R}{c}\right)^2 \left[\Gamma + d \left(\frac{R}{r}\right)^2 P_2(\sin \phi) \right] \right\}, \quad (50)$$

$$g_{12} = \frac{4m}{r} \left(\frac{\Omega R}{c}\right)^2 \frac{L_2}{16} \left(\frac{R}{r}\right)^2 \cos^2 \phi \sin 2\theta, \quad (51)$$

$$g_{24} = \pm i \frac{2m}{r} \left(\frac{\Omega R}{c}\right)^2 \Gamma \left(\frac{R}{r}\right) \cos \phi \frac{\sin \theta}{\cos \theta}. \quad (52)$$

In order to have a spherically symmetric field the very small terms proportional to m/r and $(\Omega R/c)^2$ will be neglected. In g_{44} only, the term will be retained and the second-order term

$$2m^2/r^2 \cdot [1 - 2J_2(R/r)^2 P_2(\sin \phi)]$$

will be added according to de Sitter (eq. 20). The reason why g_{44} is required

Accordingly, a position referred to in the celestial system will be transformed into the terrestrial system by

$$x = SNPw \quad (12)$$

As the terrestrial system had to be replaced by a number of geodetic systems, the celestial system is also replaced in practice by a number of systems that

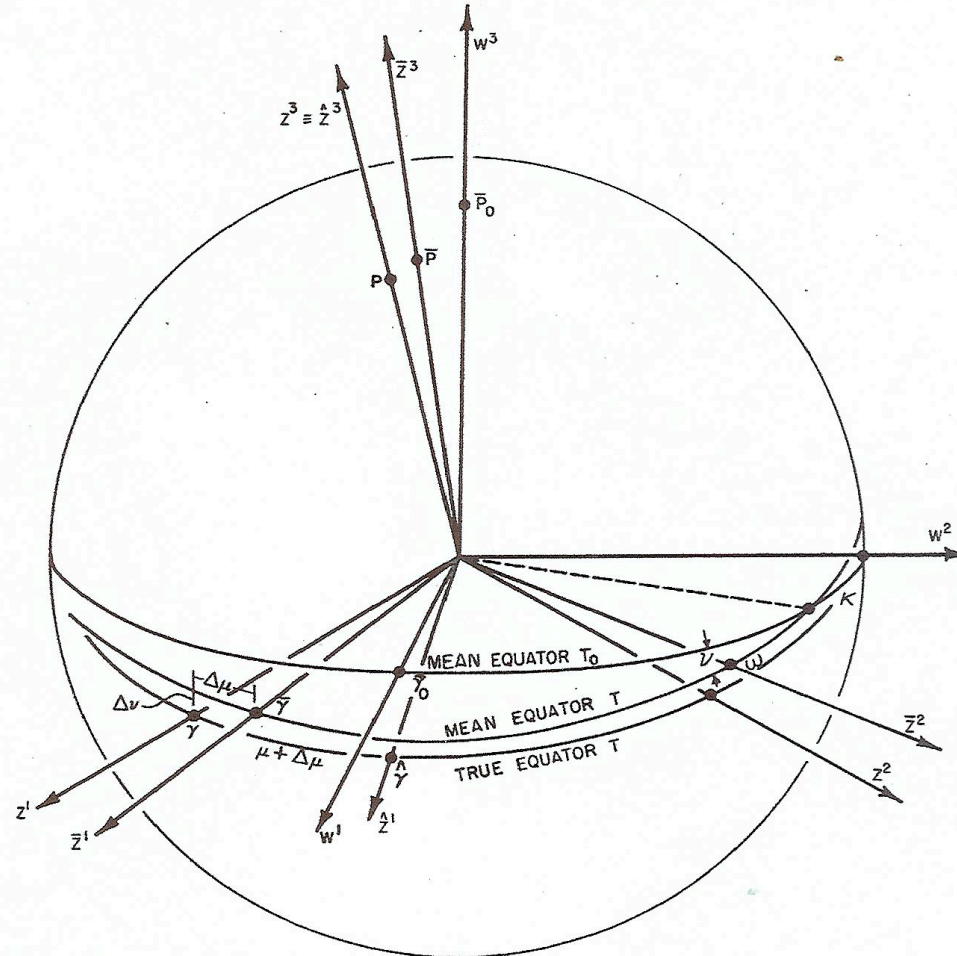


Fig. 3. Sidereal and celestial systems.

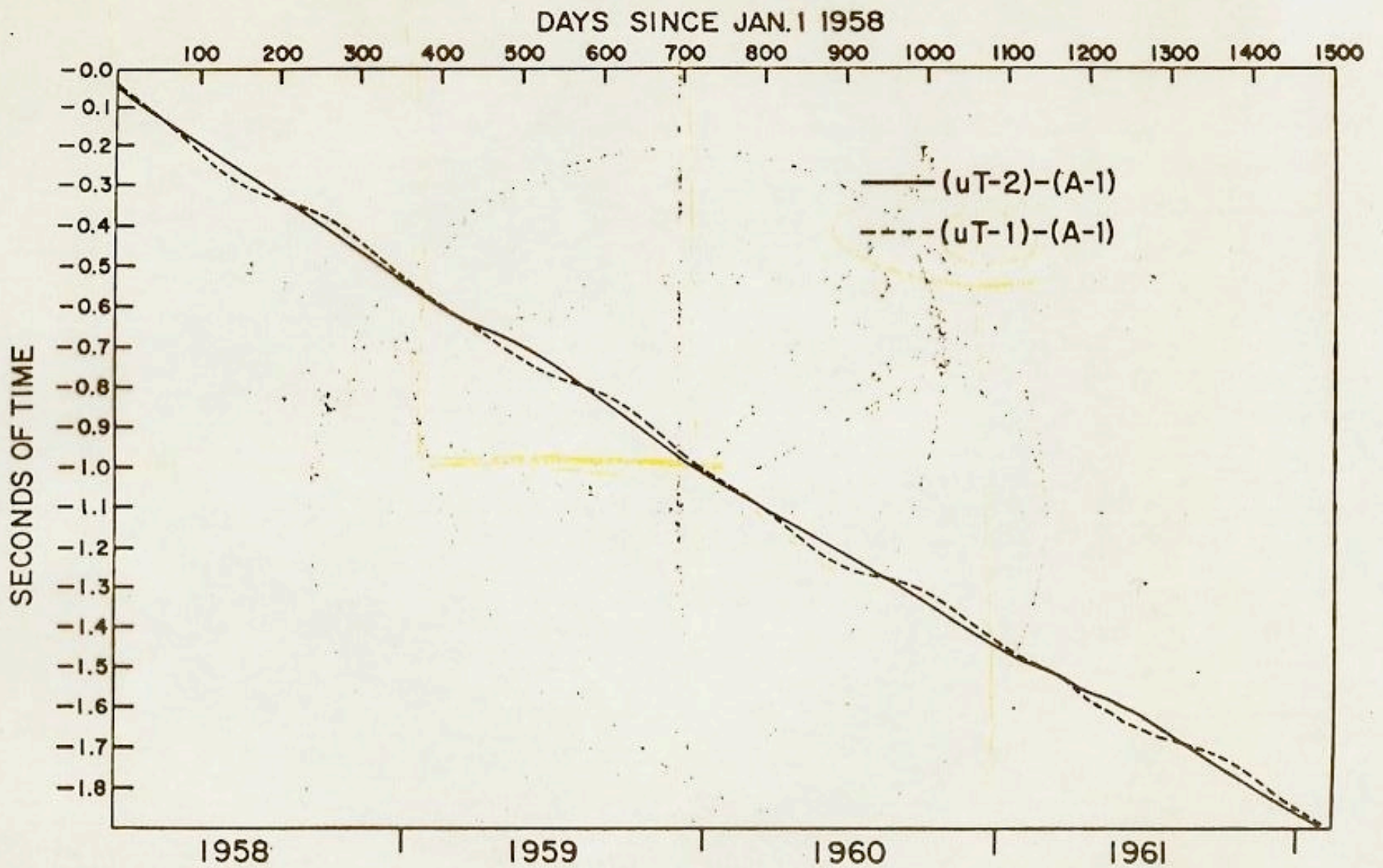


Fig. 2.--Relation between UT-1, UT-2, and A-1.

OUTLINE OF
GEODESY 861 AND 862
MODERN GEOMETRIC GEODESY

OSU
Spring 1958

I. Introduction

- a. Principle - History - Different methods and classification
- b. Geocentric rectangular system of coordinates
- c. The use of stars for determination of directions

II. Methods with the Moon

- a. General
Ephemerides - Ephemeris Time - Libration
- b. Direct observations
 1. Theory
 2. Instruments
 3. Methods of low accuracy
- c. Occultations
 1. Theory - Predictions
 2. Instruments
 3. Applications
- d. Eclipses
 1. Theory - Predictions
 2. Instruments
 3. Applications

III. Methods with rockets

- a. Theory
- b. Instruments
- c. Possible applications - accuracy

IV. Methods with artificial satellites

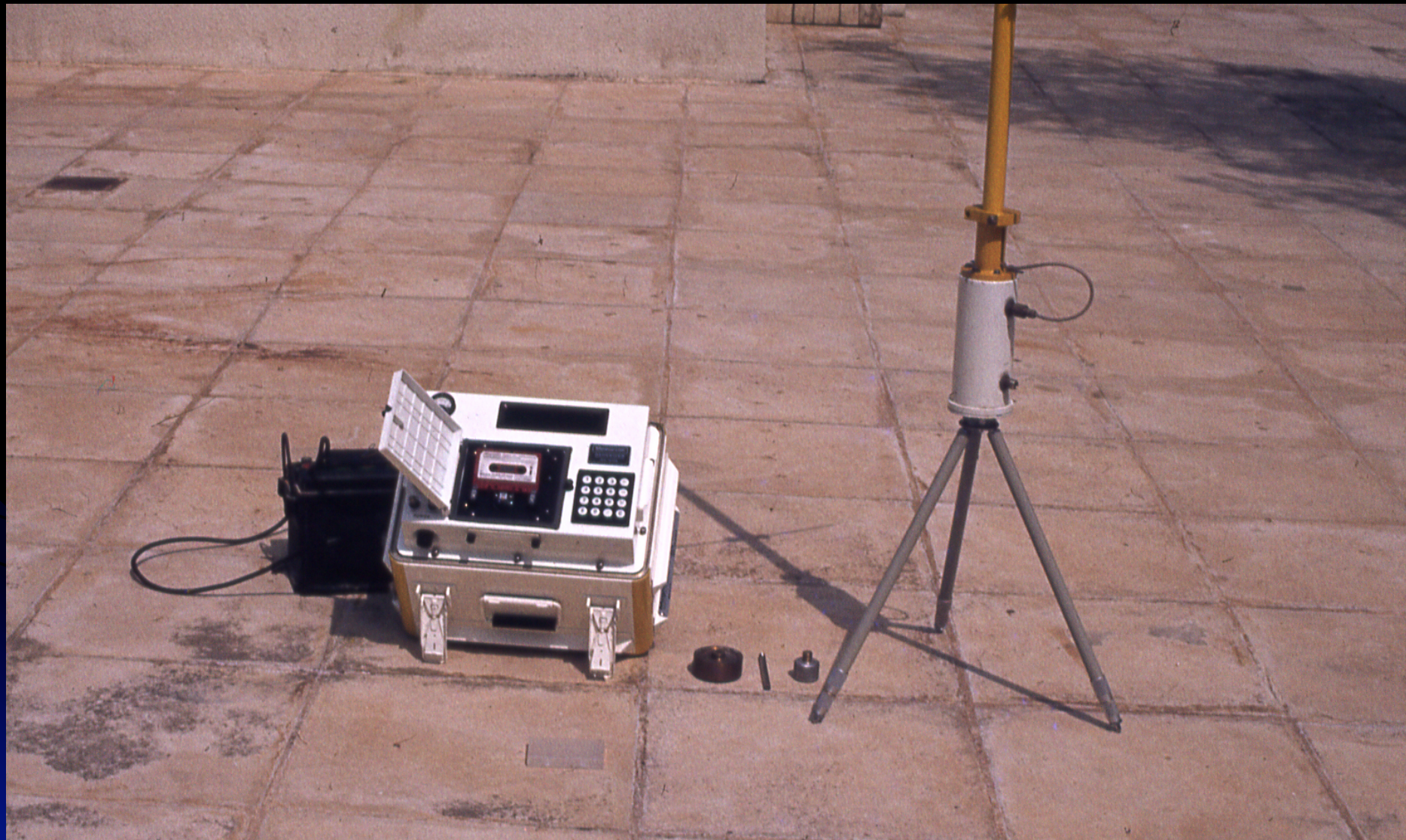
- a. Theory - Orbits
- b. Instruments
- c. Possible applications - accuracy
- d. Geometric and Dynamic Solutions.

