

Using LLR Data for Testing General Relativity with Focus on the Equivalence Principle

Jürgen Müller, Liliane Biskupek, Vishwa Singh*, Mingyue Zhang

Institute of Geodesy, Leibniz University Hannover

*now at GFZ Potsdam, Department of Geodesy

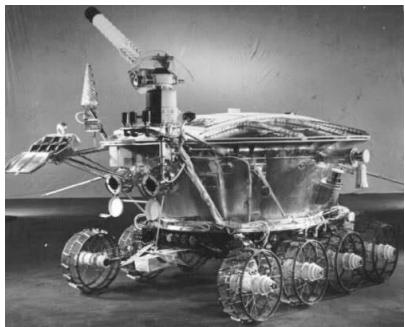
Determination of „Relativistic“ Parameters with LLR

Strategy: Newtonian effects modelled accurately enough or determined simultaneously

Estimation of selected relativistic parameters

- metric parameters (e.g. space time non-linearity, preferred frames)
- **test of the equivalence principle (Nordtvedt effect, dark matter)**
- **time-variable gravitational constant**
- $1/r^2$ -law (Yukawa terms)
- relativistic rotations (e.g. geodetic precession)

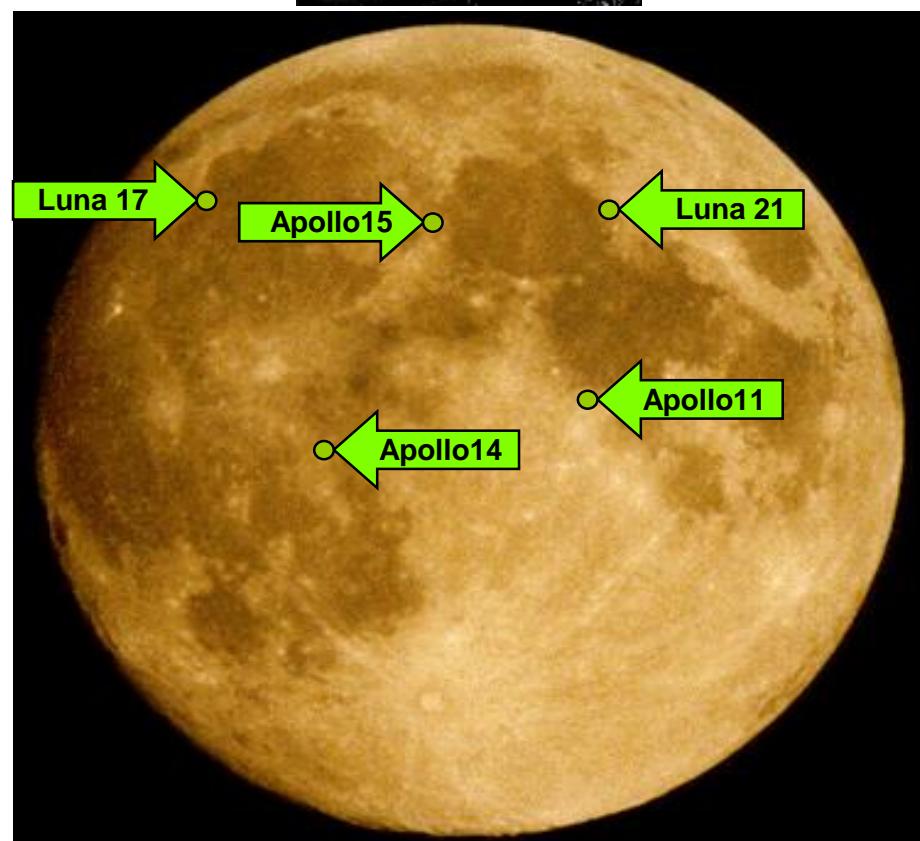
Retro-reflectors on the Moon



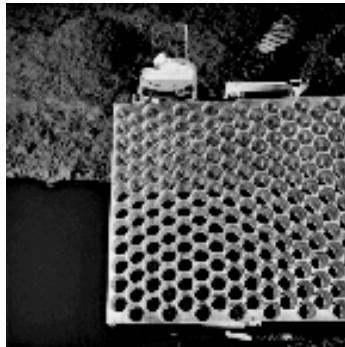
Luna 17



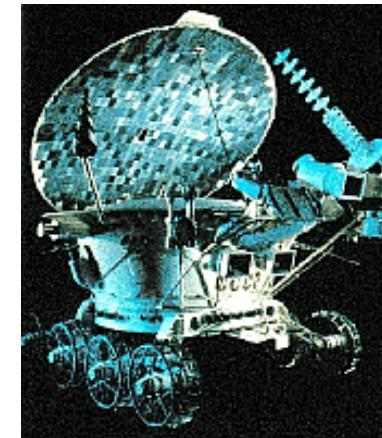
Apollo 14



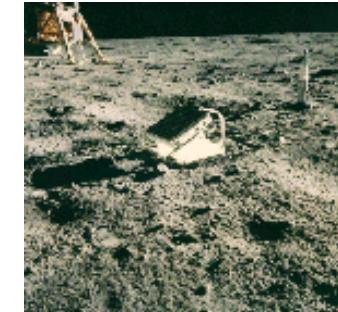
deployed 1969-1973 (more to come soon ...)



