

Satellite illumination for pointing and auto-tracking at Grasse station - France (ID7845)

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Observatoire
de la CÔTE d'AZUR

Outline

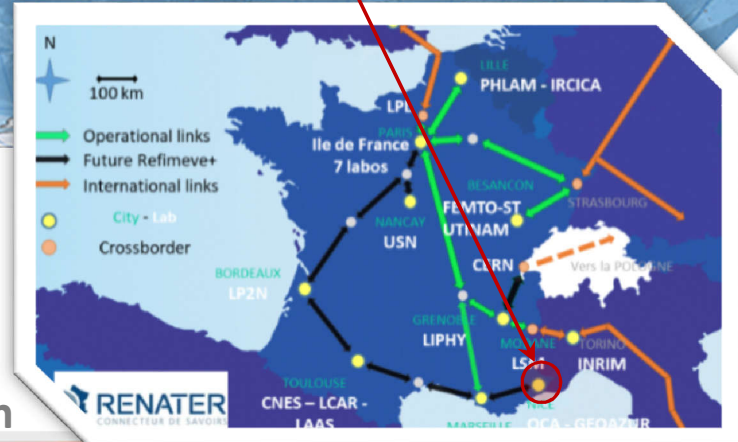
1. Experiment purposes
2. Architecture of experiment
3. Filter & centroid detection
4. Preliminary results
5. Conclusion & Future works

1. Satellite illumination – Experiment purposes

Grasse station ILRS7845 - Sciences

- ❑ **Satellite Laser Ranging**
(GNSS, Geodesy Satellites, Debris ...)
- ❑ **Lunar Laser Ranging**
(Moon Reflectors + LRO)
- ❑ **Time Transfer by Laser Link**
(T2L2, Chomptt, LRO, Hayabusa...)
- ❑ **Satellite – OGS LaserCom**
(SOTA, OPALS, OSIRIS, NorSatTD...)
- ❑ **QuantumCom demonstration**
- ❑ **Imaging / Astrometry**
(Adaptive Optics, Intensity Interferometry)
- ❑ **T/F transfer by Fiber network**
(T-Refimeve European fiber network)

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Grasse – France SLR & LLR station
Altitude = 1273 m,



- Ritchey Chretien Telescope 1.54 m
- Alt-Az mount, speed = 5°/s, absolute accuracy < 3 arcsec rms

1. Satellite illumination – Experiment purposes

❑ SLR Trends : **High rate (up to 100 kHz – MHz), Two color technique**

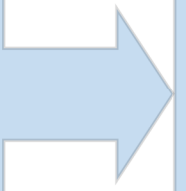
➡ **Smaller detector** (25 – 80 μm **APD** or 50 μm **SNSPD...**)

❑ **Grasse station – 1.5m F20 Telescope**

➔ **Limited Field of View** (FoV ~ 5 – 15 arcsec) on **small-size detector**

➡ Needs: **good pointing and fine-tracking** of the telescope during satellite pass.

when satellite is **not visible...**
➔ Difficulties to find the satellite
➔ **Large error pointing**

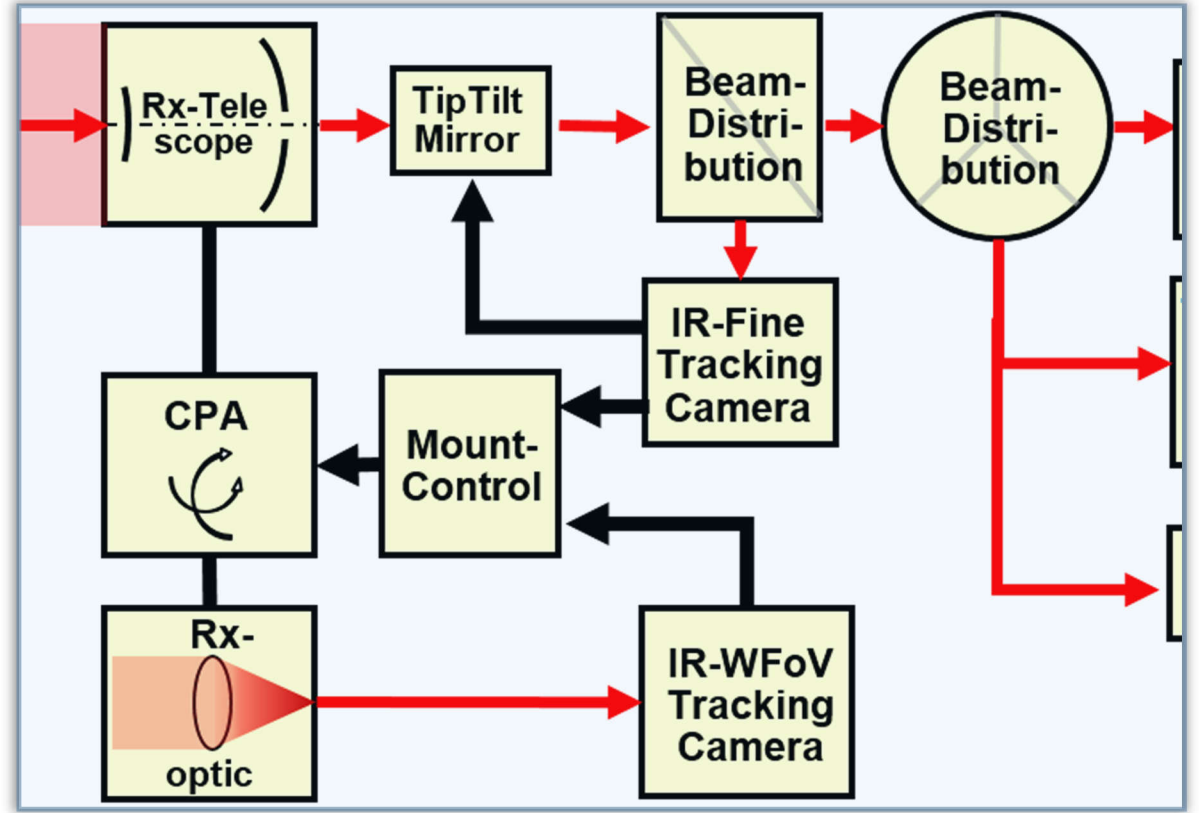
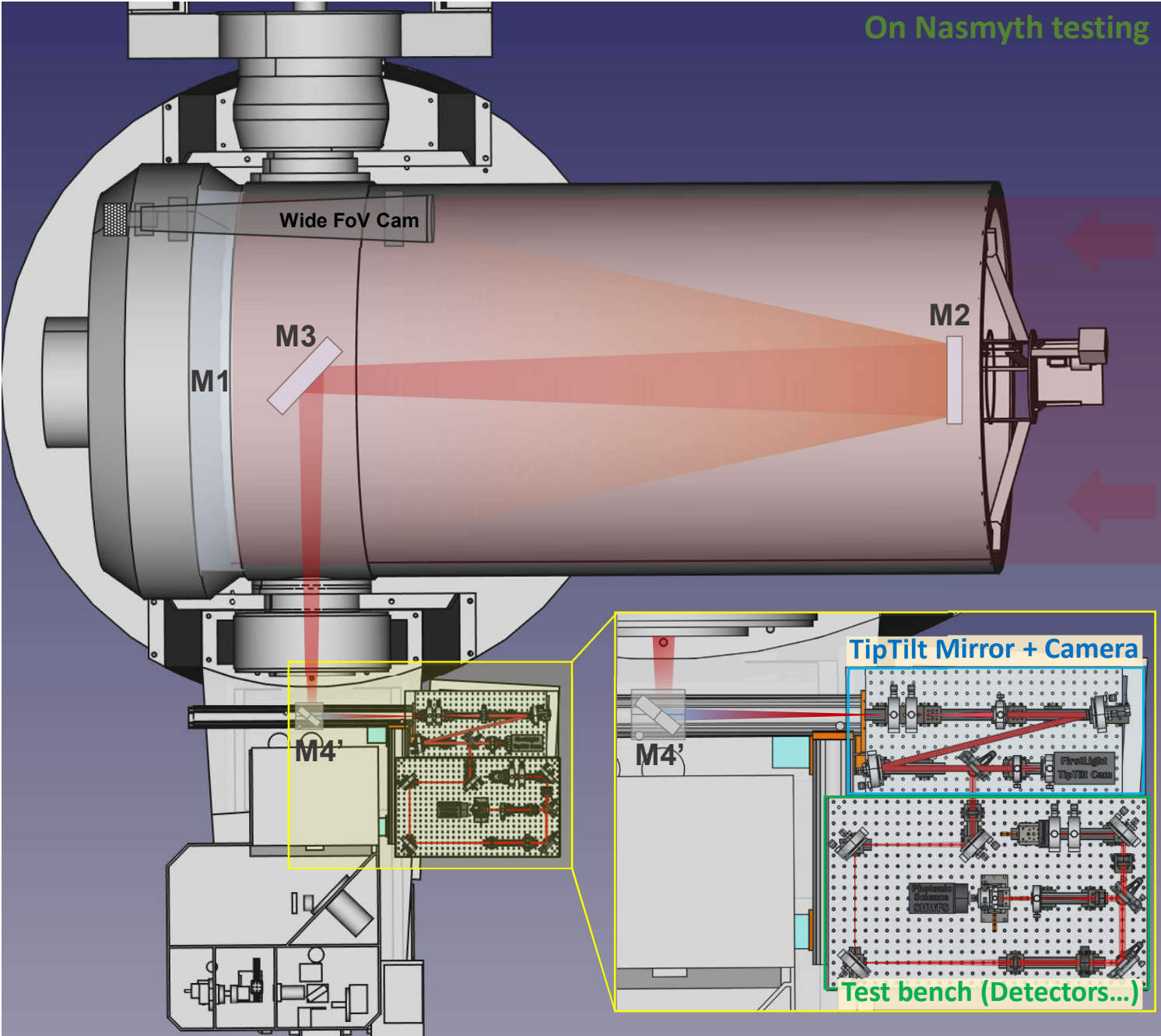