V. The direct and indirect geodetic problem for super long distances



INSTRUMENTATION UNIT

NO. 314



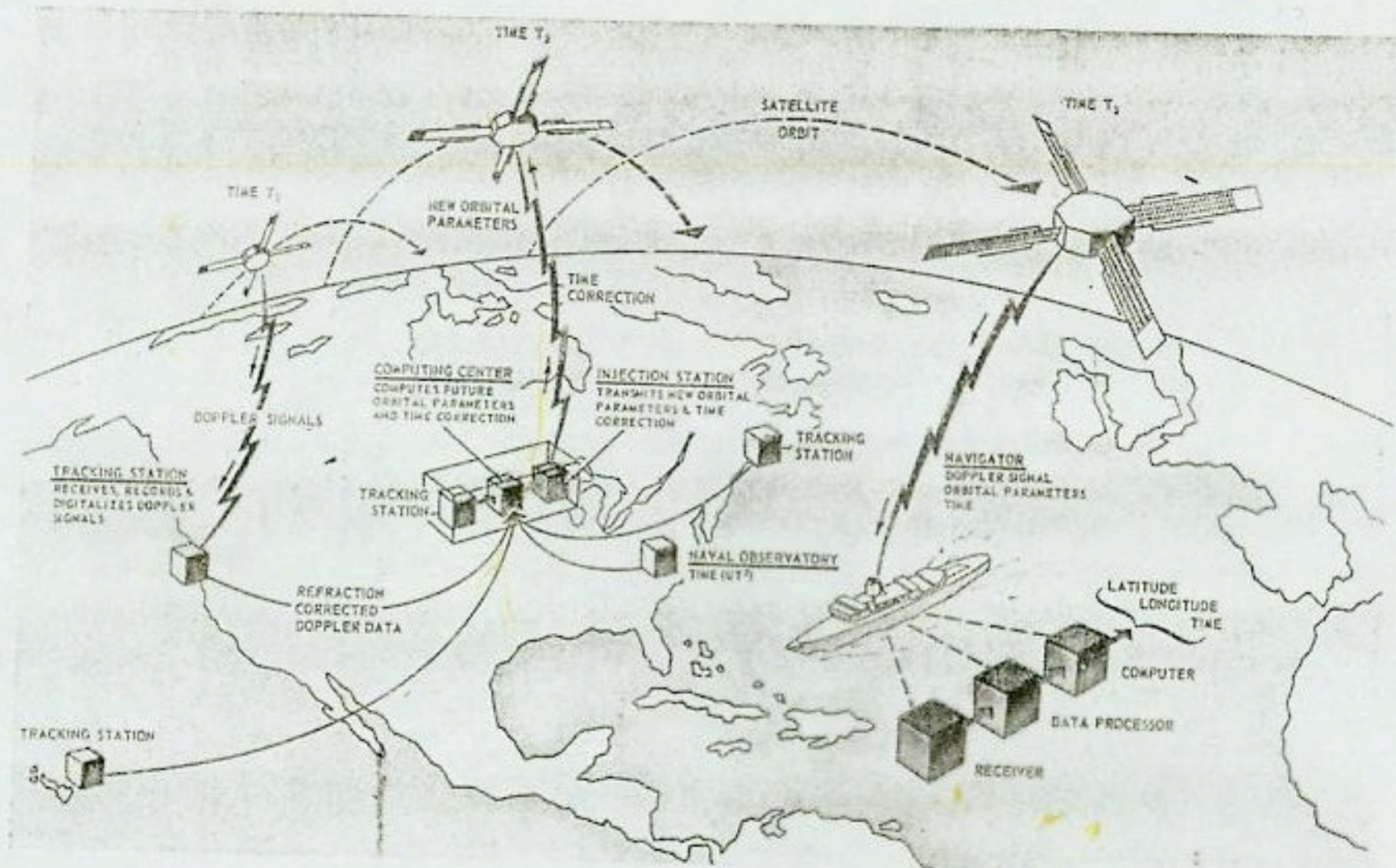


Fig. 2. Operational transit satellite navigation system (high accuracy). Project Transit. (Rev. May 1961).