Apollo 15

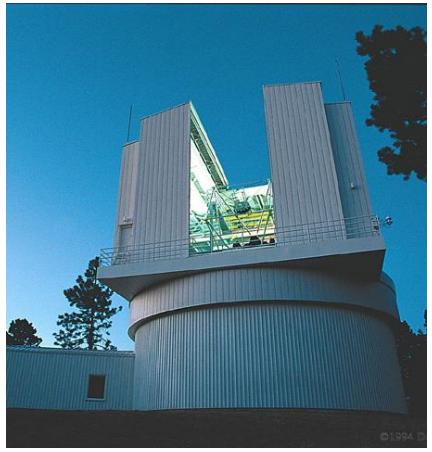


Luna 21



Apollo 11

Lunar Laser Ranging Observatories on Earth



APOULLO, 3.5 m



McDonald



Grasse

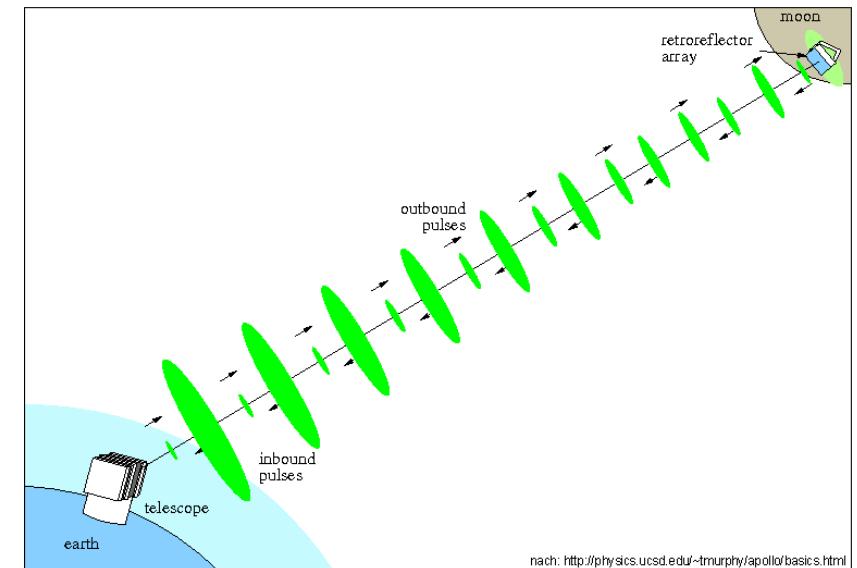


Wettzell

more sites started lunar tracking or may join soon

LLR – Measurement

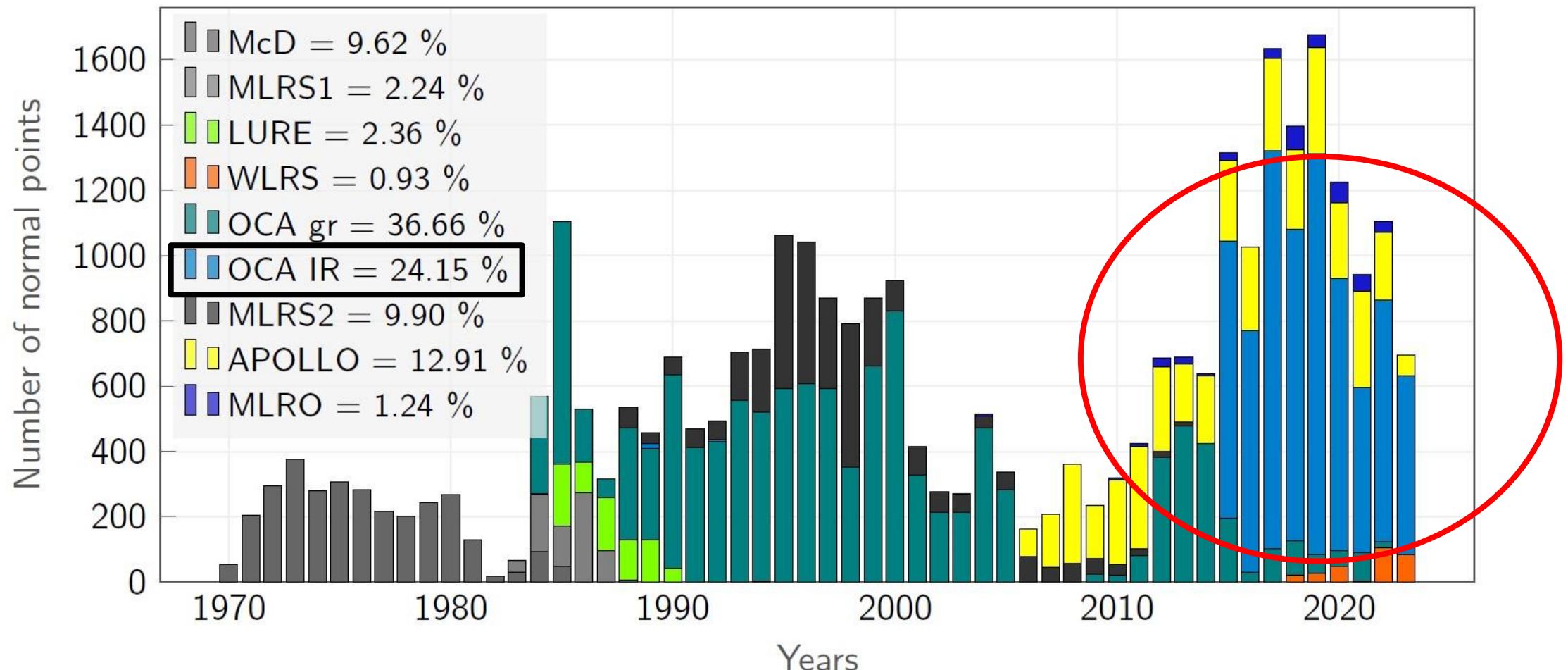
- telescope, diameter 0.7 m – 3.5 m
- pulsed laser, pulse length 0.07 ns – 0.2 ns
- footprint at the Moon ~20 km² (reflector $\frac{1}{4}$ - $\frac{1}{2}$ m²)
- 10^{18} - 10^{19} photons/pulse sent, ~1 - 100 photons received
- filtering
 - spatial
 - spectral
 - temporal
- LLR data
 - 1- 15 min → 1 normal point
 - > 31,000 normal points in 53 years



