Tracking non-cooperative low earth orbit objects via multi-static radar using GNSS satellites



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Space Situational Awareness

- Detecting hazards
- Space based services become essential
- Any shutdown or loss to space assets will seriously effect our commercial and civilian activities



Space Debris

- Size and amount [1]
 - 29,000 (larger than 10cm)
 - 750,000 (1 to 10cm)
 - More than 166 million (smaller than 1cm)
- Moving at a speed of 7.6 km/s approximately in LEO



Image source: http://www.esa.int/spaceinimages/Images/2013/04/Distribution_of_debris

Space Debris is a Problem

- Risk of collision
 - Threat to active satellites
 - Future space missions in danger
- Shortens satellite life due to fuel burn to avoid collisions
- Collisions resulting more debris which in turn increase chance of collision

Bistatic Radar Tracking

- GPS Satellites as possible transmitters
 - Transmits from MEO (~20,000)
 - Well monitored signal and stable
 - L-band, which minimises atmospheric interaction
- Direct vs Indirect
 - GPS signal as direct
 - Debris-scattered as indirect



Challenges?

- Weak-Strong problem
 - Strong direct signals are possible interferers
- Very low power debris scattered signal
 - Very long coherent integration needed to get sufficient gain
- Signal processing
 - Long integration costs additional processing burden

Detecting Debris-Scattered Weak Signal

- Weak-Strong problem
 - Strong counter signals
 - Detection of weak signal requires high correlation gain of desired signal
 - Correlation rejection for non-desired signals is also required
- Debris scattered signal has a much higher Doppler and rate of change over the observation period fairly quickly in compare to direct signal
 - Maximum Doppler Shift for direct signal ±5kHz
 - For Debris-scattered Maximum Doppler Shift is ±37kHz



Detecting Debris-Scattered Weak Signal

- The local replica is generated following the predicted trajectory
- Local replica matches the desired signal
- Strong counter signals experience change of Doppler offset during correlation



Signal Observation Period

Signal Processing

- Processing should be done in stages to minimise the processing cost of multiple possible track deviations
- Receiver generates replica for expected signal
 - With PRN, data and Doppler shifted in complex domain
- Integrate and Dump (I&D)
 - Down shifted to audio-rate signal
 - To minimize phase error
- Test possible trajectories at low cost



In-orbit Demonstration

- LEO CubeSats carrying a GPS repeater
- In-orbit Demonstration
 - Verify proposed technique
 - Helps modelling LEO debris orbit determination



Concept Demonstration

- Software Defined Radio (SDR)
 - Flexible
 - Cost efficient



Experimental results



Improved SNR through Regeneration



Plan for Technique Validation

- M1 CubeSat has SDR that received GPS
- SDR can also transmit in 2.4GHz band
- SDR can retransmit GPS L1 signal onto 2.4GHz
 - Includes Doppler shift from both reception and (scaled for frequency) transmission
- Much stronger
- 2W radiated power, 0-5dB antenna gain depending on pointing
- Regeneration is via digital baseband
 - Could also insert pilot tone to make spacecraft tracking easy
- Ideal engineering development model
- M1 has a retroreflector (laser ranging is possible too)

Conclusions

- Substantial processing gain is required to detect the extremely low power debris-scattered GNSS signals.
- Long duration integration is achievable
- Affordable processing technique
- Realistic experiments to verify proposed technique