DETERMINATION OF ORBIT AND RADIOPOSITONING INTEGRATED FROM SATELLITE

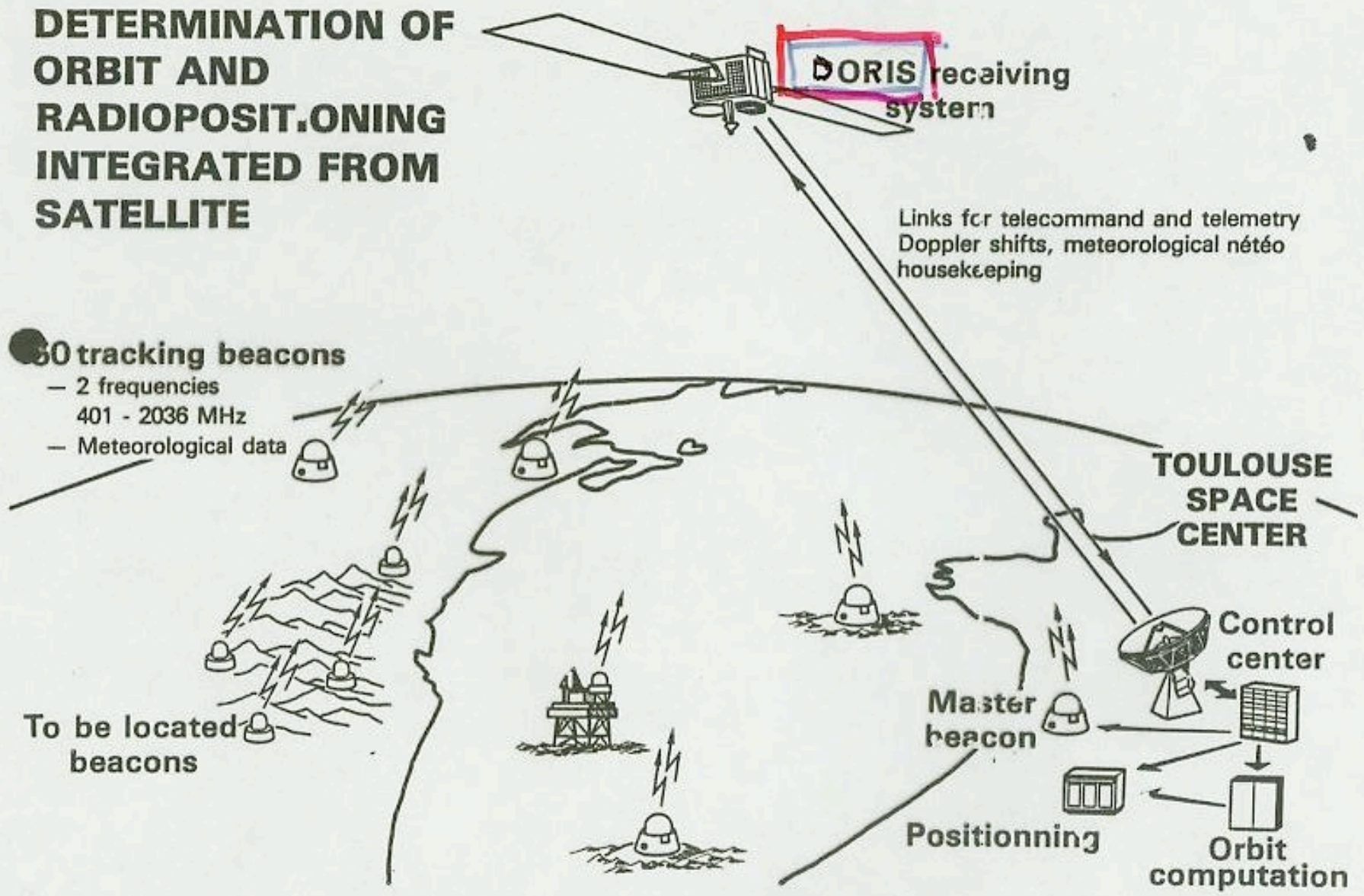


FIGURE 2

MARK M. MACOMBER: *Project Anna*

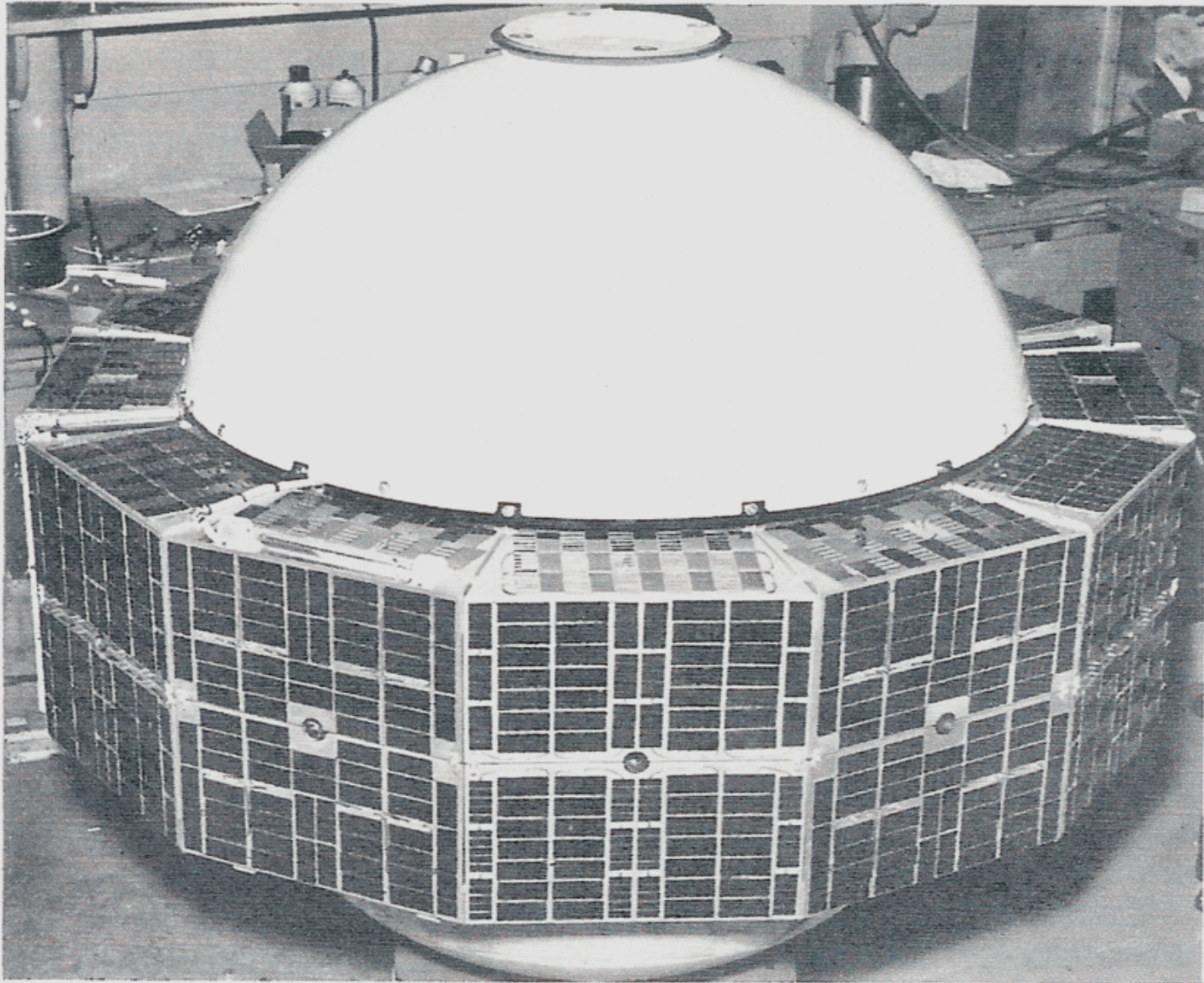


Fig. 1a. The ANNA 1A satellite when ready for launch.

TABLE 5
Numerical results for J_n

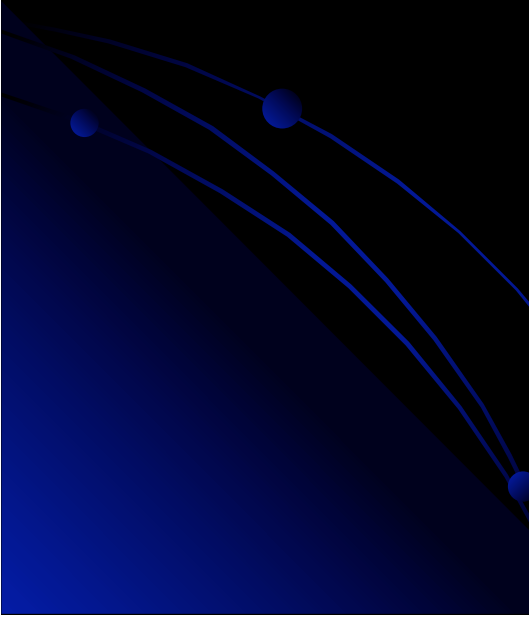
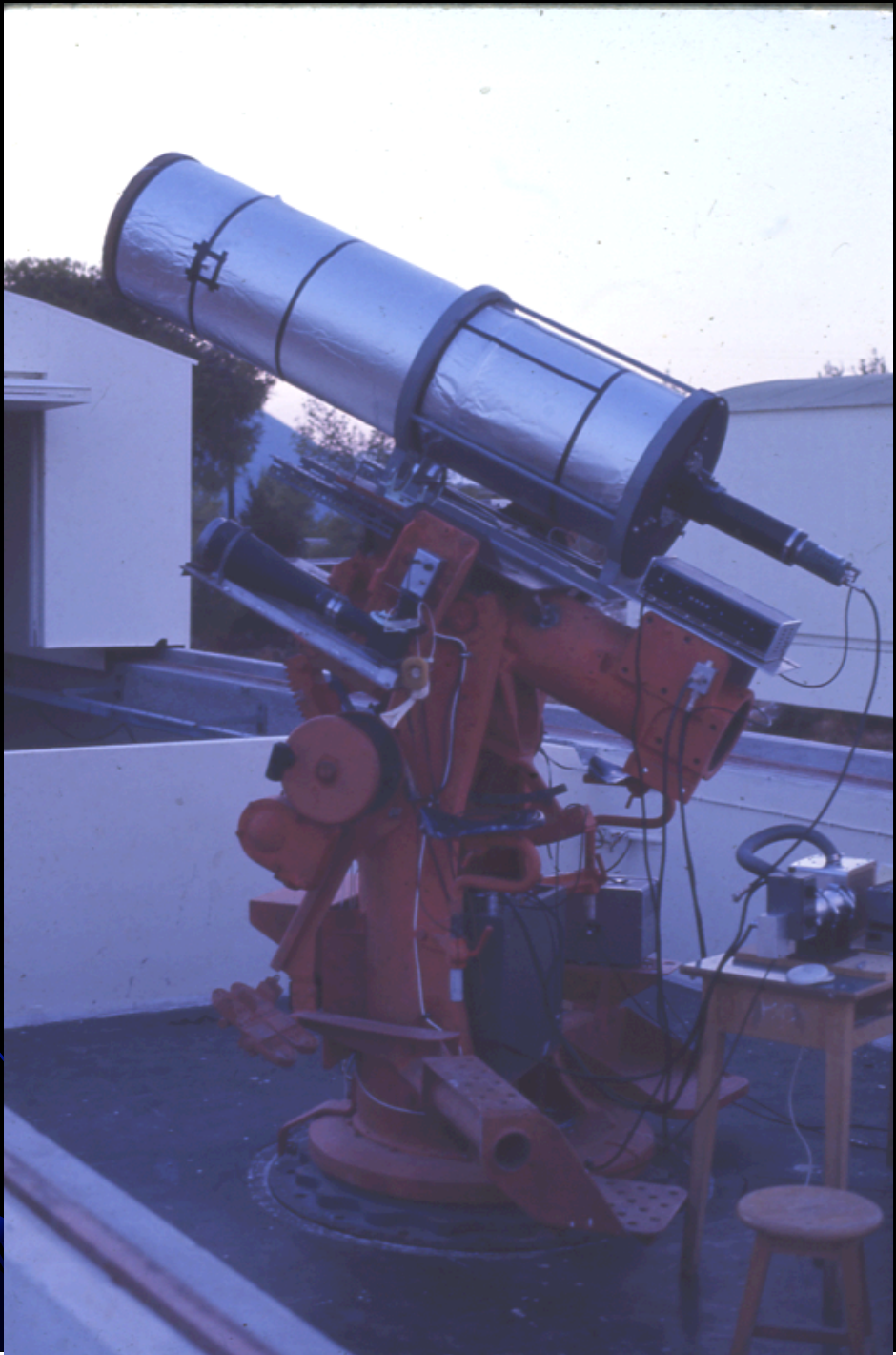
Source	$J_2 \times 10^6$	$J_3 \times 10^6$	$J_4 \times 10^6$	$J_5 \times 10^6$	$J_6 \times 10^6$	$J_7 \times 10^6$	$J_9 \times 10^6$
[6]	1082.79 ± 0.15		-1.4 ± 0.2		0.9 ± 0.8		
[4]	1082.21 ± 0.04	-2.29 ± 0.03	-2.10 ± 0.06	-0.23 ± 0.03			
[15]	1082.48 ± 0.04	-2.56 ± 0.01	-1.84 ± 0.09	-0.06 ± 0.01	0.39 ± 0.09	-0.47 ± 0.01	0.12 ± 0.01
[17]		-2.42 ± 0.10		-0.22 ± 0.07		-0.27 ± 0.07	
[5]	1082.49 ± 0.06	-2.39 ± 0.26	-1.70 ± 0.06	-0.30 ± 0.53			
[11]	1082.61 ± 0.05		-1.52 ± 0.08		0.73 ± 0.10		
[14.18]	1083.15 ± 0.20	-2.37 ± 0.18	-1.4 ± 0.3	-0.05 ± 0.15	0.7 ± 0.6		
[13]	1083.3 ± 0.7	-2 ± 3	-4.1 ± 0.7				

Y.K. 1962

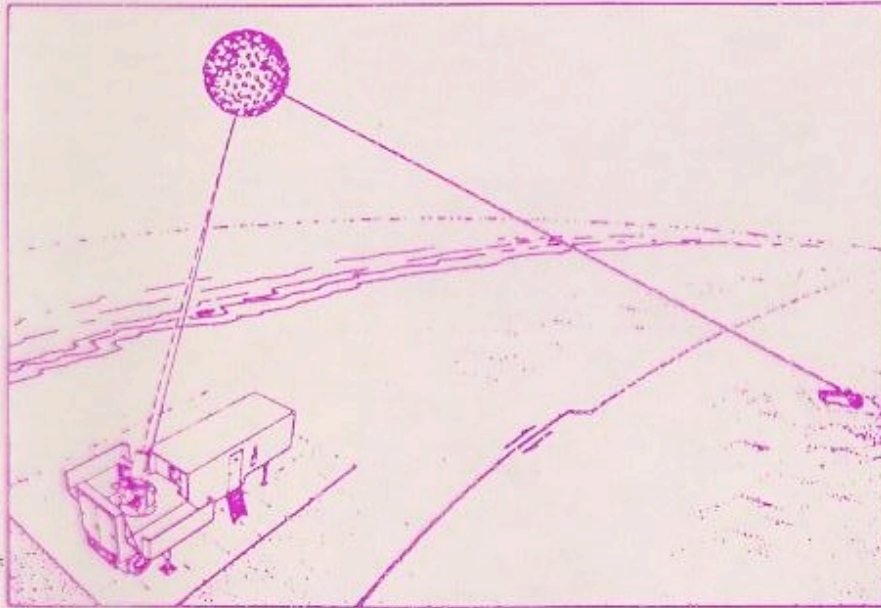
	Station	x^1	x^2	x^3		Number of observations
9001	Organ Pass	-1.535713	-5.167030	+3.401099	±15	5 131
9002	Olifantsfontein	+5.056137	+2.716534	-2.775806	10	5 922
9003	Woomera	-3.983618	+3.743212	-3.275642	7	7 257
9004	San Fernando	+5.105602	-0.555230	+3.769708	20	2 715
9005	Tokyo	-3.946563	+3.366400	+3.698878	18	2 459
9006	Naini Tal	+1.018190	+5.471170	+3.109601	30	1 799
9007	Arequipa	+1.942755	-5.804100	-1.796895	15	2 976
9008	Shiraz	+3.376916	+4.404028	+3.136311	20	2 733
9009	Curaçao	+2.251790	-5.816950	+1.327212	10	3 092
9010	Jupiter	+0.976314	-5.601416	+2.880301	12	3 607
9011	Villa Dolores	+2.280624	-4.914540	-3.355451	12	4 514
9012	Maui	-5.466100	-2.404157	+2.242353	22	4 330