nach: <http://physics.ucsd.edu/~tmurphy/apollo/basics.html>

New LLR Normal Points in Infrared

Since 2015 new LLR measurements in **infrared**, especially from Grasse

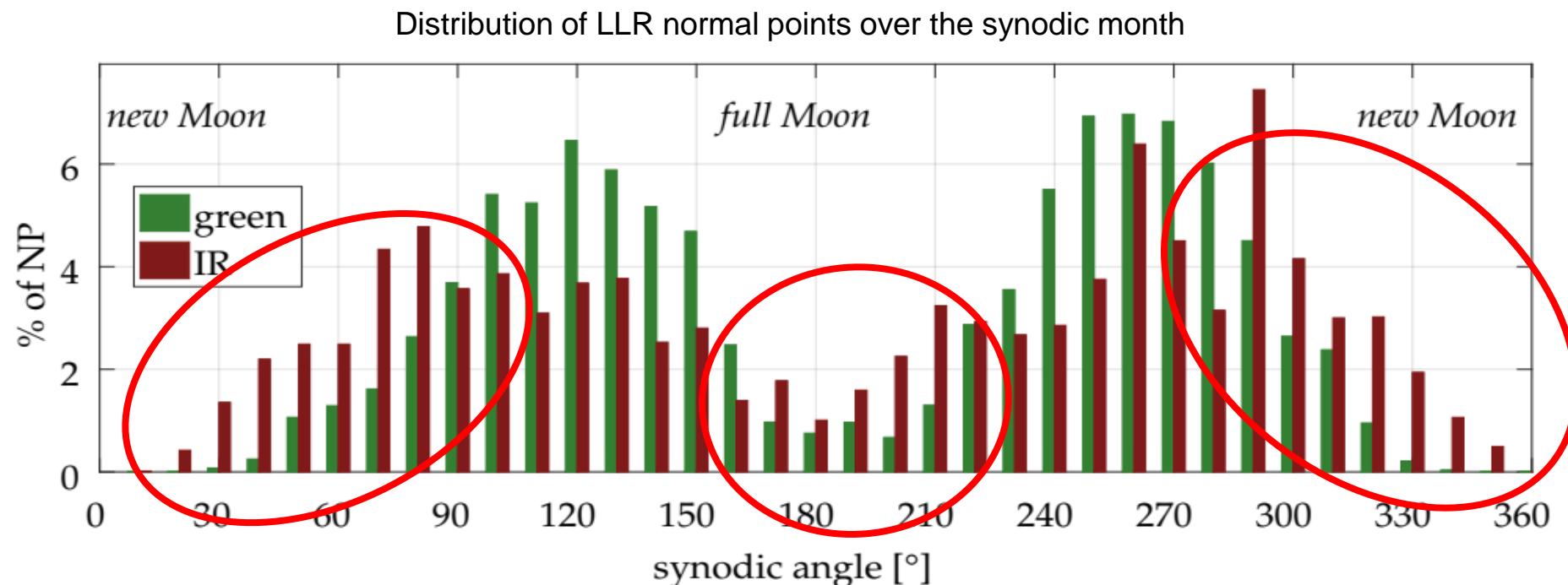


New LLR Normal Points in Infrared

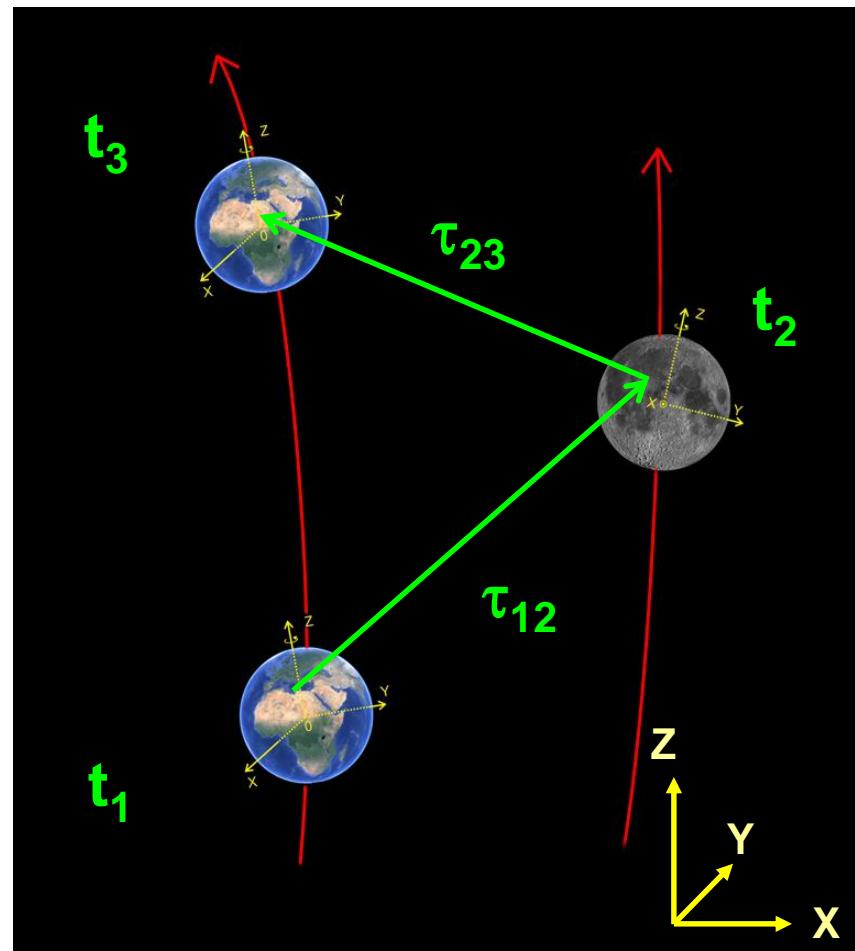
Since 2015 new LLR measurements in **infrared**

