

NASA

Analysis and application of 1-way laser ranging data from ILRS ground stations to LRO

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Knowledge for Tomorrow

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Data selection LRO SPK Jumps



LRO SPK Jumps throughout the mission

Data selection Observation passes

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Observation passes in the selected timeframe

Data selection Clock -6.8┌ ^{10⁻⁸} S M S M 0 00 M 0 3 0 0 0 0 0 M 0 0 0 -6.85 Selected timeframe for OD: -6.9 Day 571 until 576 (ref. Is Jan 1st 2009) -6.95 Rate in s/s -7 -7.05 -7.1 Old rates • 11 Improved rates -7.15 Averaged rate 400 450 500 550 600 650 700 750 800 Days since 1.1.2009

Ratio matched to fired shots of the observation passes In the selected timeframe

Estimation setup Approach

- Software

- TUDAT & ILR from TU Delft

- Estimation

– Adjustment of parameters of interest via batch least squares

- Trajectory

– Initially usage of SPK, afterwards numerical integration

– Data

- Matched shots from **DLR Berlin**
- Preliminary data analysis
 - Apriori covariance for parameters and their variation
 - Weights for balancing data quality and quantity
- Comparison of the resulting trajectory to the SPK's



Estimation setup Cases

Clock		Clock, Initial state		Clock, Initial state	_		
Clock	In	Clock, itial state		Clock, Initial state Empirical accelerations			
Variation of Weights	Va	ariation of Weights		Variation of Weights			
Variation of Apriori COV	Va Ap	ariation of priori COV		Variation of Apriori COV	Cloc	per day Clock parameters	
					per p	ass	

Analysis: looking at the estimated parameters, the residuals, the covariance and the correlations



Estimation results Initial state and per pass clock



- Residuals are low (std ~5 m) but realism of orbit is questionable



Estimation results Initial state and per pass clock



- Large scatter of clock parameters indicates orbit error absorption



Estimation results Main conclusions

- Application of LR data
 - data is very suitable for clock synchronization but results in high correlation for the estimated parameters
 - Further referencing needed
- Application of weights and apriori covariance
 - Helps to balance data quality & quantity as well as to reduce the correlations of the estimated parameters
- Application of clock approximation over longer timespan
 - Reduces number of parameters to be estimated
 - Optimal length has to be determined
- Application of empirical accelerations
 - Helps to decorrelate the estimated clock parameters and therefore seems to be the right approach



Summary

- Successful linking of our work/approaches (data & estimation)
- Successful application of real data into ILR (modifying ILR & data)
- Validation of models by comparison (relativistic effects, GS position)
- Assessment of LRO orbit quality by the checking the jumps in the SPK
- OD improvement is not yet feasible by only using LR data
 - All miss modeling is influencing all estimated parameters
 - Total referencing in time is not very precise
- But from using only LR we get
 - Clock characterization that is very precise
 - A strategy how to process the data for enabling OD improvements



Next steps & outlook

- Developing more sophisticated models to remove errors in the OD that come from miss modeling (solar radiation pressure, planetary albedo, meteo data)
- Incorporate more details as maneuvers and HGA/LOLA reference points
- Incorporate passes that contain LR data from multiple stations
- Use simulated radio Doppler data to access improvement in the correlations
- Usage of NPT data
- Update the ILR software interfaces and incorporate the stated changes



Thank you for your attention! Questions?

