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### **1. Introduction**

Goal: ① To develop a small and cost-effective laser ranging arrays (LRAs) for satellite laser ranging (SLR)
 ② To mount onto upper stages of a rocket that are disposed of without controlled re-entry

Outcome:

- ① Grasp of motion of space debris
  - Importance of motion grasp in the active debris removal (ADM)
  - No case where the motion of upper stages of a rocket is measured
  - Capability of SLR to measure target's rotational motion by kHz ranging
- 2 Improvement of accuracy and precision of orbit determination in re-entry prediction



Active debris removal using

The main propellant tank of the second

Increase of observational data

electrodynamic tethers (JAXA)

stage of a Delta 2 launch vehicle landed near Georgetown

# 2. Details of LRA

### A) Overview of CCRs

□ The LRA consists of only seven corner cube reflectors (CCRs)





CCRs typically consist of three mirrors or reflective prism faces which reflects an incident light beam directly towards the source.

### Specification of CCRs

No.	ltem	Specification
1	Material (CCR)	Fused Silica

### B) Overview of body frame

CCRs are aligned into each holes of the body frame



Resin model of the body frame of developing LRA manufactured by a 3D printer (JAXA); 7 CCRs are aligned into each holes.

The drawing of the body frame

D-D(1:1)





2	Height [mm]	19.05
3	Diameter [mm]	25.40
4	Dihedral angle offset ["]	+1.0
5	Coating	Uncoated

There is a possibility that these specifications are modified.

### C) Rocket interface

- The range in which the LRA responds to the laser pulse is ±45 degrees at the zenith angle
  - Six LRAs are necessary if covering all directions



Five LRAs are planned to be mounted onto upper stages of a rocket with screws





A part of the drawing of the body frame. Some information is erased.

#### D Specification of body frame

No.	ltem	Specification
1	Material (Body)	Aluminum
2	Material (contact portion b/w body and CCR)	Silicon rubber
3	Height [mm]	31
4	Diameter [mm]	110

Rocket interface of LRAs; is the LRA; another LRA is onto back-side of the rocket. There is a possibility that the interface is modified.

 $\rightarrow$  complete!

# 3. Roadmap of development

- $\succ CAD model \rightarrow complete!$
- Resin model
- $\succ$  Experimental model (EM) → finish until Dec. 2018
- ➢ Flight model (FM)
  → finish until Jan. 2019
- → Mount onto a H-IIA rocket → finish until Feb. 2019

#### 5 Weight [g]

6

Angle b/w central axes of adjacent CCRs [deg]

30.0

Less than 300

There is a possibility that these specifications are modified.

## 4. Conclusion and future work

- ✓ JAXA is steadily developing small and cost-effective LRAs which is mounted onto upper stages of a rocket.
- ✓ LRA will be finally developed at approx. US\$3,500/piece.
- ✓ JAXA is aiming to standardize the installation of the developed LRA on the upper stage of a rocket (at least made in Japan).
- Selling developed LRAs or distributing design drawings are under consideration.

#### 21<sup>st</sup> International Workshop on Laser Ranging, Canberra, Australia, Nov. 04-09, 2018