- more LLR observations
- higher accuracy
- better orbit and reflector coverage
- potentially more observatories

Very good for testing the equivalence principle (EP)



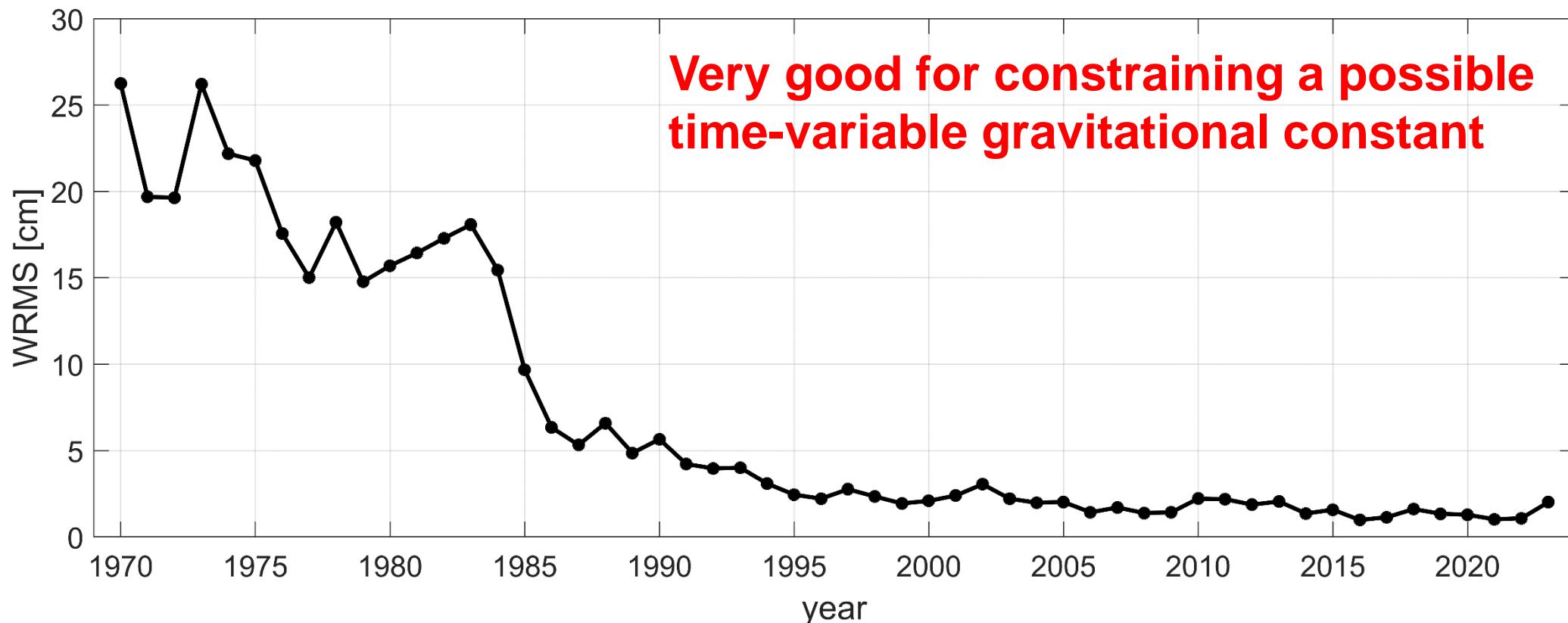
LLR – Analysis Model



- Earth/Moon are moving during measurement
 - Need accurate position/orientation at different times
 - Earth:
 - position from ephemeris
 - orientation from IERS Conv. 2010
 - Moon:
 - position from ephemeris
 - orientation from ephemeris
- Fully (relativistic) model also for
 - Relative change of stations on Earth and reflectors on the Moon
 - Signal propagation
 - ...