G.V. 1964



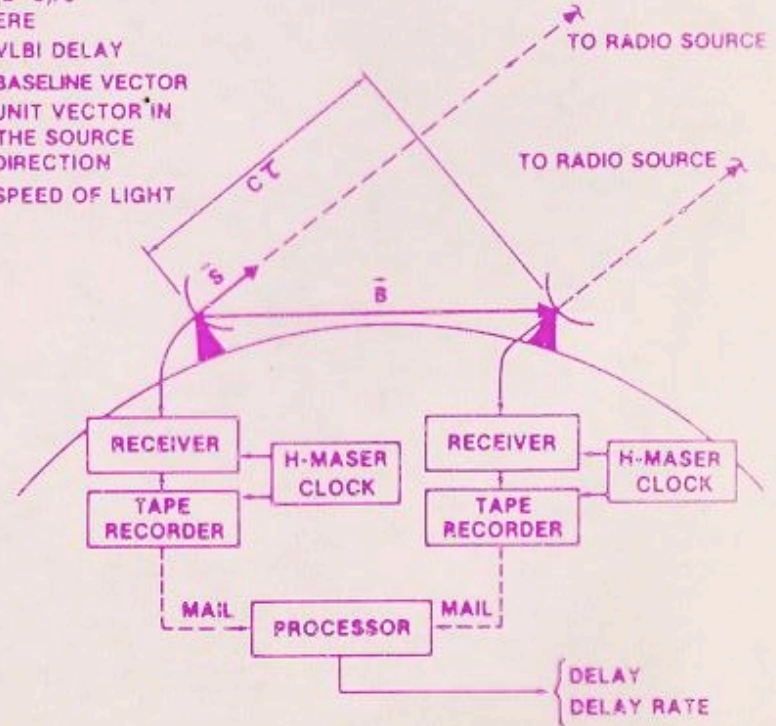


BASELINES BETWEEN LASER TRACKING STATIONS AND BETWEEN VLBI STATIONS ARE TO BE MEASURED TO A PRECISION OF A FEW CENTIMETERS



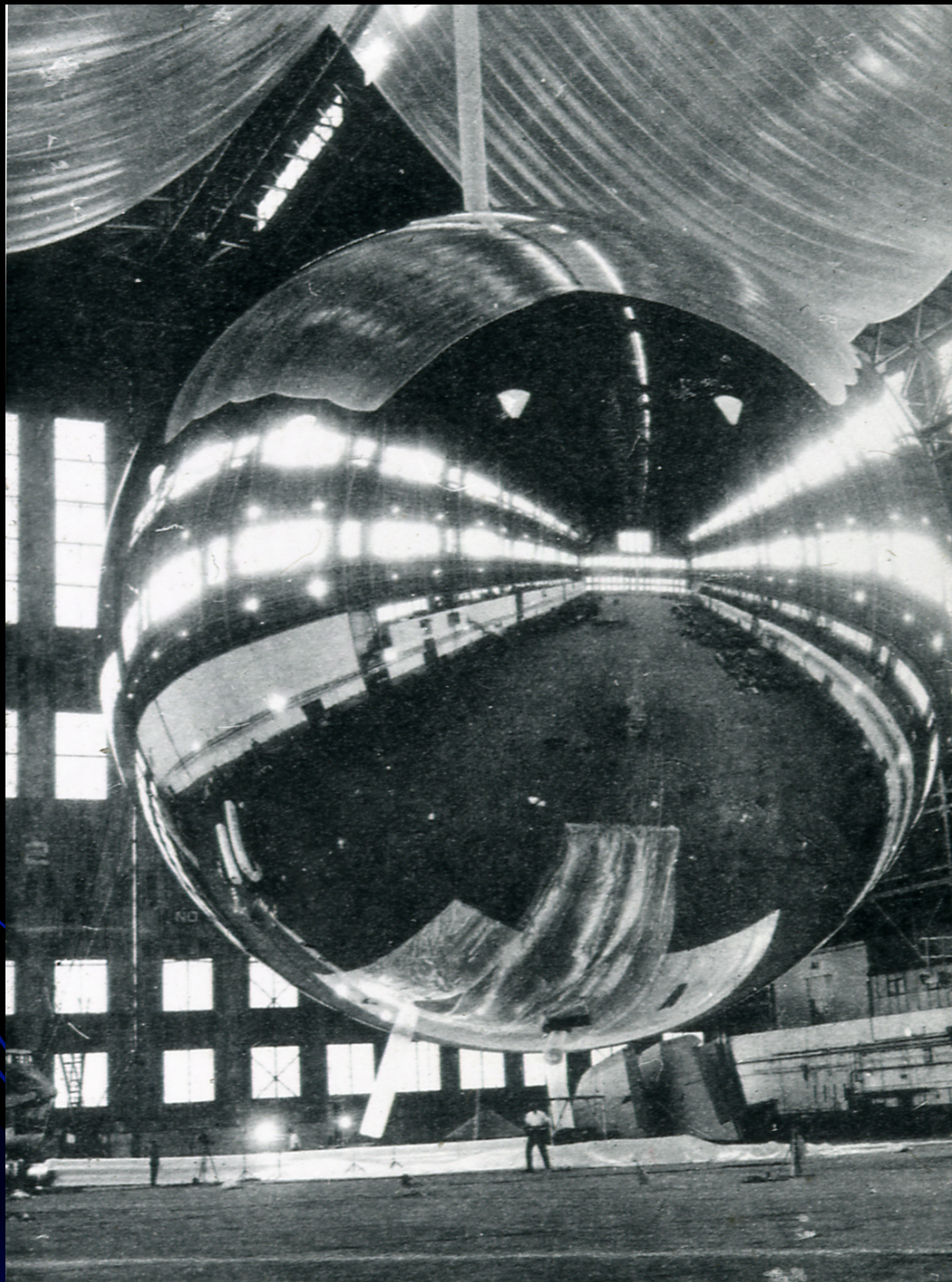
SATELLITE LASER RANGING

$\tau = (\vec{B} \cdot \vec{S})/c$
 WHERE
 τ = VLBI DELAY
 \vec{B} = BASELINE VECTOR
 \vec{S} = UNIT VECTOR IN THE SOURCE DIRECTION
 c = SPEED OF LIGHT