➔ **Discontinuities** in the ranging data...
➔ **Errors** in timing detection...
➔ **Difficult to activate**
‘autonomous operation’

❑ **Solution:**
illuminate the satellite (by **high-power laser**) and using **fine-tracking** (TipTilt mirror) in order to **maintain** the **returning signal** from satellite in the **center of detector**.

2. Architecture of experiment – Fine-Tracking

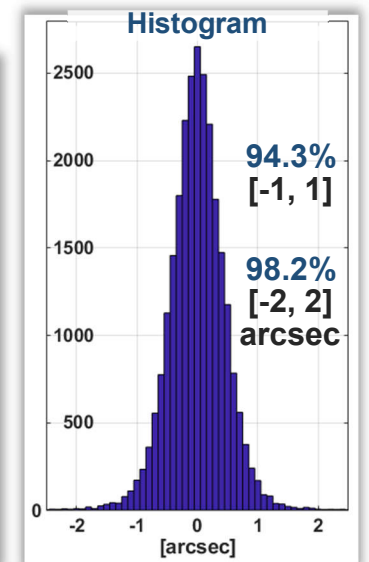
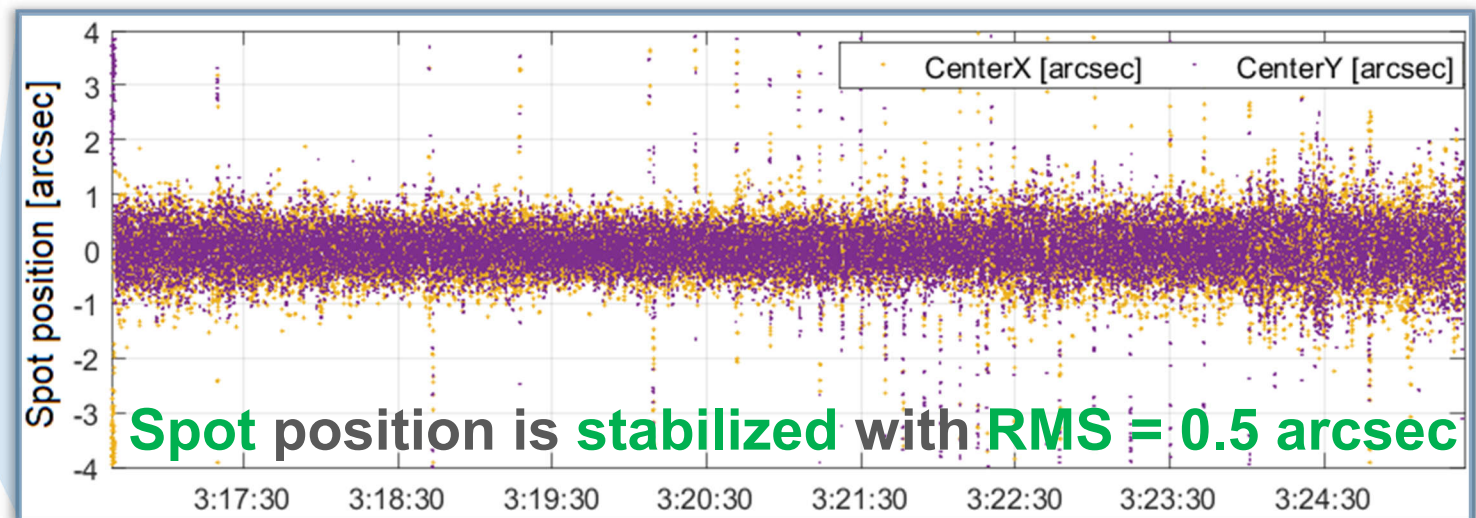
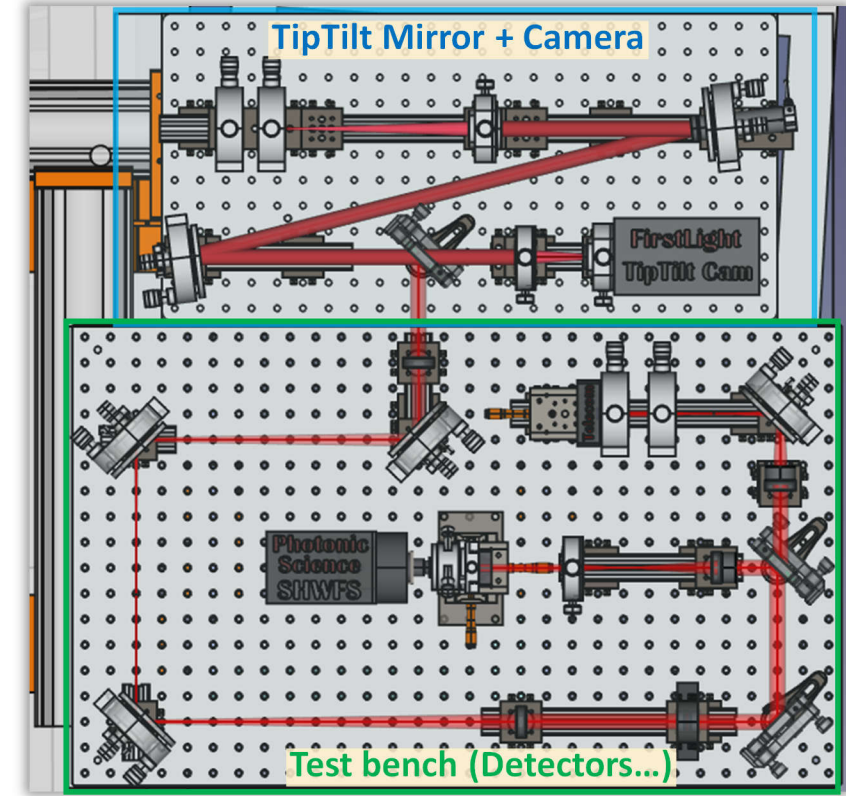
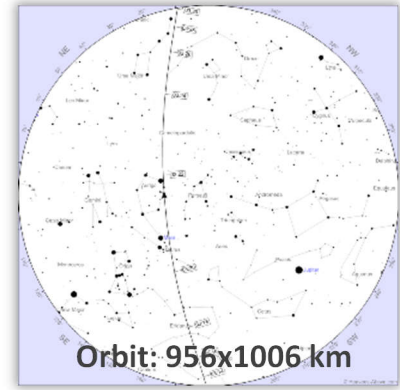
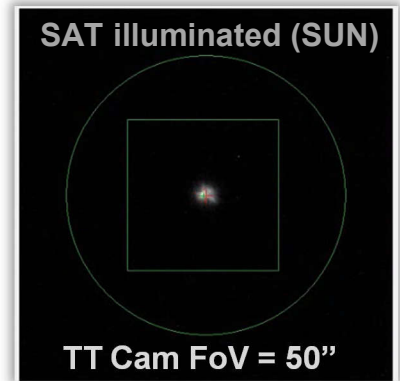
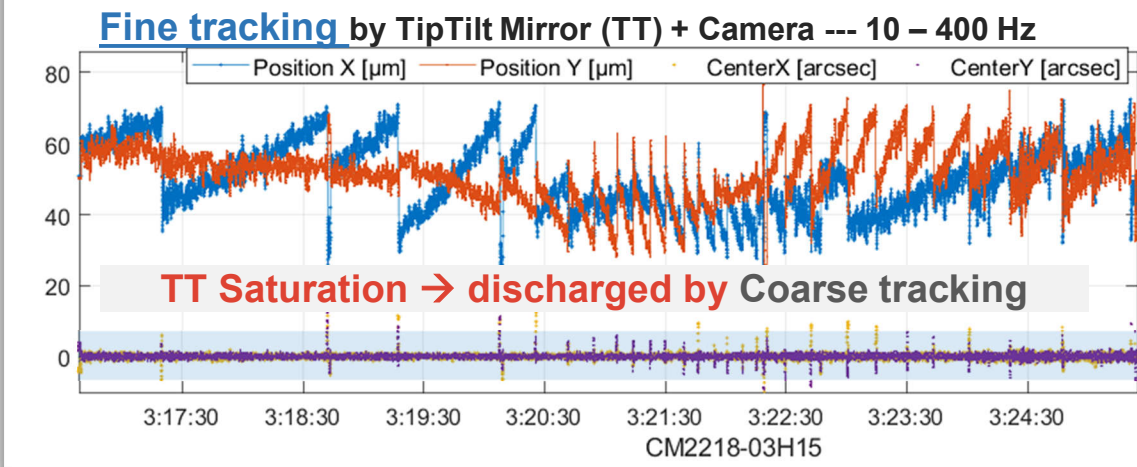
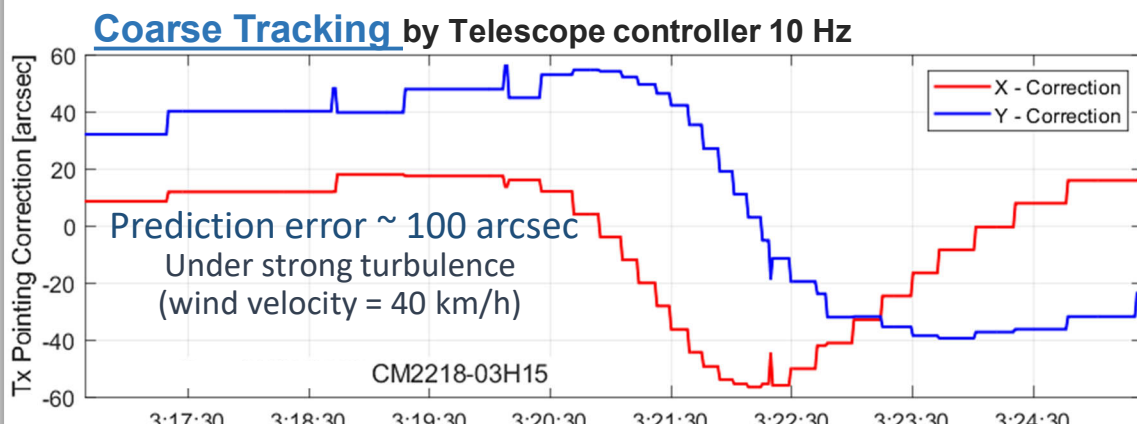
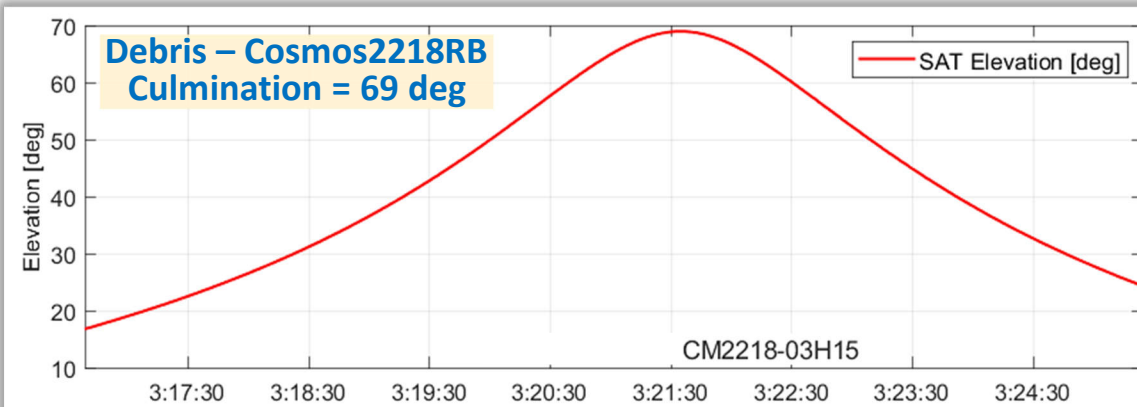
Coarse-tracking (1.5m telescope controller) and **Fine-tracking** (TipTilt mirror + Camera)



