Verfication of ELT performance by Monte Carlo Simulations

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The ELT experiment





Optical time transfer



- One way: $tof_{1W} = R_{COM} + \tau_{troposphere} + \tau_{Sagnac} + \tau_{Shapiro} + \tau_{attitudeDetector}$

Two way:

$$tof_{2W} = 2 * (R_{COM} + \tau_{troposphere} + \tau_{Shapiro} + \tau_{attitudeReflector}) + \tau_{Reflector}$$

Time transfer:
$$\tau = \frac{t_{return} + t_{start}}{2} - t_{detector} + \tau_{corr} = \frac{t_{of_{2W}}}{2} + t_{start} - t_{detector} + \tau_{corr}$$

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Simulation tool

Geometric components

- Earth orientation (IERS 2010 Conventions)
- ISS attitude simulation
 (3 axes, constant offsets and oscillations)
- Detector and reflector position
- Intra-reflector delay (function of incidence angle)
- Visibility constraints (minimum elevation)

Signal delays

- Troposphere (including cloud cover)
- Sagnac effect (processing in ITRF)
- Shapiro delay

Relativistic effects on clocks

- Drift of clocks w.r.t. to UTC
 - ... due to special relativity (relative velocity)
 - ... due to different gravitational potential

Stochastic components

- Background noise
- Laser Jitter
- Pulse width
- Noise of ground- and space-based clocks
- Cloud coverage (frequency and duration)

Simulation tool



n	Nama	Precise filename	(mat)		X [m] Y [m]	Z [m]		_
	Matterall	Frecise-mename select	(inde)	Reflector offset	40.770	7 004	Select reflectors	
8834 (weti) 🗸	wettzei	ISS_2017-04-03_8834	Ts1_pass15_full		10.776 9	7.021	ELT reflector	5
X [m]	4075576 613	Orbit errors in and intings (CD			X [m] Y [m]	Z [m]	JEM Hemi A	5
	1010010	cr0	r) cr1 cr2	Detector offset	40.078	7.024	JEM Hemi B	5
Y [m]	031785 727	Radial [m] 0	0 0	Detector entect	10.978 9	7.021	IDA 1 Hemi B	
	331103.121	ca0	ca1 ca2	Satellite Angle	Ar const ArAr	ArAr Ar	C2V2 S3 Forward	
7 [m]	4004500 700	Along-track [m] 50	0 0	rotation [deg]	[deg] [deg]	period [s] type	C2V2 P3 Nadir	5
- 1-1	4001503.739	CCU	CC1 CC2	Roll (X) 0.7	0.5 0			
		Cross-track [m] 0	0 0	0.1	0.5			
tion Parameters -		_		Pitch (y) _0.4	-0.5 0			
Station Dance Bise	[m] 0	Orbit time bias [ns] 0		-0.4	-0.0			
leteorological Data	[in]			Yaw (z) _4	0.5 0			
Temperature [K	288.4				0.0			
		Clock Parameters						
Pressure [hPa]	947.4	Station clock offset to UTC [s]	Laser		etector parameters			
		0	Jitter [ns^2] Pu	ulse length [ns]	Jitter [ns^2] Gate	width [ns] Ac	tivation time [ns]	
Rel. Humidity [9	61 86		0.0003	0.02	0.0014	100	0	
		Station clock noise type						
Calibration Offset [r	ns] 9999	ahm : Active Hydroge 🗸	Sampling [Hz]		One-way detection B	ackground Mir	nimum elevation	
Two-way Detection	,		100		probability no	oise rate [1/s] for	visibility [deg]	
Probability	0.1	ACES clock offset to UTC [s]			0.1	8e+05	0	
		0	Zenith gap 🗸					
Jitter [ns^2]	0.0014		Minimum elevation					
Minimum elevation		ACES clock noise type	zenith gap [deg]	80				
for visibility [deg]	10	ACES: SHM + PHARAO V						
Background			J		Load defaults	Save inputs as	new defaults	
noise rate [1/s]	8e+05		_					
			Write simula	ation tiles				
Cloudy sky 🗌								
Cloud cover [%] Cloud width [s]	Optional folder end	Release(s) (0 to 99	9) 0	Evaluation	1	Start	
		for storing the files						

Multi-reflector problem





Reflector identification

Binominal filtering

Monte-Carlo simulations

- Data simulation and processing for identical parameters
- Passes

-

- Laser system characteristics
- Signal propagation characteristics
- ... (neglecting multiple reflectors on the ISS)
- Randomness introduced by the following sources
 - Background noise
 - Laser jitter
 - Pulse width
 - Clock noise

- Studies without systematic errors
 - Expected to converge to "true" clock offset
 - ... if filtering does not fail
 - ... and yields unbiased time transfer triplets
 - How does <u>filtering</u> perform statistically?

- Studies with systematic errors

- Unknown <u>attitude</u> and orbit errors will be present (particularly in quick-look processing)
- Effects of <u>cloud coverage</u> and other constraints on performance

Results

Background	Noise	Time transfer			
noise rate [1/s]	reduction	σ [ps]			
5.00E+05	no	1.6979			
5.00E+05	yes	0.4541			
5.00E+06	no	4.7743			
5.00E+06	yes	1.8626			

Monte-Carlo simulations

Attitude error (only roll)

Real-time TB correction

100 m along-track orbit error, 1 m radial orbit error, constant 0.5° attitude error