Weighted Annual LLR Residuals

weighted residuals (observed – computed Earth-Moon distance),
annually averaged – for 31,620 NP between April 1970 and July 2023



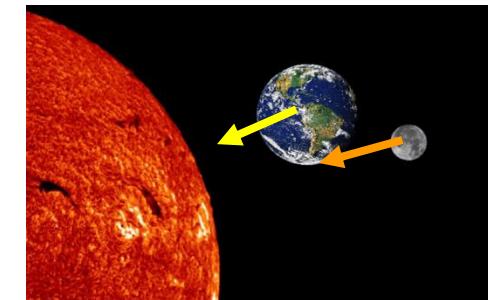
Model Refinement – for Relativistic Tests

Optional extension of the ephemeris model

- **time variable gravitational constant** $G = G_0 + \dot{G}\Delta t + \frac{1}{2}\ddot{G}\Delta t^2$
- geodetic precession of the lunar orbit in addition to EIH
- **violation of equivalence principle – 3 versions (see next page)**
- Yukawa term for modifying Newton's $1/r^2$ law of gravity
- preferred-frame effects and metric parameters (Will, 1993)
- gravitomagnetic effects (Soffel et al., 2008)
- optional spin-orbit coupling (Brumberg/Kopeikin)

Three Versions of the Equivalence Principle (EP)

- A. Equivalence of **inertial** and **gravitational mass** in the gravitational field of the **Sun**

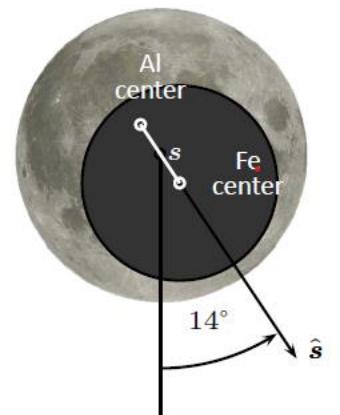


- B. Equivalence of **inertial** and **gravitational mass** with respect to assumed dark matter in the center of the **galaxy**



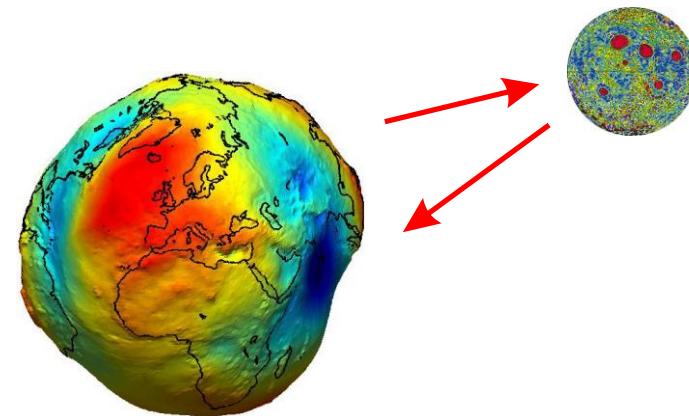
- C. Equivalence of **active** and **passive gravitational mass** of the **Moon**

... using Lunar Laser Ranging



Equation of Motion – EP Tests

$$m_i \ddot{r} \approx -\frac{G m_{pg} m_{ag}}{r_{EM}^2} \dots$$



EP tests

- Equivalence of inertial mass m_i and gravitational passive mass m_{pg} in the gravitational field of the
 - Sun
 - galactic center
- Equivalence of passive m_{pg} and active gravitational mass m_{ag}

A. Test of Strong Equivalence Principle

Earth–Moon system mainly sensitive to **EP test** in gravitational field of the **Sun**

- EP violation would lead to additional acceleration of the Moon w.r.t. Earth depending on

$$\Delta(m_g / m_i)_{EM} = (m_g / m_i)_{Earth} - (m_g / m_i)_{Moon}$$

- resulting in additional range term with synodic period
(scaling factor $S = -2.9 \times 10^{10}$ m)

$$\Delta r_{EM} = S \Delta(m_g / m_i)_{EM} \cos D$$

- EP test with LLR is a combination of weak and strong EP test, i.e. due to
(i) different compositions and (ii) gravitational self energy

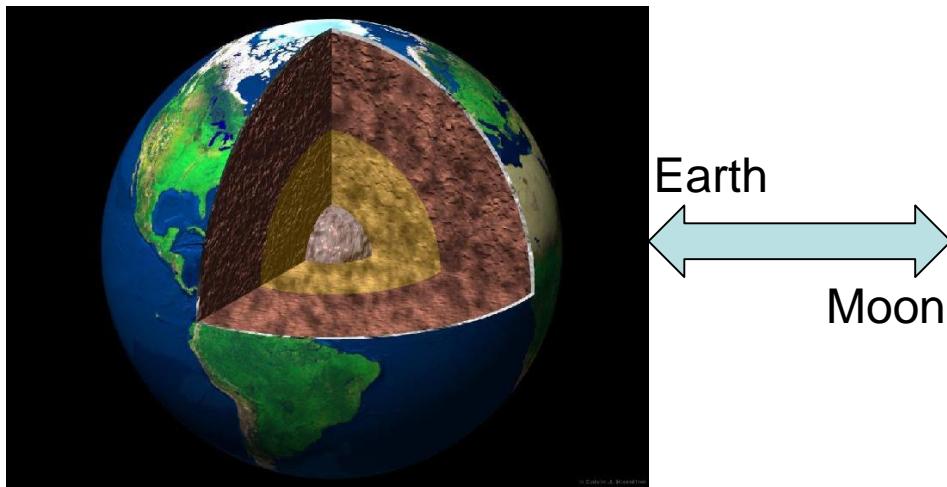
Test of Strong Equivalence Principle (2)