□ **Auto-tracking**
Fine-tracking discharges its correction by **Coarse-tracking** when the corrections approach the TipTilt saturation limit.

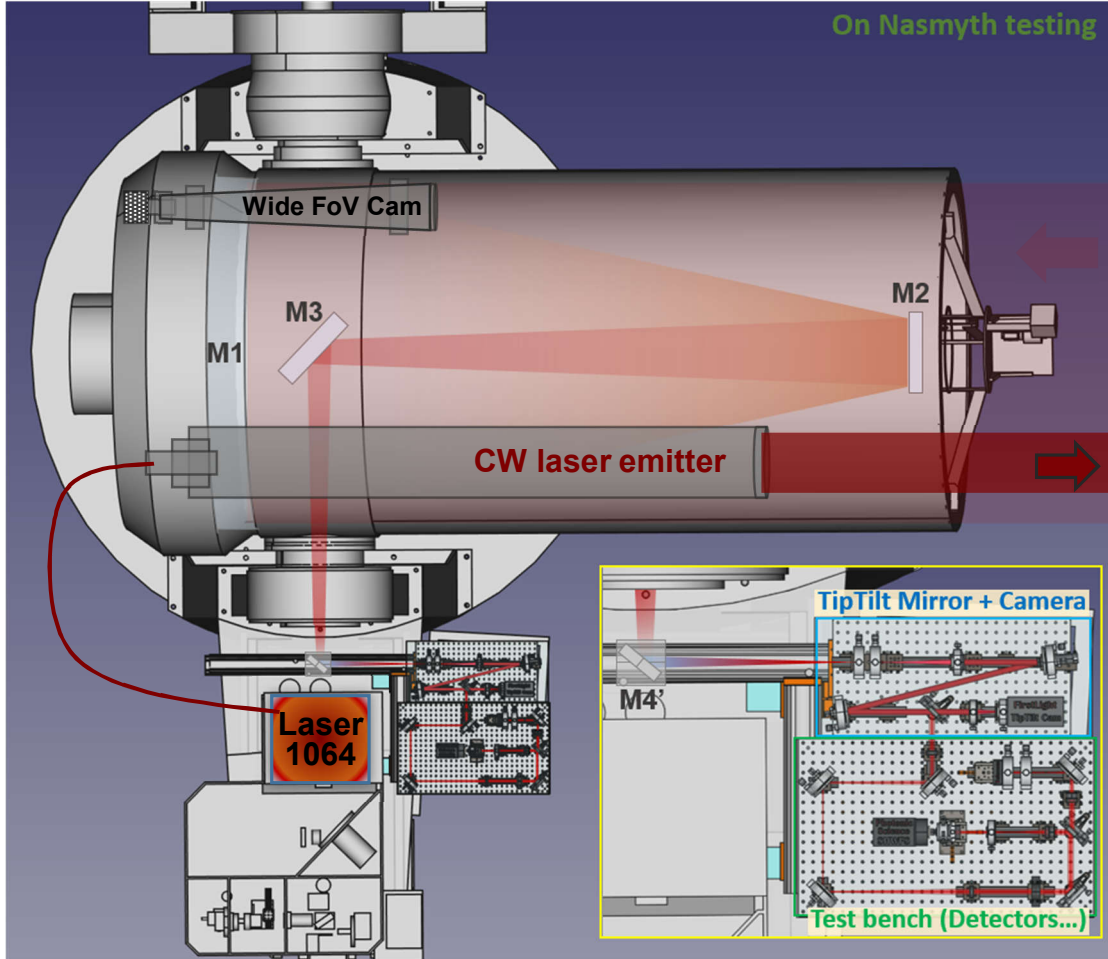
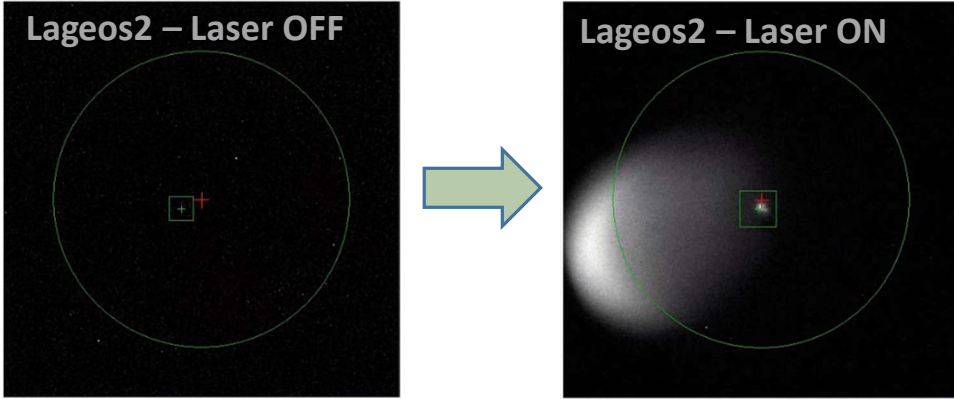
2. Architecture of experiment – Fine-Tracking