VERY LONG BASELINE INTERFEROMETRY





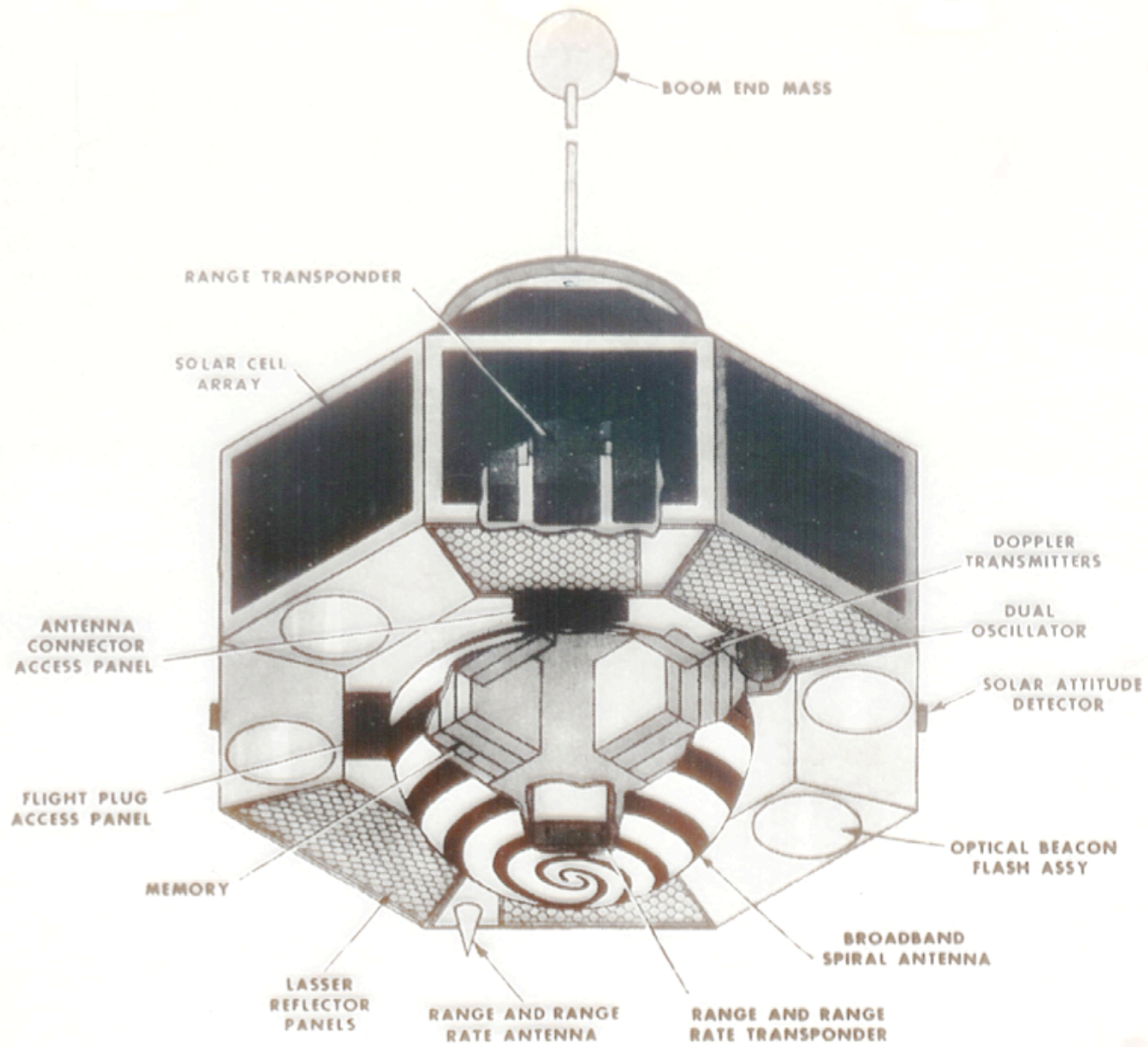
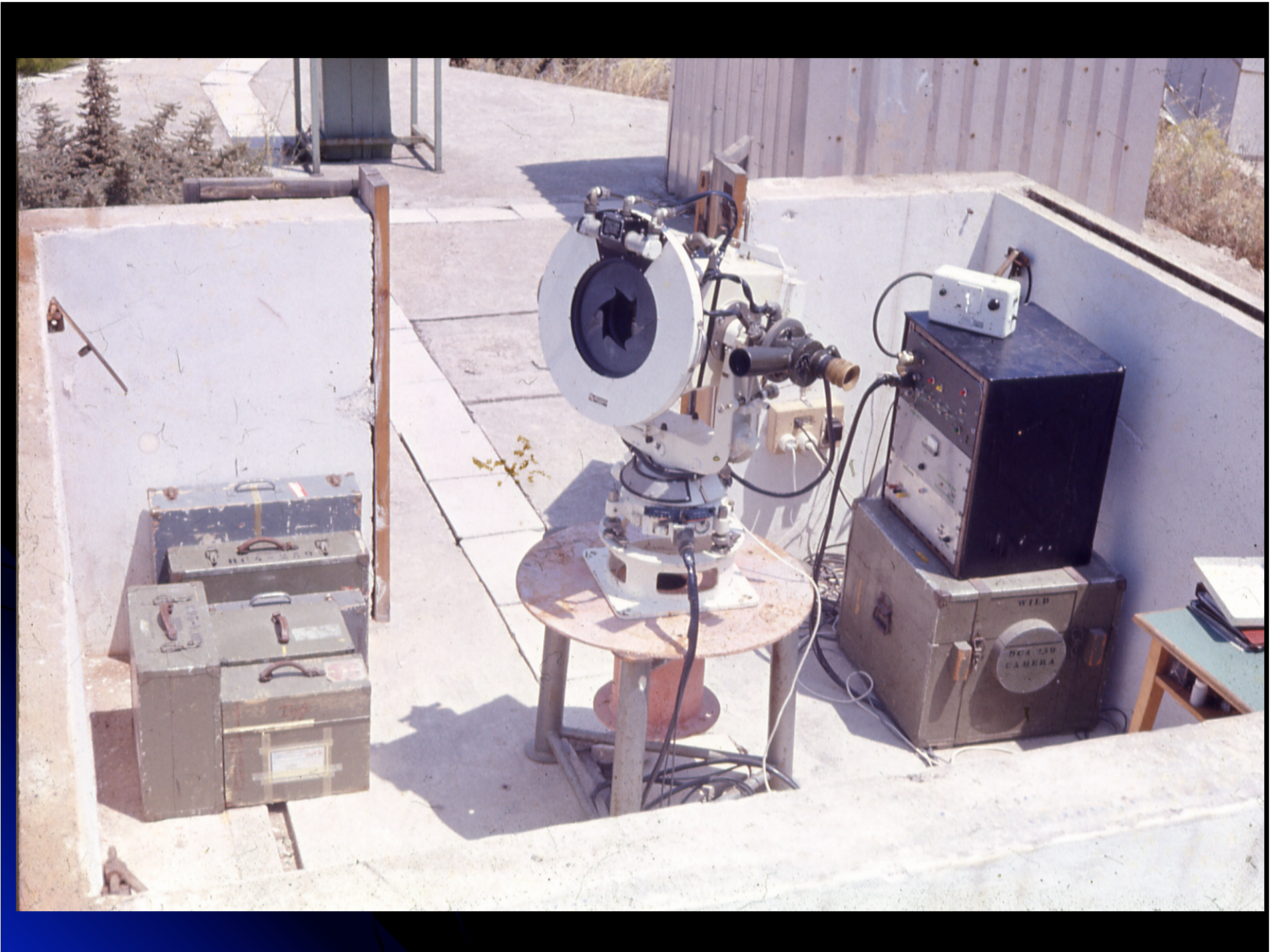
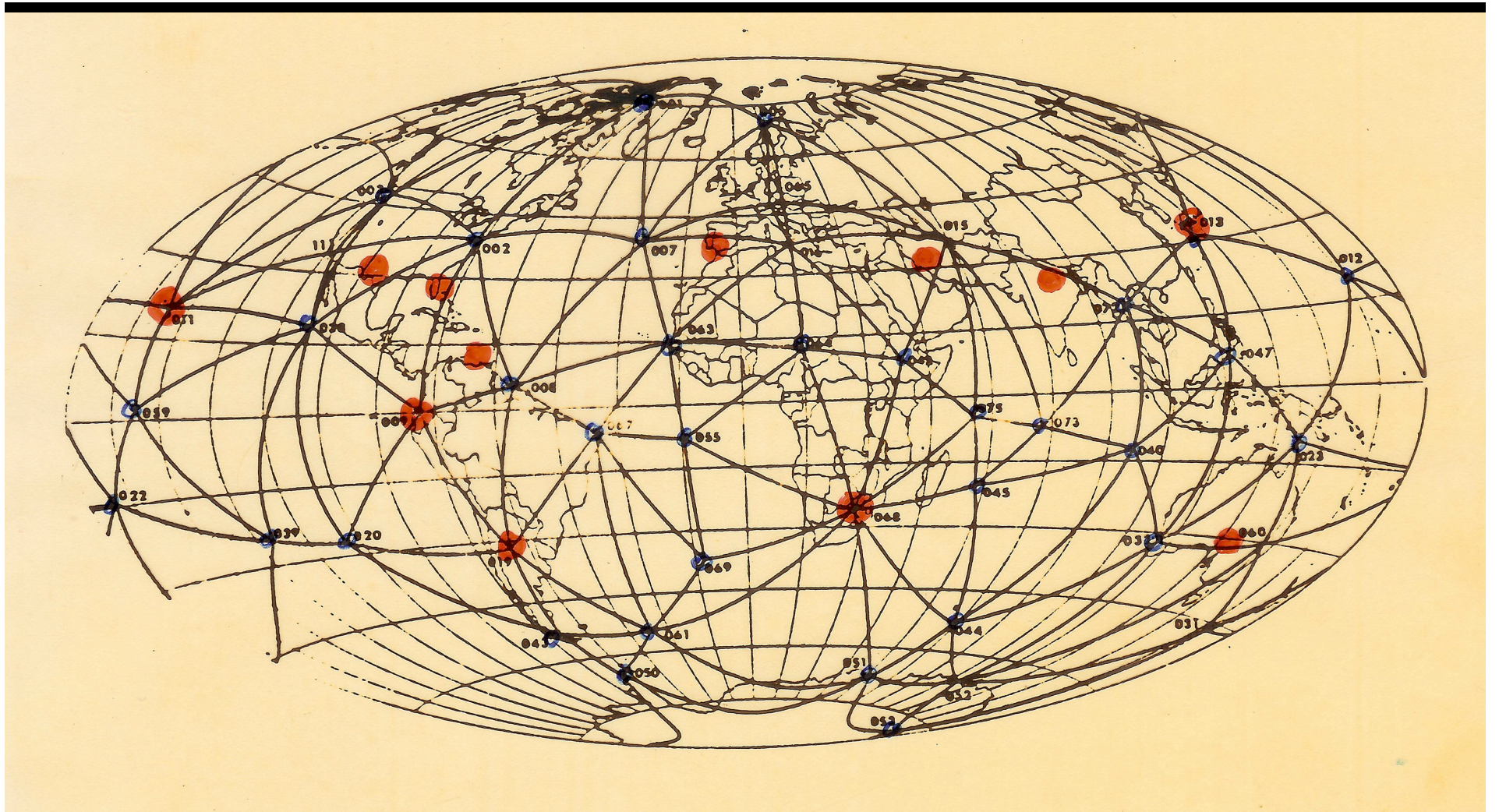


Figure 2.