Earth and Moon have non-negligible amount of gravitational self energy

- strong form of EP can be tested
- if strong EP is violated, gravitational self energy itself affects gravitational acceleration
- ratio m_g/m_i can be expressed by

$$\frac{m_g}{m_i} = 1 + \eta \left(\frac{U}{Mc^2} \right)$$

η Nordtvedt parameter
 U gravitational self energy
 M mass of the body
 c speed of light



Test of Strong Equivalence Principle (3)

Introducing Earth/Moon values

$$\left(\frac{U}{Mc^2}\right)_{Earth} = -4.64 \times 10^{-10}$$

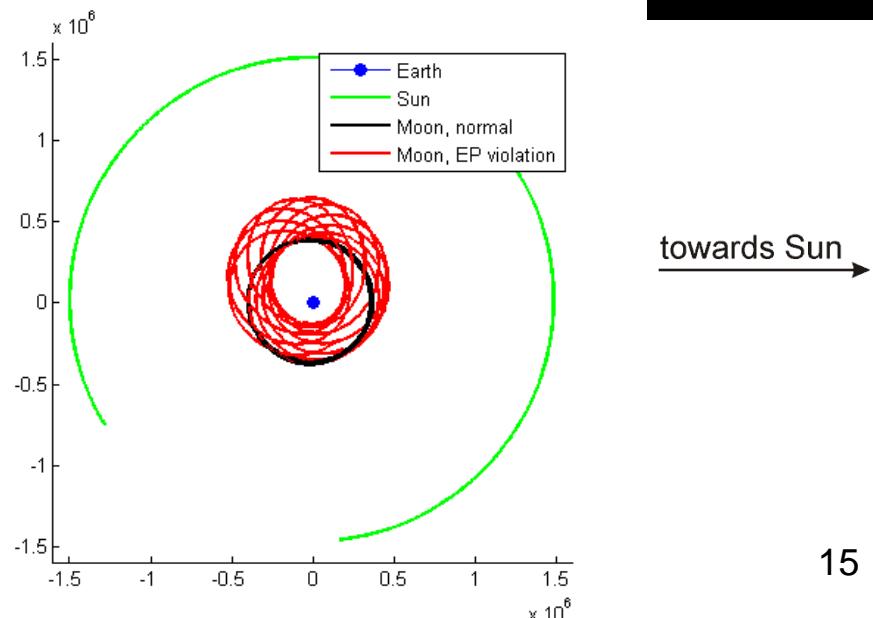
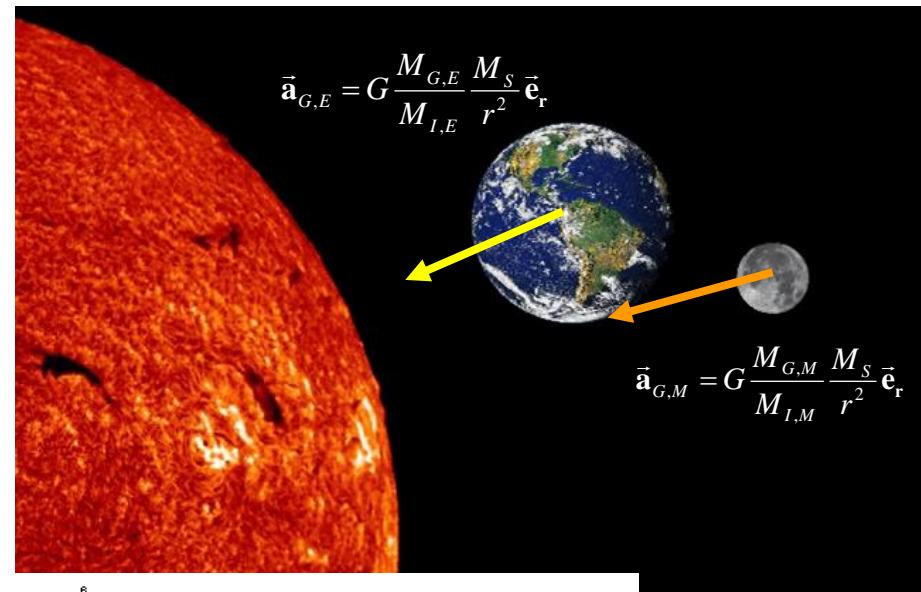
$$\left(\frac{U}{Mc^2}\right)_{Moon} = -1.90 \times 10^{-11}$$

gives $\Delta r_{EM} = 13.1 \text{ m } \eta \cos D$

If $\eta \neq 0$

- different accelerations of Earth and Moon
- polarisation of lunar orbit

Result: $\boxed{\eta = (-0.2 \pm 1.1) \times 10^{-4}}$



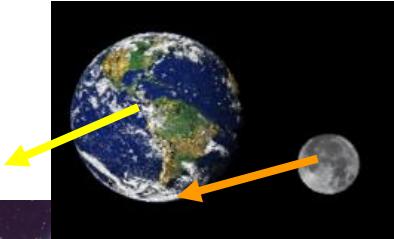
B. „Dark Matter“ EP Test – Galactic Center

- part of galactic matter is not observable by electromagnetic radiation
- strong hint of "dark matter" from discrepancy between observed and calculated orbit velocities of stars close to the center
- test of possible differential acceleration δg between dark matter and different compositions of Earth and Moon (Nordtvedt 1994)

$$\Delta r_{EM} = A \cos(\omega t - \Theta)$$

$$A = -\frac{3}{2} \frac{\delta g}{\omega(\omega - \omega_0)}$$

*A amplitude of in-plane component
 ω sidereal frequency
 Θ longitude of galactic center*