Coarse-tracking (1.5m telescope controller) and Fine-tracking (TipTilt mirror + Camera)



2. Architecture of experiment – Fine-Tracking

When satellite is **not illuminated** (by the SUN), we use a **high-power (30W), continuous laser 1064 nm**
Full Divergence = 100 – 200 μ rad



High power cw laser (30W)

- 195 mm telescope
 - Aperture: 195 mm, F/9
 - Material: Carbon
 - Lens: Apochromatic
 - Limited Diffraction

□ 1064 nm laser ML30-CW

- Power: 3 – 30 W
- Output: Monomode fiber
- FWHM: < 3 nm
- Beam quality: $M^2 < 1.1$

□ Tuning

- Optical Power (Software)
- Divergence (20 – 200 μ rad)
- Orientation (manual)



High-speed IR Camera

- FirstLight C-RED 2
 - Resolution: 640x512 pixel²
 - Pixel size: 15 μ m
 - Quantization: 14 bits
 - Frame rate: 400 Hz
 - Noise: 30e- at 400 fps



TipTilt Mirror - Motorized

□ PI S-330.8SL + E.505/E.509

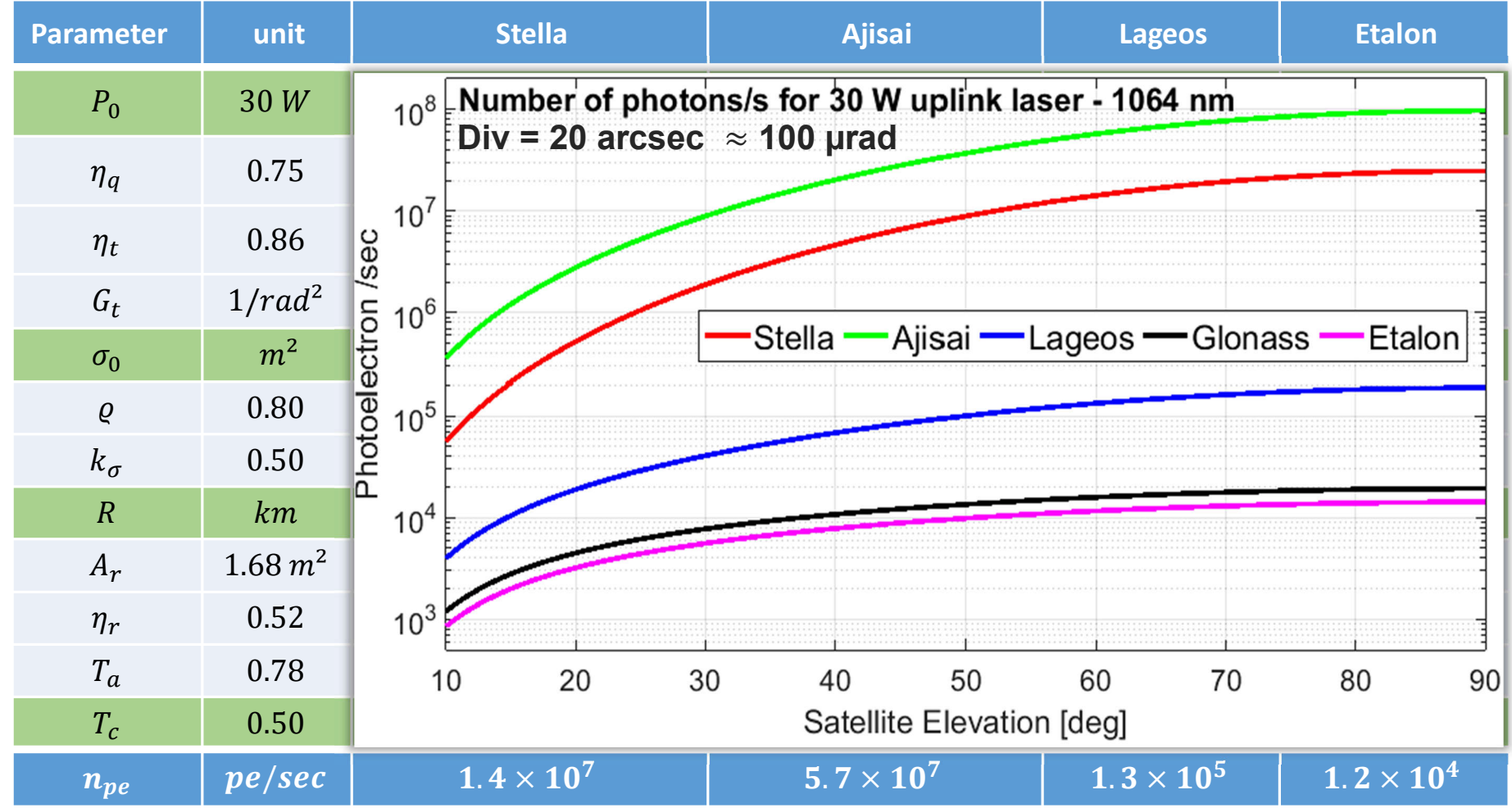
- Dynamic Range: 10 mrad
→ 50 arcsec Correction
- Jitter: < 0.2 μ rad, 1.5 kHz
- Mirror Diameter: 25 mm



2. Architecture of experiment – Link budget

Link budget -- **high-power (30 W), continuous laser 1064 nm**

$$n_{pe1s} = \eta_q \cdot \left(P_0 \frac{\lambda}{hc} \right) \cdot \eta_t \cdot \frac{G_t}{4\pi R^2} \cdot \frac{\rho \sigma_0 k_\sigma}{4\pi R^2} \cdot A_r \cdot \eta_r \cdot T_a^2 \cdot T_c^2$$



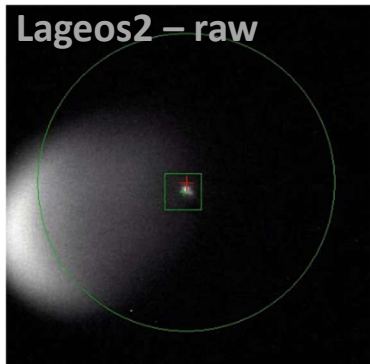
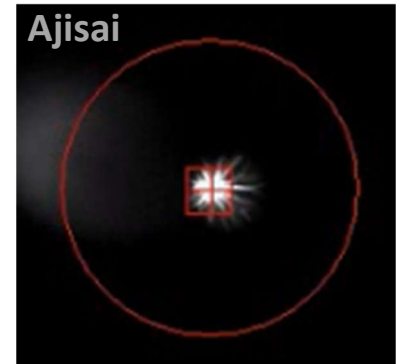
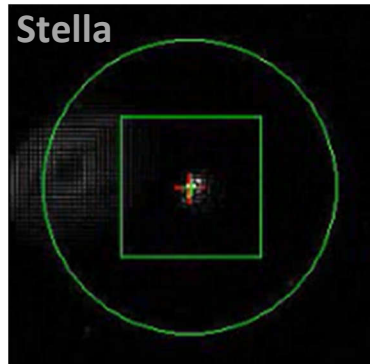
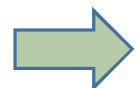
Spot size in the tracking cam:
5×5 pixels (0.32"/pixel)
 Camera noise of **30e-/pixel**, we need at least 100 pe/pixel in order to achieve a good spot movement measurement for auto tracking process.