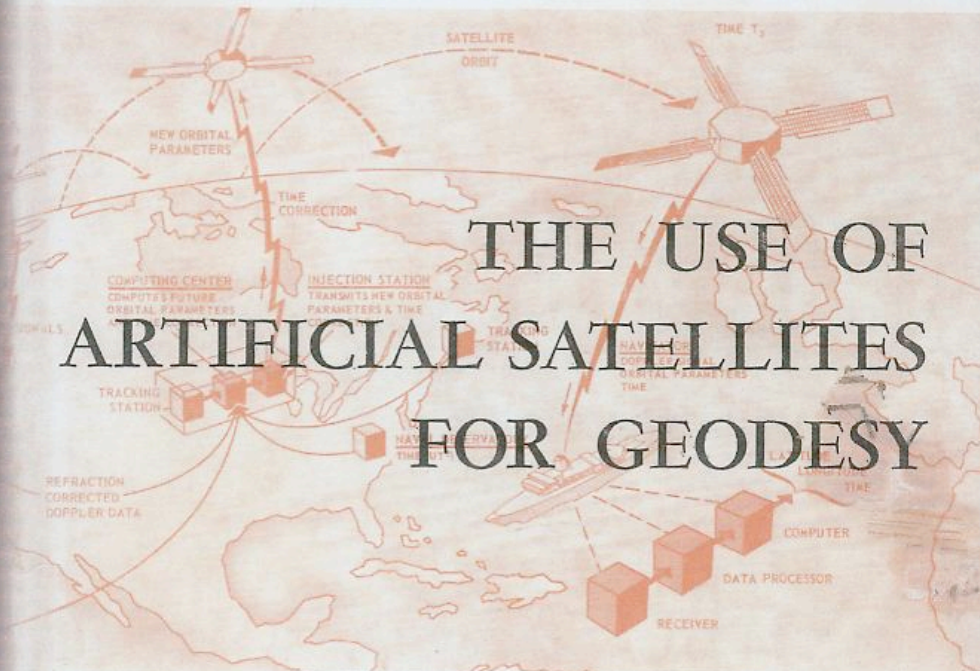




● BC 4

● B/N

Editor G. VEIS



THE USE OF ARTIFICIAL SATELLITES FOR GEODESY

PROCEEDINGS OF THE FIRST
INTERNATIONAL SYMPOSIUM ON THE USE
OF ARTIFICIAL SATELLITES FOR GEODESY

WASHINGTON, D.C., 1962

NORTH-HOLLAND PUBLISHING COMPANY

G. VEIS
Editor

THE USE OF ARTIFICIAL SATELLITES FOR GEODESY

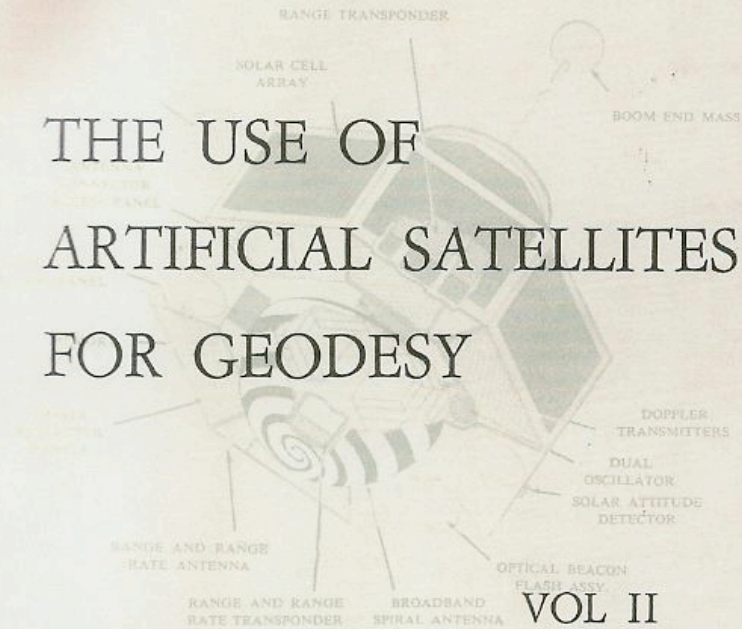
II



N. T. U.

PUBLICATION OF THE NATIONAL TECHNICAL UNIVERSITY OF ATHENS

THE USE OF ARTIFICIAL SATELLITES FOR GEODESY



VOL II

PROCEEDINGS OF THE SECOND INTERNATIONAL SYMPOSIUM ON
THE USE OF ARTIFICIAL SATELLITES FOR GEODESY

ATHENS, GREECE, 1965

EDITED BY GEORGE VEIS

Research in Space Science

SAO Special Report No. 200

GEODETTIC PARAMETERS FOR A
1966 SMITHSONIAN INSTITUTION STANDARD EARTH

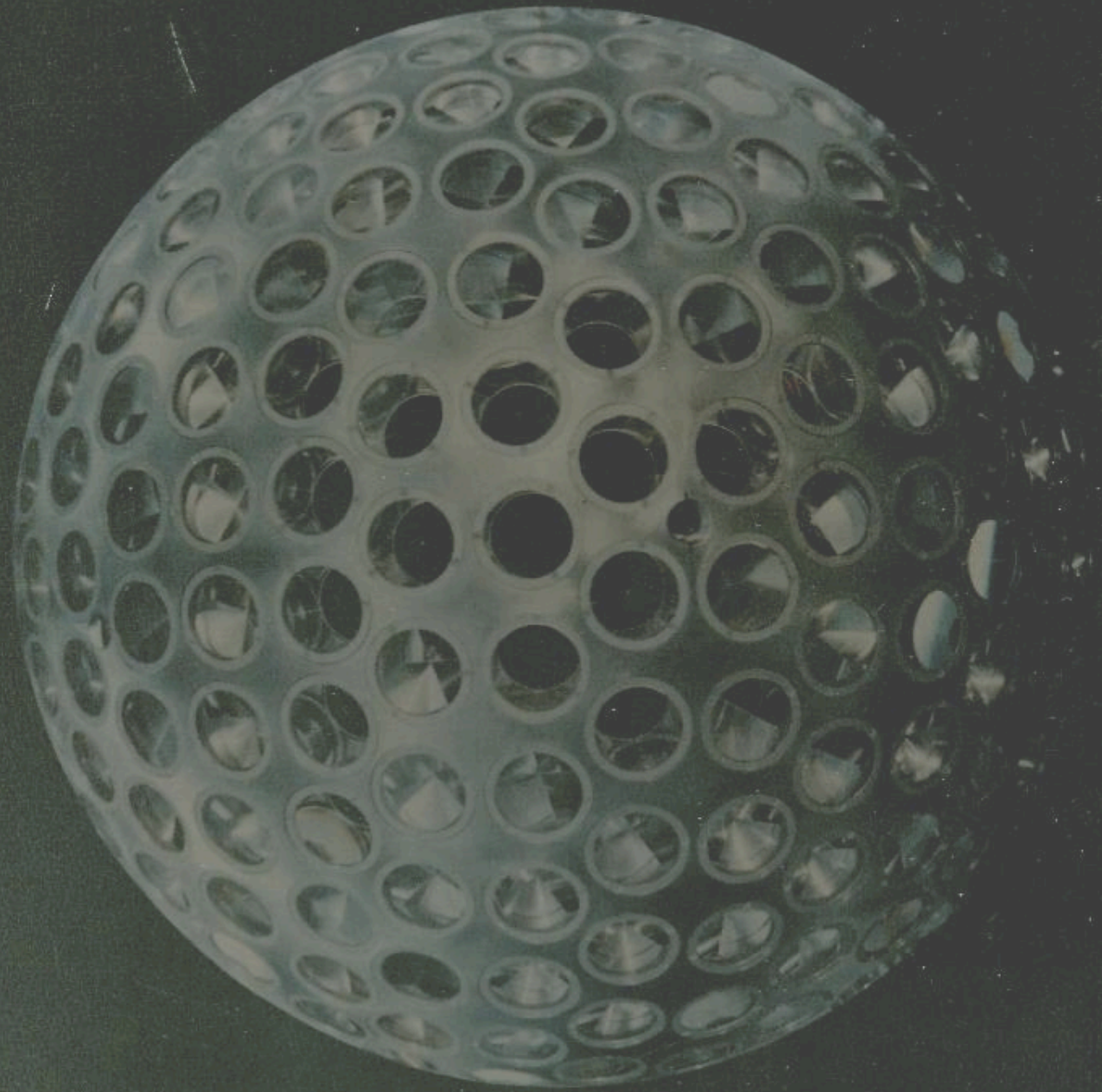
Volume 1

edited by

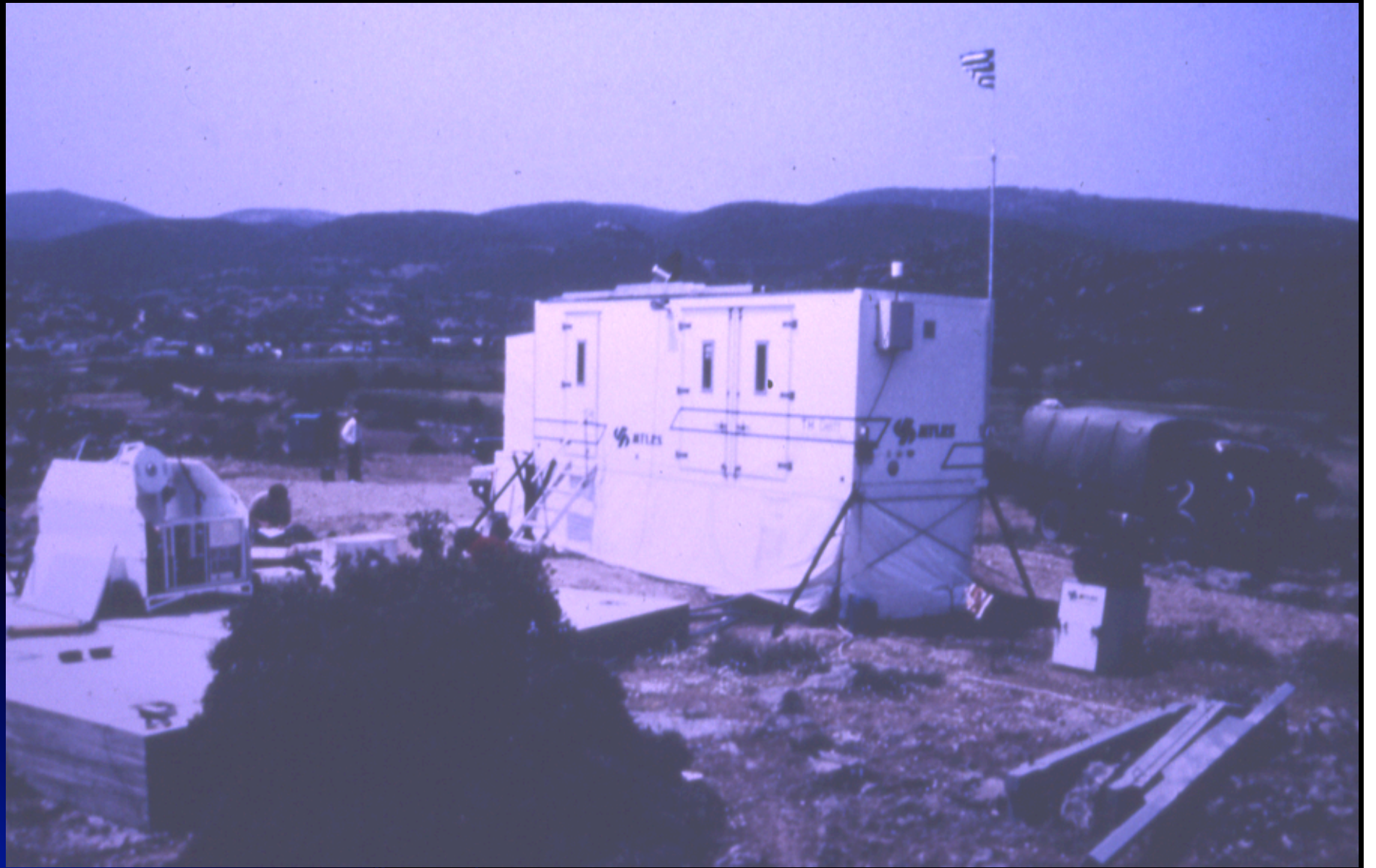
Charles A. Lundquist and George Veis

1966

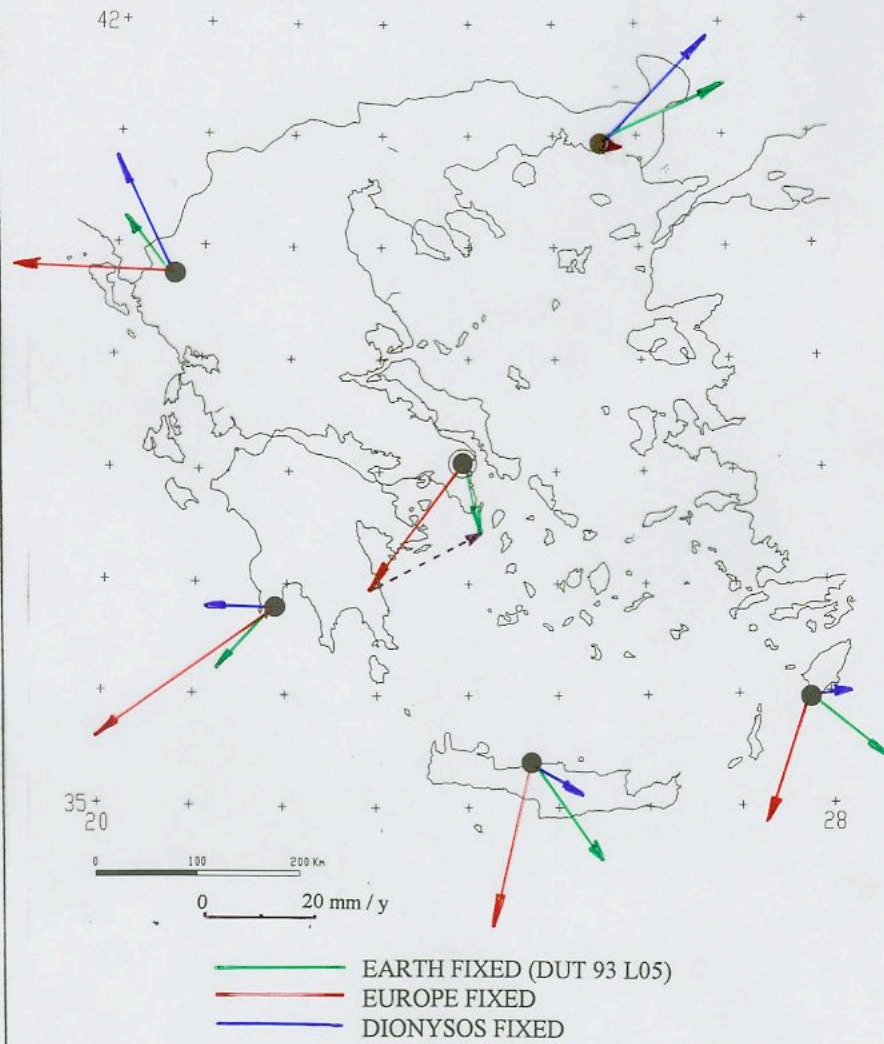
Smithsonian Institution
Astrophysical Observatory
Cambridge, Massachusetts, 02138