- LLR EP test by estimating A ,

$\delta A < 1$ mm, corresponding to

$$\delta g < 3 \times 10^{-15} \text{ cm/s}^2$$

C. Equivalence of Active and Passive Gravitational Mass

$$m_i \ddot{r} \approx -\frac{G m_{pg} m_{ag}}{r_{EM}^2} \dots$$

measurement of self-acceleration of center-of-mass

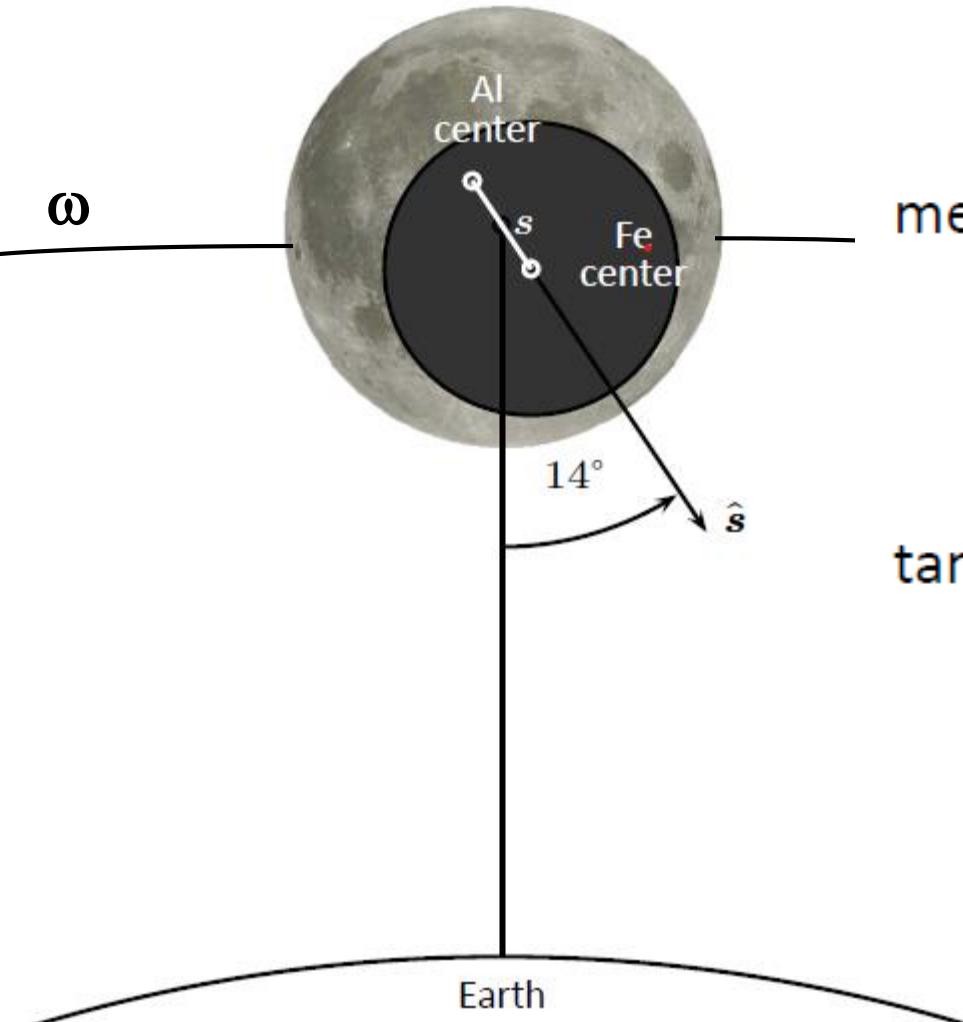
$$\frac{\mathbf{F}_{self}}{\mathbf{F}_{EM}} = C_{Al-Fe} \frac{M_M}{M_\oplus} \frac{r_{EM}^2}{r_M^2} \frac{s}{r_M} \frac{\rho}{\Delta\rho} \hat{\mathbf{s}}$$

Bartlett and van Buren, 1986

tangential component of force increases orbital velocity

$$\frac{\Delta\omega}{\omega} = 6\pi \frac{F_{self}}{F_{EM}} \sin 14^\circ \text{ per month}$$

Result: $C_{Al-Fe} = 4 \cdot 10^{-14}$



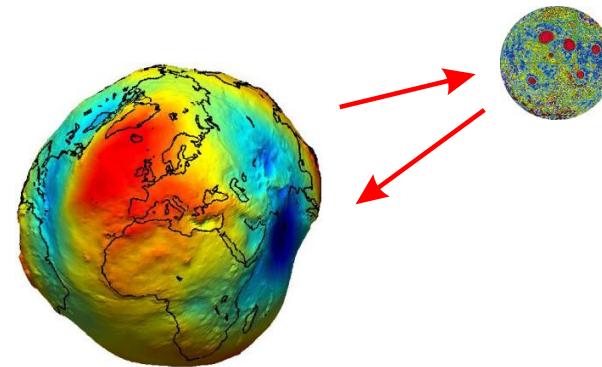
Variation of the Gravitational Constant G