Lageos: 60 deg Elevation
 by TipTilt Mirror + Camera --- 10 – 400 Hz

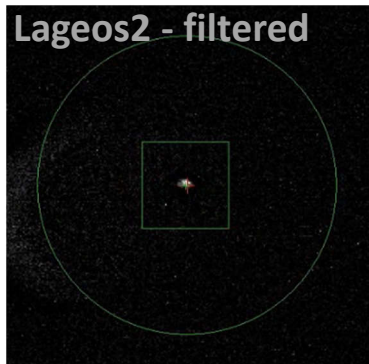
At **10 Hz**, we expect to have:
400 pe/pixel → good SNR
 → activate fine-tracking

Link budget estimation OK

Laser 1064 nm - ON



Lageos2
 ~400pe/pixel
 Diffusion issue...
Image Filtering

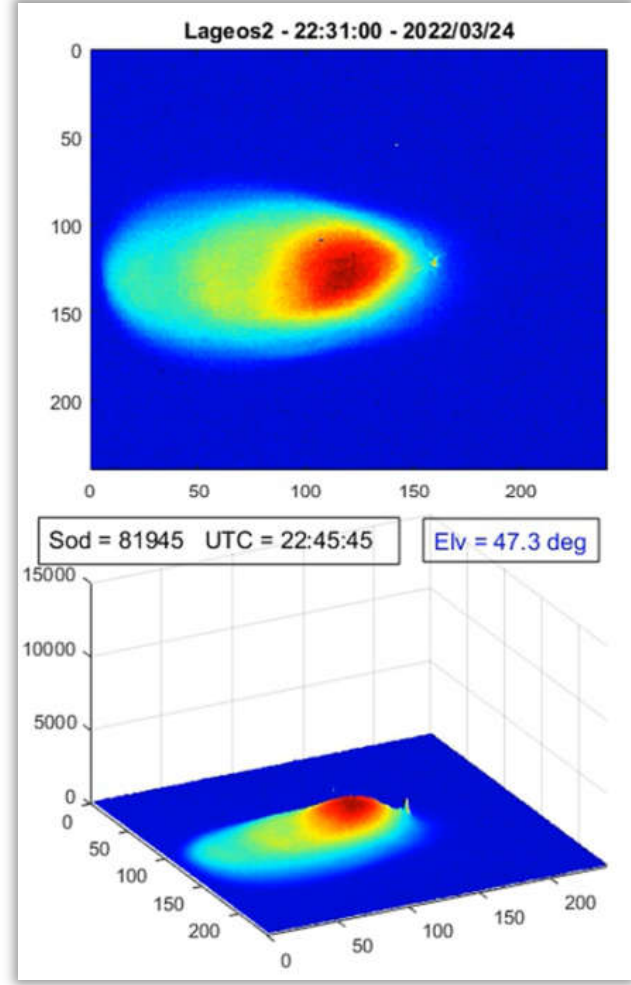
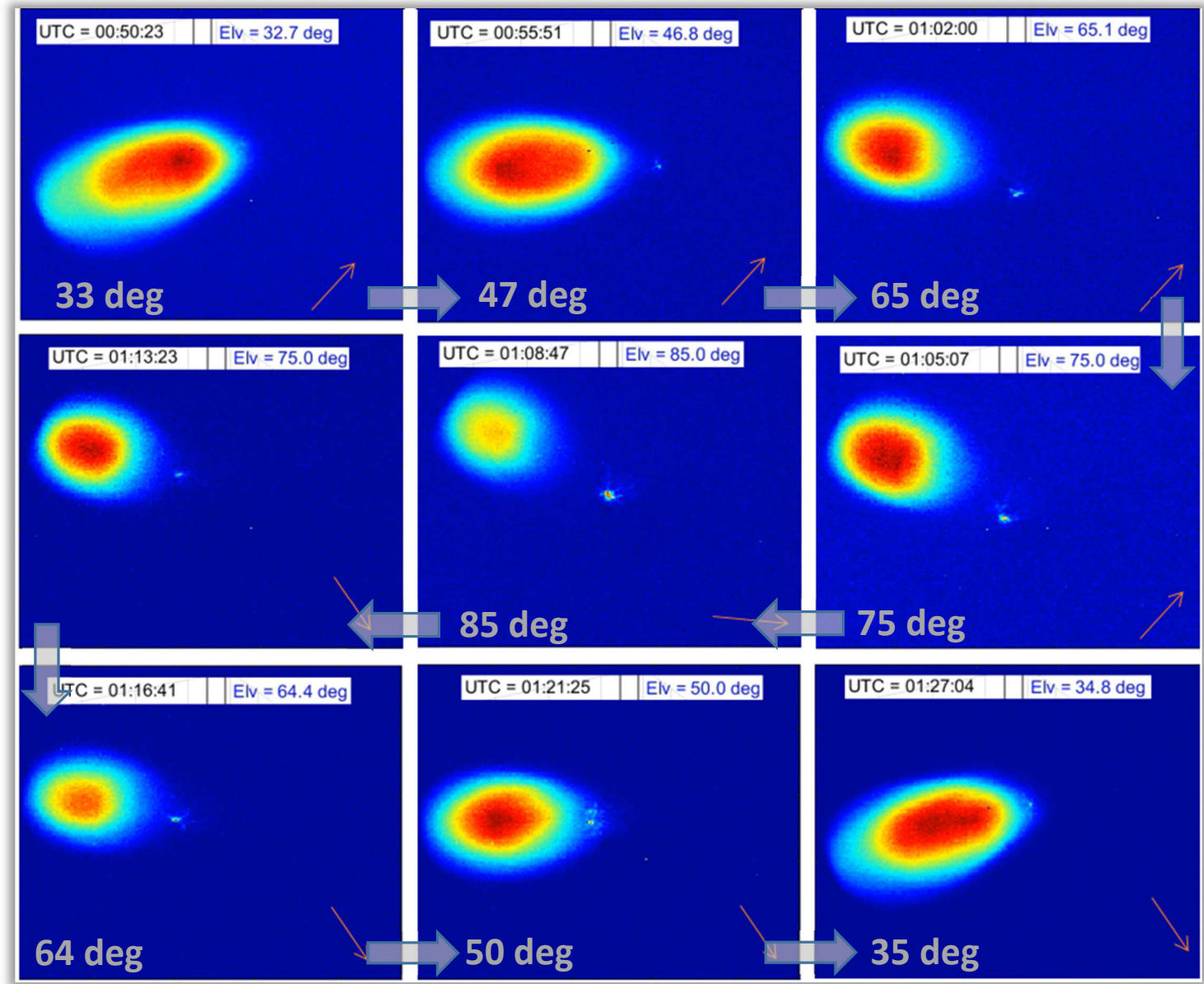
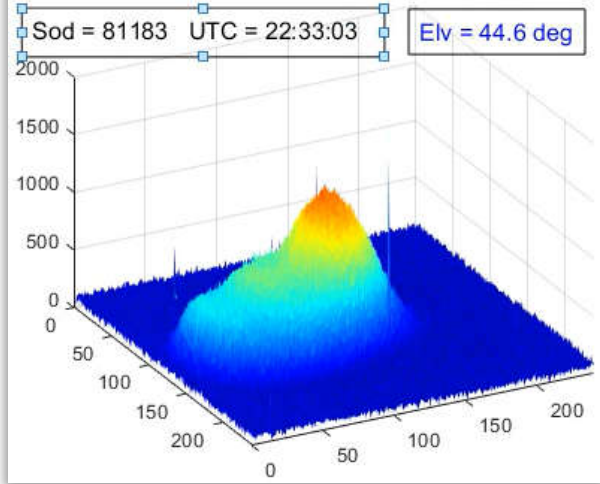
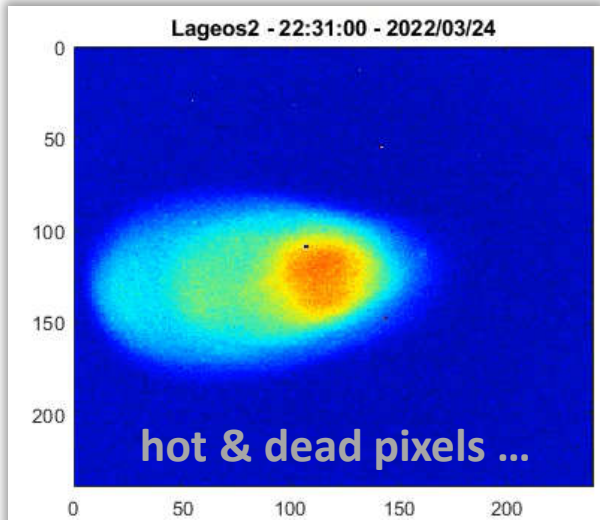


3. Filter & centroid detection – SNR optimization

Image Filter – SNR optimization on centroid detection – Lageos (limited link budget)

- hot & dead pixels
- offset noise

➤ Diffusion...(size and position change following telescope elevation)