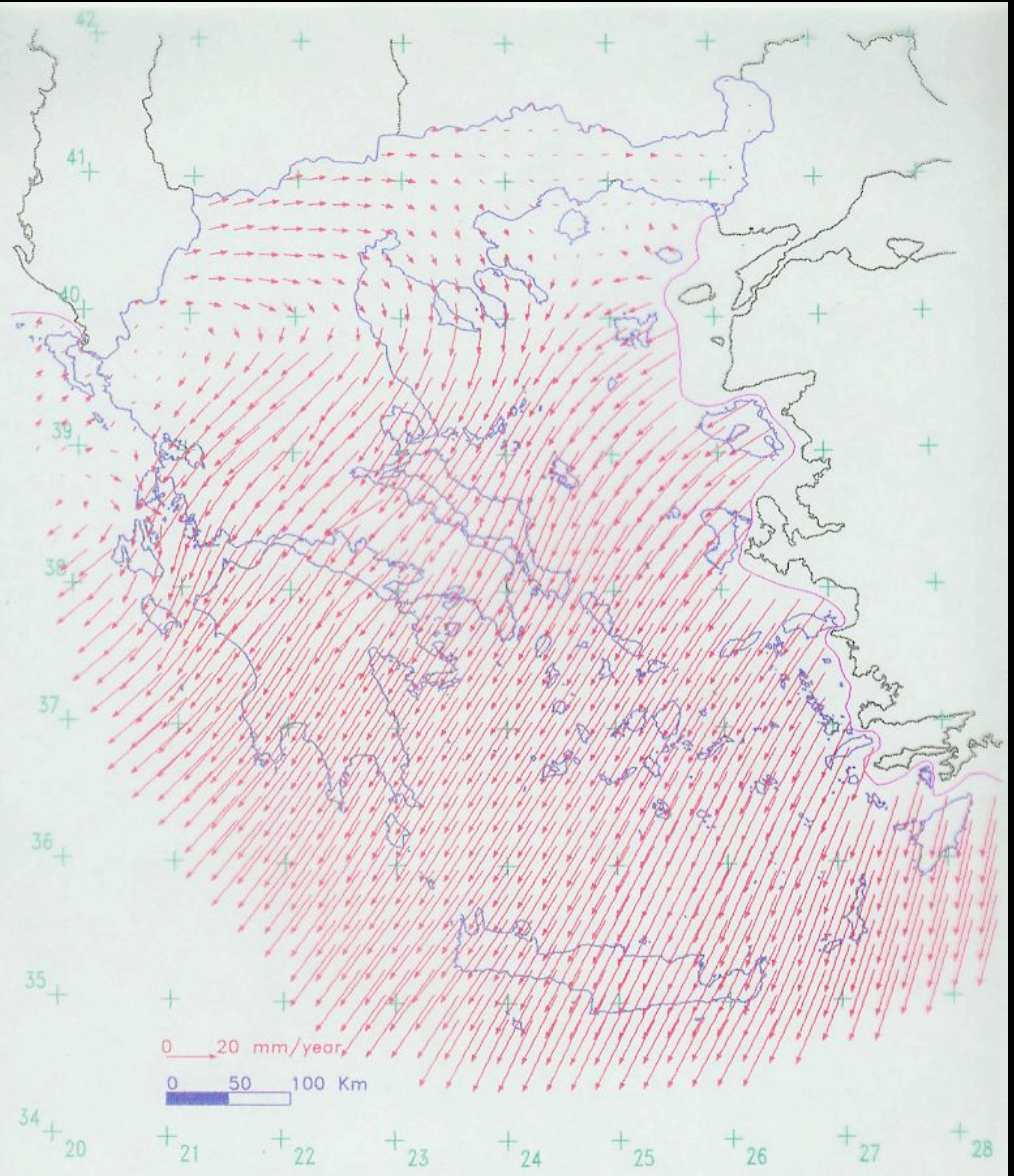






SLR STATION VELOCITIES



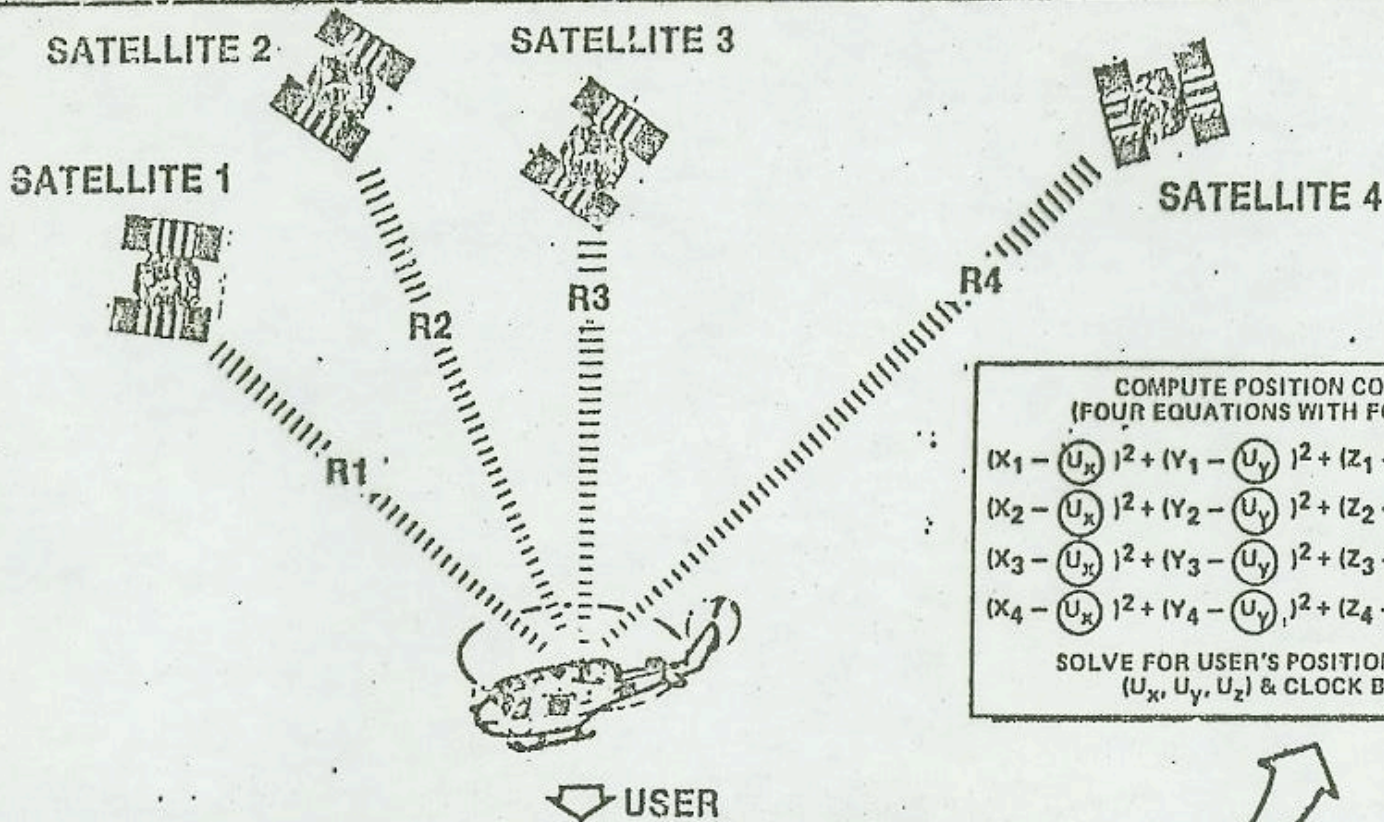


VELOCITIES WITH FIXED EUROPE



NAVIGATING WITH THE GPS

FIGURE 3: NAVIGATING WITH GPS



COMPUTE POSITION COORDINATES
(FOUR EQUATIONS WITH FOUR UNKNOWNNS)

$$(X_1 - U_x)^2 + (Y_1 - U_y)^2 + (Z_1 - U_z)^2 = (R_1 - C_B)^2$$

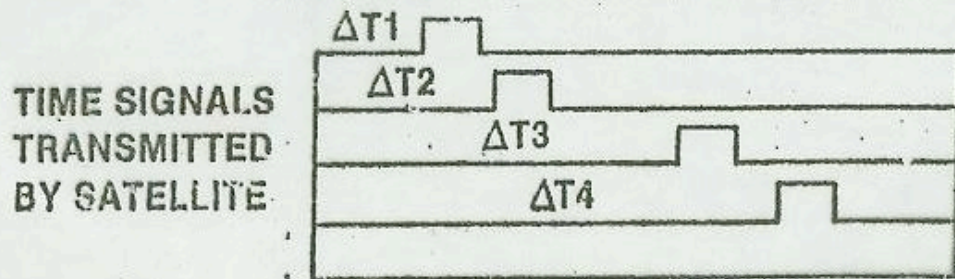
$$(X_2 - U_x)^2 + (Y_2 - U_y)^2 + (Z_2 - U_z)^2 = (R_2 - C_B)^2$$

$$(X_3 - U_x)^2 + (Y_3 - U_y)^2 + (Z_3 - U_z)^2 = (R_3 - C_B)^2$$

$$(X_4 - U_x)^2 + (Y_4 - U_y)^2 + (Z_4 - U_z)^2 = (R_4 - C_B)^2$$

SOLVE FOR USER'S POSITION COORDINATES
(U_x, U_y, U_z) & CLOCK BIAS (C_B)

COMPUTE FOUR PSEUDO-RANGE VALUES



$$R_1 = C \times T_1$$

$$R_2 = C \times T_2$$

$$R_3 = C \times T_3$$

$$R_4 = C \times T_4$$

(C = SPEED OF LIGHT)