Use ansatz

in equations of motion

$$G = G_0 + \dot{G}\Delta t + \frac{1}{2}\ddot{G}\Delta t^2$$

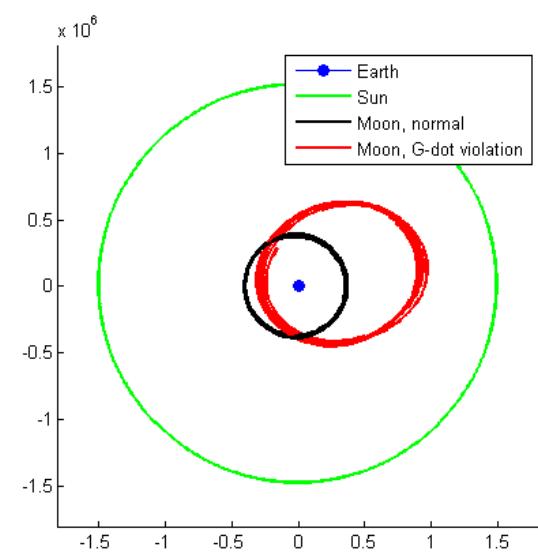
$$\ddot{r}_{EM} \approx -\frac{GM_{E+M}}{r_{EM}^2} \dots$$



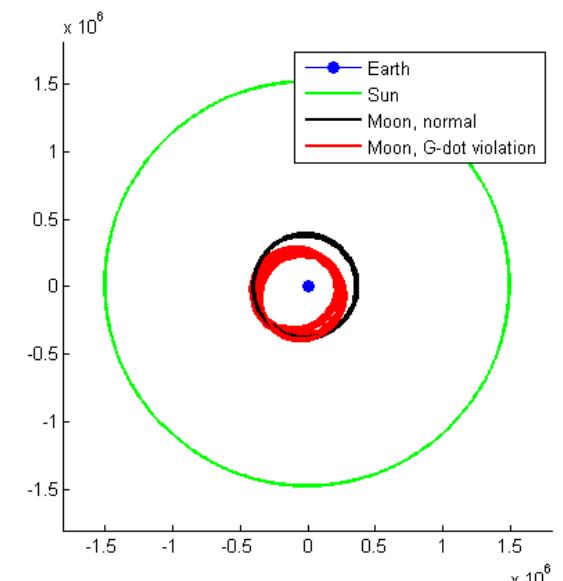
$$\frac{\dot{G}}{G} = (-5.0 \pm 9.6) \times 10^{-15} \text{ yr}^{-1}$$

$$\frac{\ddot{G}}{G} = (1.6 \pm 2.0) \times 10^{-16} \text{ yr}^{-2}$$

high correlation with $k_2\delta$



\dot{G} positive



\dot{G} negative

Results – Relativistic Quantities

Parameter	Results
Nordtvedt parameter η (test of the strong equivalence principle)	$(-0.2 \pm 1.1) \cdot 10^{-4}$
time variable gravitational constant (→ unification of the fundamental interactions)	$\dot{G}/G \text{ [yr}^{-1}]$ $\ddot{G}/G \text{ [yr}^{-2}]$
difference of geodetic precession $\Omega_{\text{GP}} - \Omega_{\text{deSIT}}$ ["/cy] (1.92 "/cy predicted by Einstein's theory of gravitation)	$(-5.6 \pm 8.5) \cdot 10^{-4}$
metric parameter $\gamma - 1$ (space curvature; $\gamma = 1$ in Einstein)	$(1.7 \pm 1.6) \cdot 10^{-4}$
metric parameter $\beta - 1$ (non-linearity; $\beta = 1$) or using $\eta = 4\beta - \gamma_{\text{Cassini}} - 3$ with $\gamma_{\text{Cassini}} - 1 \sim 10^{-5}$	$(6.2 \pm 7.2) \cdot 10^{-5}$ $(0.5 \pm 2.5) \cdot 10^{-5}$

Results – Relativistic Quantities (2)

Parameter	Results
Yukawa coupling constant $\alpha_{\lambda=400\,000\,\text{km}}$ (test of Newton's inverse square law for the Earth-Moon distance)	$(-3.7 \pm 4.5) \cdot 10^{-12}$
equivalence of passive and active gravitational mass $C_{\text{Al-Fe}}$	$4 \cdot 10^{-14}$
influence of dark matter $\delta_{gc} [\text{cm/s}^2]$ (in the center of the galaxy; test of strong equivalence principle)	$(2 \pm 3) \cdot 10^{-15}$
preferred frame effects α_1 α_2 (coupled with velocity of the solar system)	$(-1.1 \pm 1.5) \cdot 10^{-5}$ $(-6.0 \pm 9.0) \cdot 10^{-6}$
preferred frame effect α_1 (coupled with dynamics within the solar system)	$(6 \pm 6) \cdot 10^{-4}$

Conclusions

- LLR is a unique tool for studying the Earth-Moon system and testing general relativity
- Test of the **equivalence principle**, with respect to
 - **the Sun** $\eta = (-0.2 \pm 1.1) \times 10^{-4}$
 - **the galactic center** $\delta g \leq 3 \times 10^{-15} \text{ cm/s}^2$
 - **active/passive mass** $C_{Al-Fe} = 4 \cdot 10^{-14}$
- **Gravitational constant** $\dot{G}/G = (-5.0 \pm 9.6) \cdot 10^{-15} \text{ 1/ yr}$
- Only possible thanks to fantastic long-term lunar tracking by observatories (> 53 years of data)

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Studies by

L. Biskupek, F. Hofmann, E. Mai, V. Singh, M. Zhang