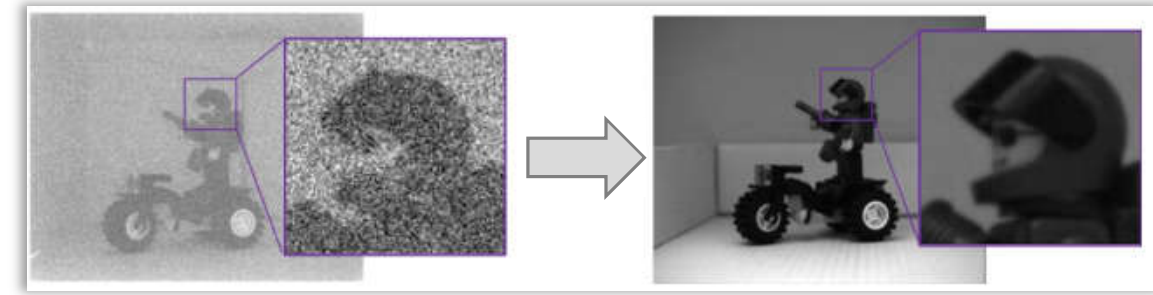
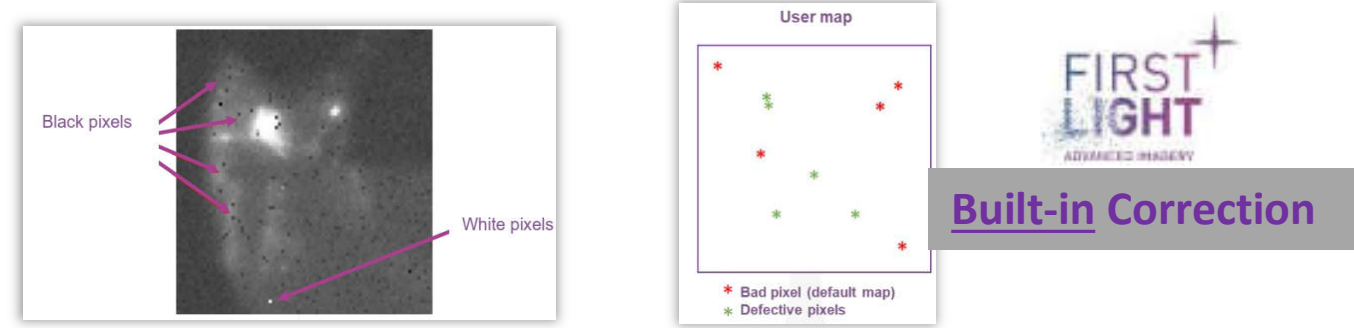


➔ Spot detection???

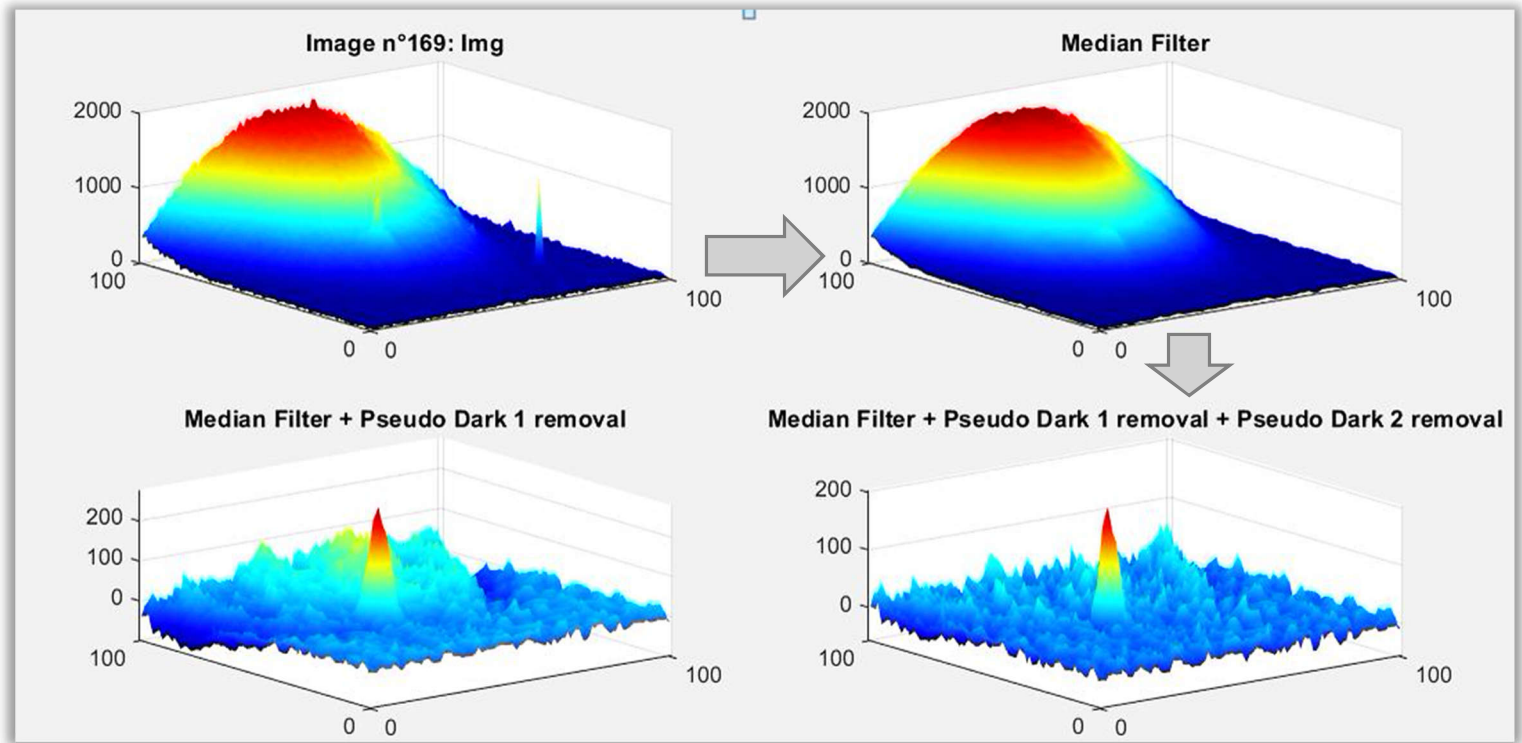
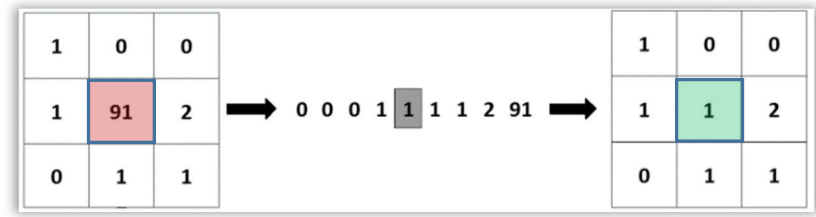
3. Filter & centroid detection – for Lageos fine tracking

Image Filter – SNR optimization on centroid detection

- Cooling (noise \searrow) + Dark suppression (offset + uniformity \searrow),
- + Correction map FirstLight CRED-2 Camera (hot & dead pixels 0.28% correction) --- Camera Build-in Tools

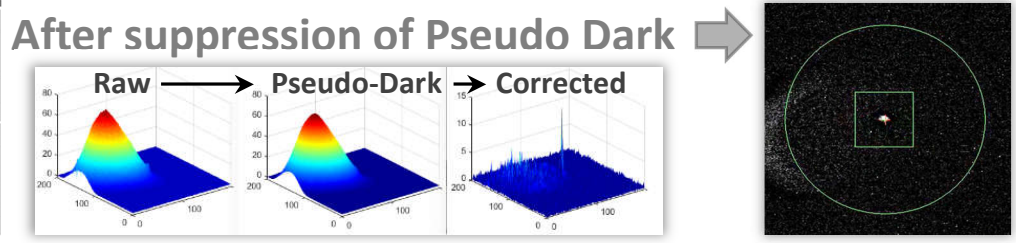
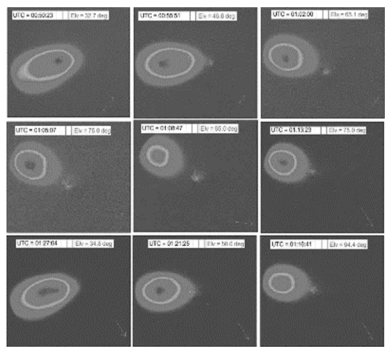


- Median filter (filter hot pixels), on 3x3 or 5x5 pixels



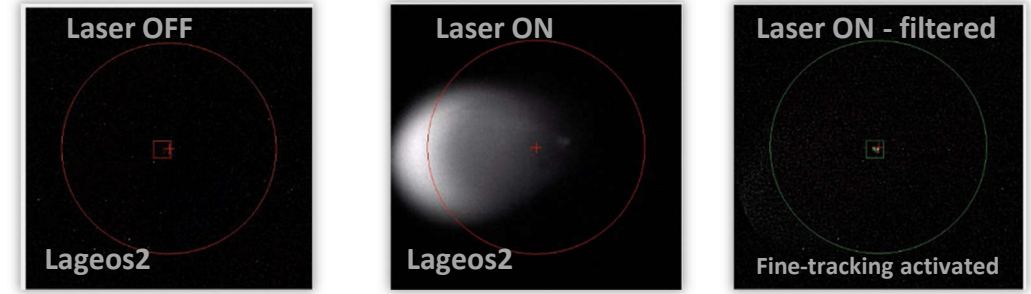
- Box blur or low-pass filter (Diffusion Suppression)

Aim to recalculate the Pseudo Dark
Kernel size = 7x7 (larger than spot size)



4. Preliminary results – Lageos fine tracking

After optimize the spot detection by using some filters, we can activate the fine-tracking on Lageos illuminated by the high power laser.