GPS PROGRAM SCHEDULE

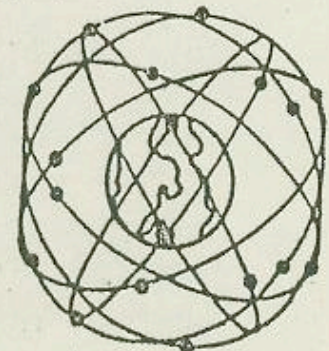
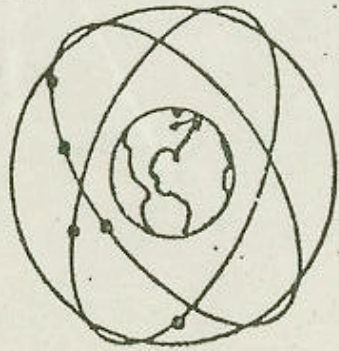
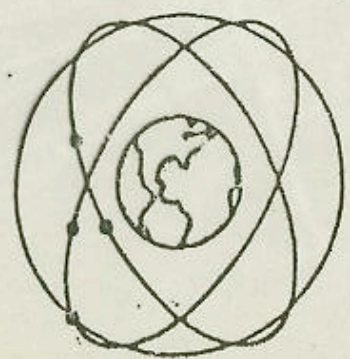
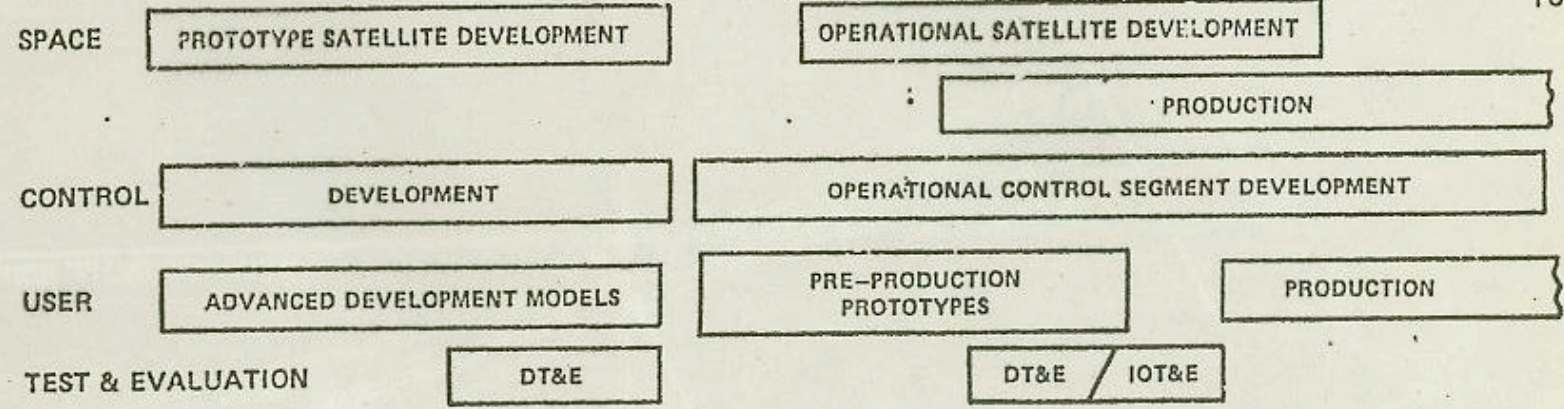
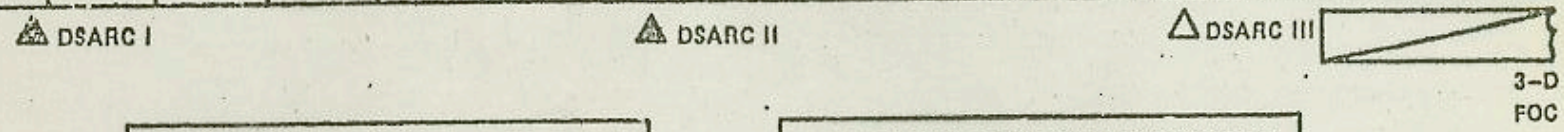
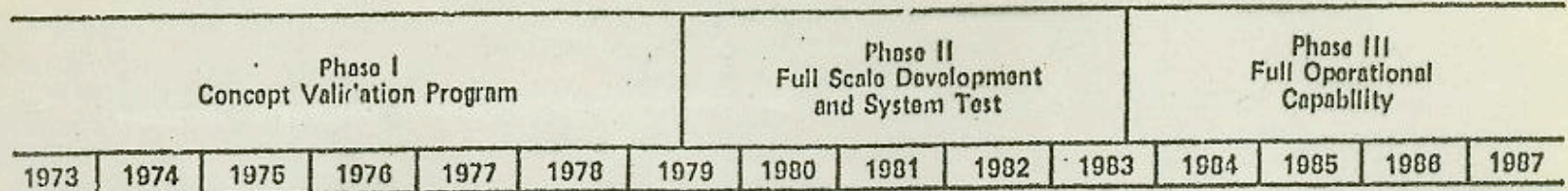
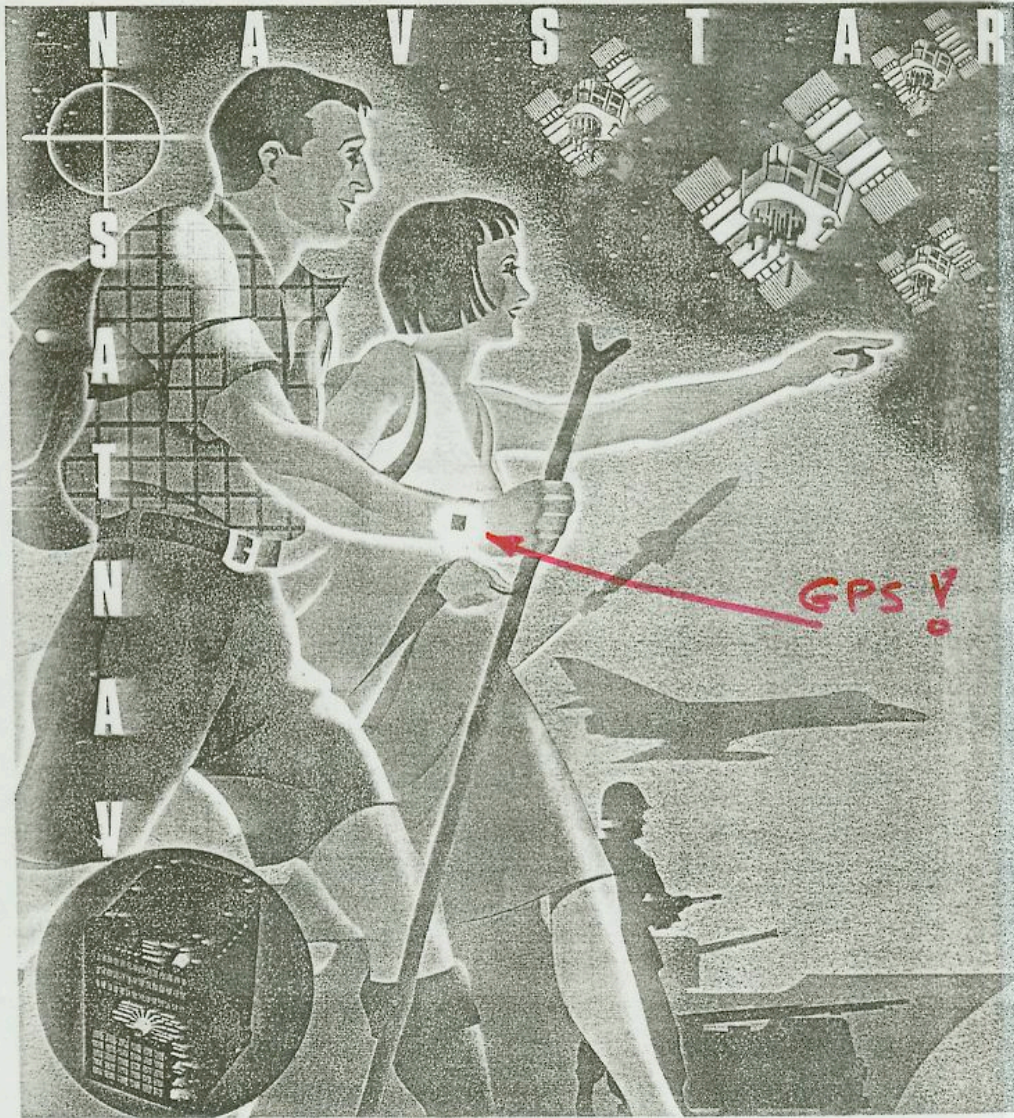


FIGURE 5: GPS PROGRAM SCHEDULE



newscientist



Follow that satellite

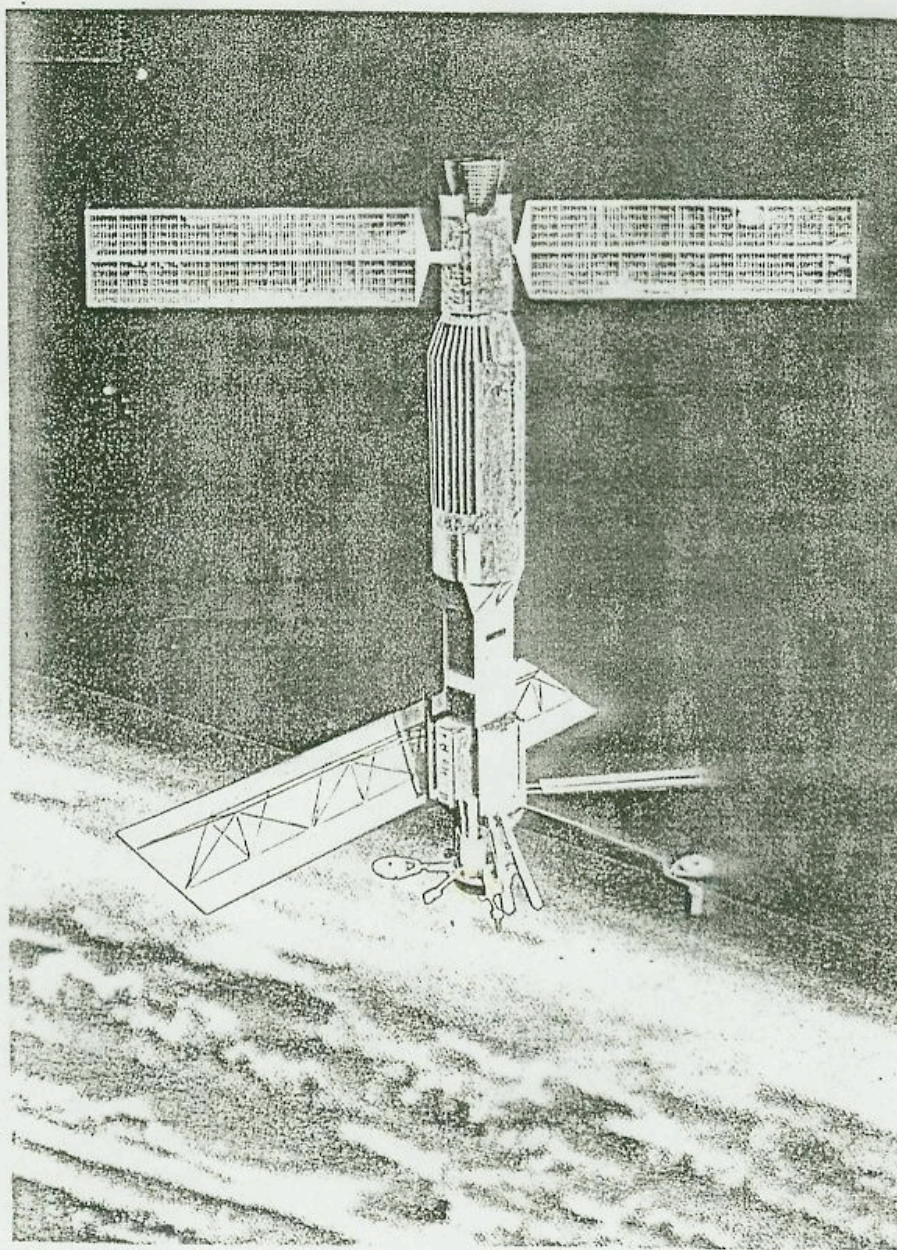


Figure 3. The first dedicated oceanographic satellite, Seasat, operated for three months in 1978 and demonstrated the feasibility of measuring surface winds, waves, sea level changes, and the marine geoid.