TipTilt correction
 ~21 arcsec on X – axis
 ~18 arcsec on Y – axis



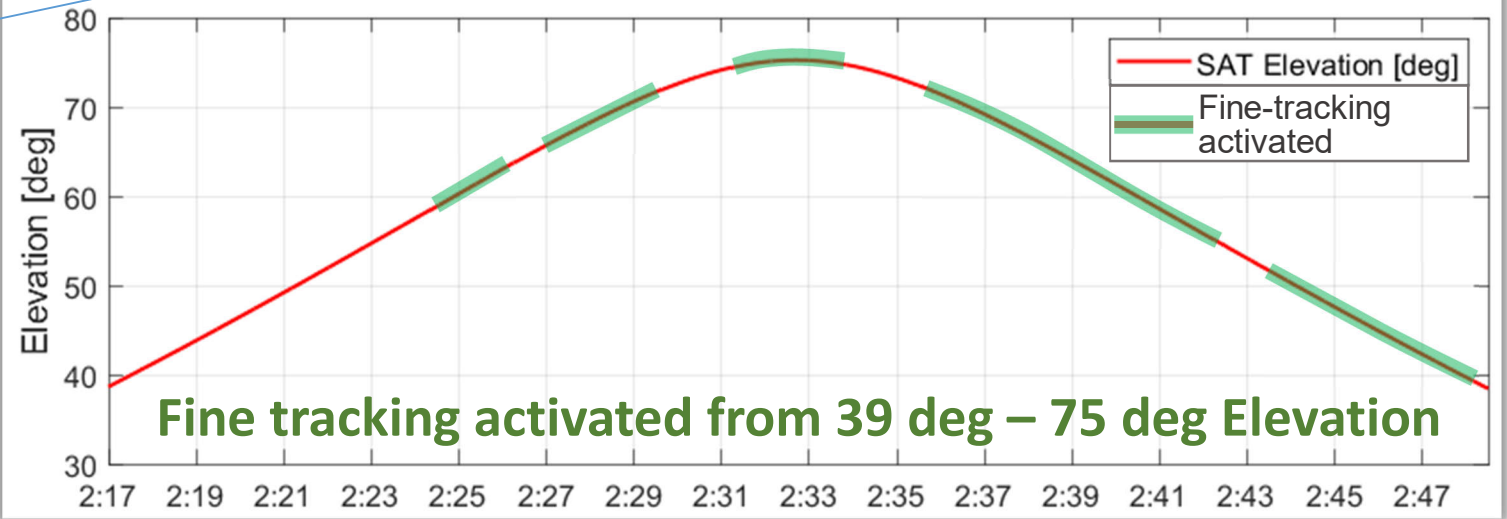
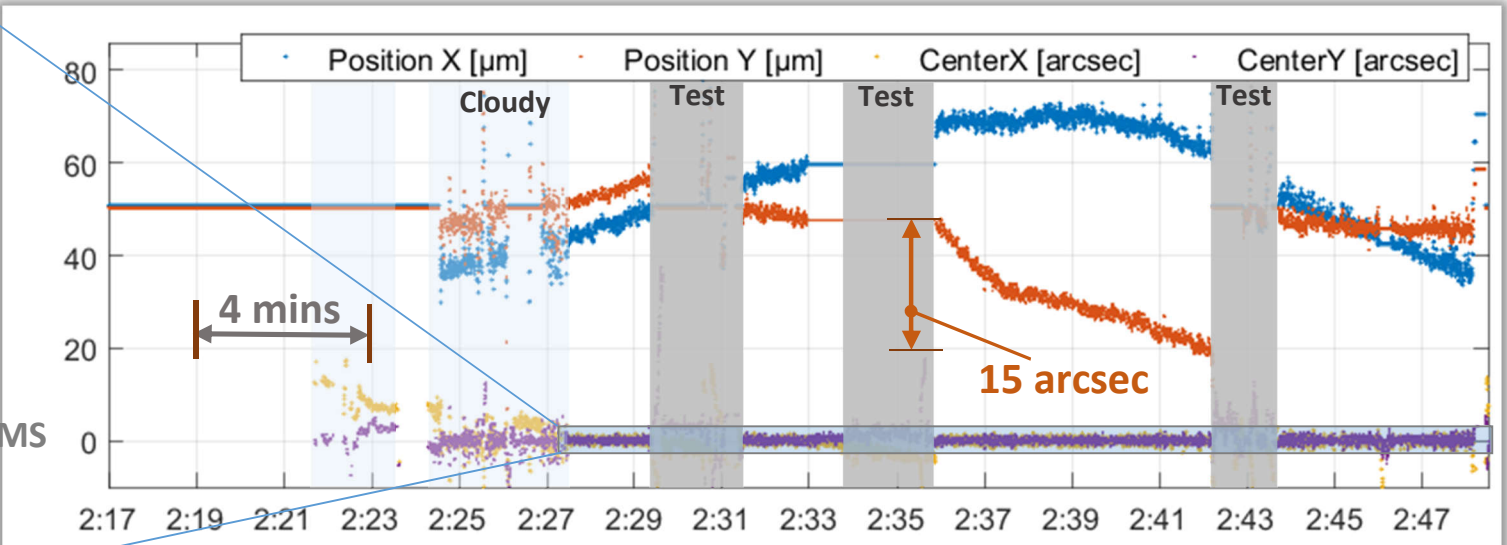
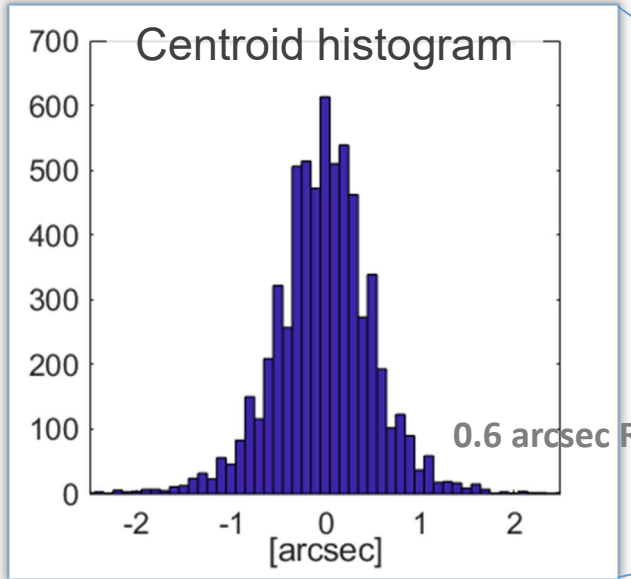
Telescope Jitter
 Telescope error pointing
 Prediction error
 Aberration velocity

0.6 arcsec RMS

With Image Filter + optimization on centroid detection

Auto- Fine tracking on Lageos can be activated from elv > 39 deg under strong turbulence condition (seeing > 2 arcsec)

Spot motion RMS = 0.6 arcsec (spot size ~2 arcsec FWHM) during 20 minutes (~ 50 % of Lageos tracking time)



5. Conclusion & Prospective applications

- ❑ Preliminary results on satellite illumination with Imaging Filter – SNR optimization
 Ajisai, Stella ---- OK during daylight, Lageos ---- OK during night-time.
 Lageos during daylight → to be optimized... (1550 nm + filter)
- ❑ Satellite velocity (6 – 10 arcsec) may be modeled and corrected by TipTilt mirror
 also the misalignment between transmitter & receiver to increase the link budget
 → better SNR to activate fine tracking and higher signal for detection
- ❑ Integration the Fine-Tracking (TipTilt mirror) into actual SLR schema
- ❑ SLR measurement test with fine tracking → Estimate the improvement on SLR performance
 (Sat illuminated by the SUN or by a high-power laser)
- ❑ Image SNR optimization process (adapting blurs size) can be applied for other camera to filter diffusions from cloud/dome/telescope during daylight. It can improve 1 or 2 star-magnitude on star detection (day-light).
- ❑ Latency from Imaging filter → Applying physical filter... instead of Blurs box (numerical) filter.
 Ultra high-pass Fourier filtering – small pin-block in '4-f' spatial filtering system

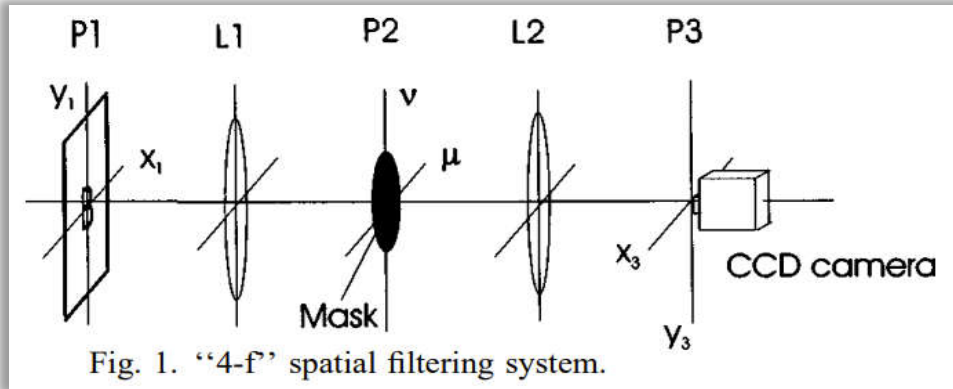
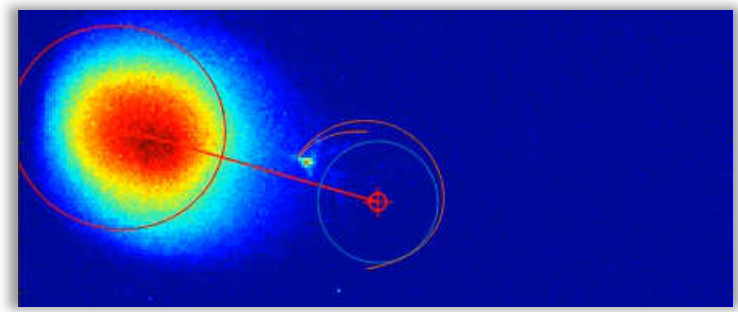
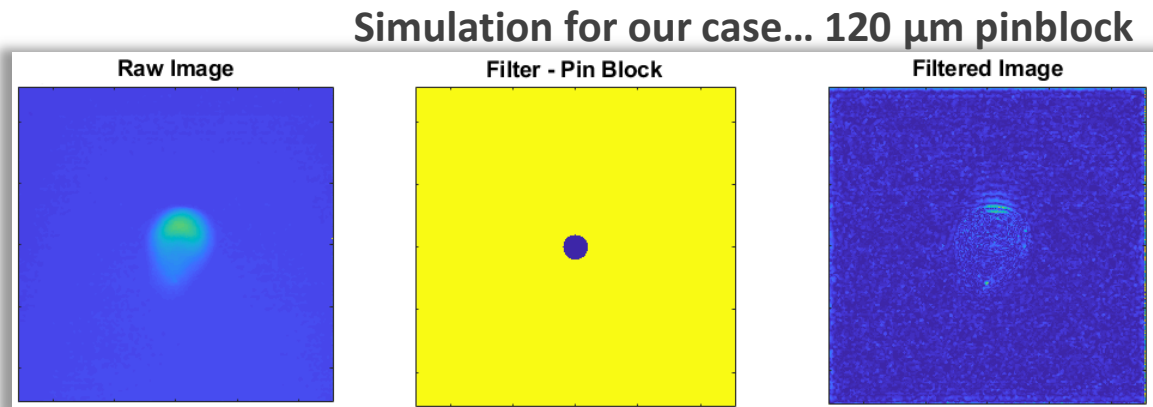


Fig. 1. "4-f" spatial filtering system.

Am. J. Phys. 69 (5), May2001



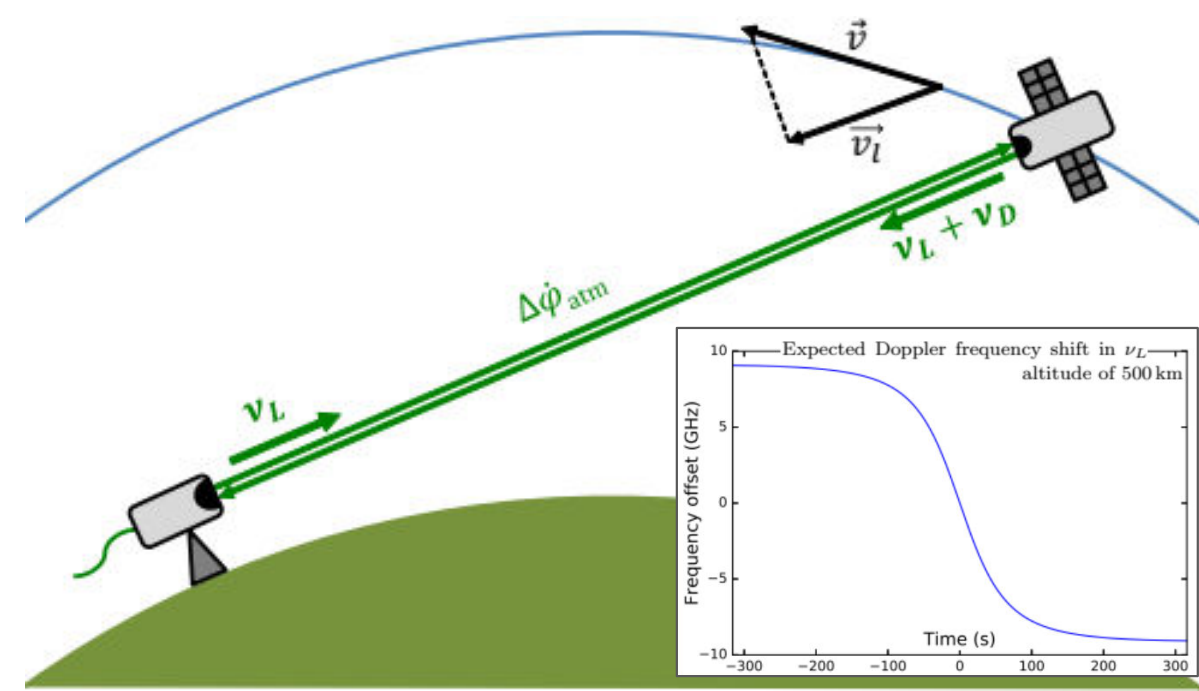
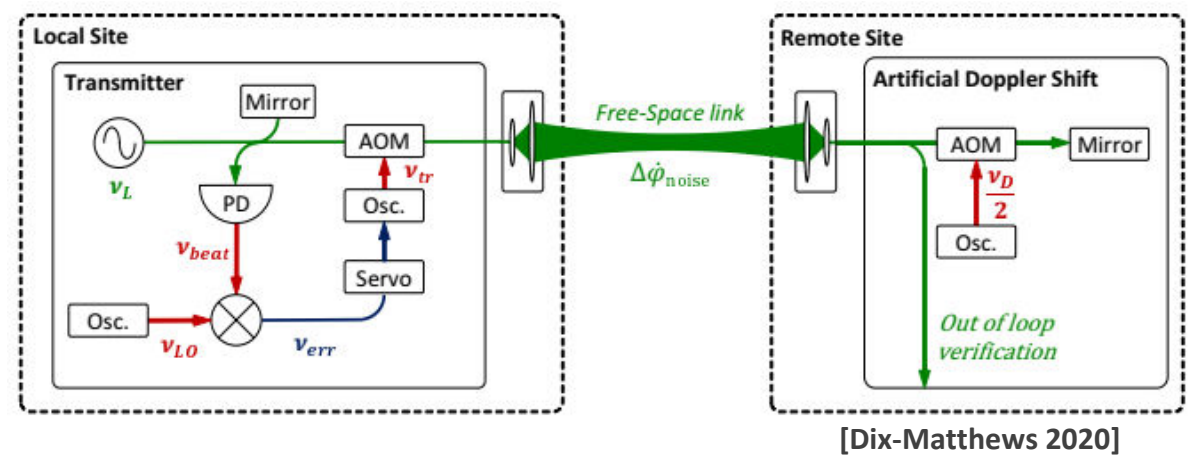
Simulation for our case... 120 μm pinblock

To be continued...

5. Conclusion & Prospective applications

Modulation & Detection of high-power laser

First-step for 'Methods for coherent optical Doppler orbitography'



Atmospheric phase-noise (imprinted on Doppler measurement, ν_D) can be suppressed using the predicted Doppler shift, $\hat{\nu}_D$.

$$\nu_{err} = 2\Delta\nu_{tr} + 2\Delta\dot{\phi}_{atm} + (\nu_D - \hat{\nu}_D)$$

The stabilized **Doppler measurement** may be obtained from this **component of $\Delta\nu_{tr}$**

The most significant obstacle: **optical power losses** during a long-distance **ground to space transmission**
1 W transmitted → receive **~nW (-60dBm)** with high-speed photodiode...

→ **Fiber coupling & amplification may be inevitable!**

→ Experiment with our **1550 nm 42 dBm (15W) amplifier (MHz - GHz modulation)** for the transmitter and measure the returning signal (fiber coupling & amplified → detection).

Dix-Matthews, B.P., Schediwy, S.W., Gozzard, D.R. et al. Methods for coherent optical Doppler orbitography. <https://doi.org/10.1007/s00190-020-01380-w>

Promising performance on 2.2 km free-space:
estimated range rate precision of **9.0 nm/s** at **1 s** of integration,

Thank you for your attention!

