Determination of the SLR station coordinates and velocities on the basis of laser observations of low satellites

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Abstract

The orbits of three low satellites Ajisai, Starlette and Stella have been determined on the basis of the data collected in 2001-2005 from the best 14 Satellite Laser Ranging stations. The positions and velocities of four SLR stations Graz (7839), Greenbelt (7105), Herstmonceux (7840) and Yarragadee (7090) were determined. Additionally, the station velocities were compared with the geological model NNR-NUVEL1A. All calculations have been made assuming the model of the Earth gravity field EIGEN-GRACE02S. All the calculations have been performed with the use of GEODYN-II program. The results presented in this work show that the data from low satellites such as Ajisai, Starlette or Stella can be successfully applied for determination of the SLR station coordinates and velocities.

1. Introduction

The high accuracy of laser observations and large number of observations allow for determination of the stations' coordinates and velocities from orbits of low satellites such as Ajisai, Starlette and Stella. Ajisai was launched by the JAXA on August 12, 1986. It is a spherical satellite covered with 1436 corner cube reflectors for SLR tracking and 318 mirrors for photography. The main purpose of the Ajisai was determination of the position of Japanese islands situated on the border of four tectonic plates Eurasian, Pacific, North American and Philippine and determination of plate motion. The laser data of Ajisai are used for definition and realization of ITRF and determination it changes in time, determination of geopotential coefficients (Bianco et al., 1997) and determination of spin axis of the satellite (Kirchner et al., 2006).

The twins Starlette and Stella with 60 corner cube reflectors were launched by the CNES, on February 6, 1975 and September 26, 1993, respectively. The laser observations to these satellites are used mainly for determination of Earth's gravity field coefficients, investigation of Earth and ocean tides, and determination of Earth rotation parameters (Cheng et al., 1990, Cheng et al., 1991, Schutz et al., 1989). The orbital and technical characteristics of the three satellites are presented in Table 1.

The first promising results of the stations' coordinates determination were obtained for Starlette and Stella (Lejba et al., 2007). In this work the period of the laser data were expanded to 5 years (2001-2005). Such period is taking as a minimum for determination of stations' velocities.

	AJISAI	STARLETTE	STELLA					
Sponsor	JAXA (JAPAN)	CNES (FRANCE)	CNES (FRANCE)					
Lounch data	August 12	February 6	September 26					
Launen date	1986	1975	1993					
ID number	8606101	7501001 9306102						
TECHNICAL AND PHYSICAL PARAMETERS								
Number of retroreflectors	1436	60	60					
Shape	sphere	sphere	sphere					
Diameter [cm]	214	24	24					
Centre of mass correction [mm]	1010	75	75					
Mass [kg]	685	47.25	48					
CSA [m ²]	3.5968	0.0452	0.0452					
CSA/Mass [m ² /kg]	0.00525 0.00096		0.00094					
ORBITAL PARAMETERS OF THE SATELLITES								
Inclination	50.0°	49.8°	98.6°					
Eccentricity	0.001	0.02 0.002						
Perigee [km]	1480	810 800						
Period [min.]	116	104	101					

Table 1. Characterisation of the satellites Ajisai, Starlette and Stella.

2. Determination of orbital arcs

The orbits of Ajisai, Starlette and Stella were determined using the program GEODYN-II (Pavlis et al., 1998) in two cases with empirical acceleration coefficients determined every 6 and 12 hours. The perturbing forces, constants and parameters used are listed in Table 3.

The satellites arcs were calculated using the observation data from the 14 best ILRS stations (Table 2), collected during the period from January 1st 2001 to December 25th, 2005 (Pearlman et al., 2002). This period was divided into 130 orbital arcs of 14-days each for Ajisai and 182 orbital arcs of 10-days each for Starlette and Stella. Due to similar orbital and technical parameters the data of Starlette and Stella were combined in the calculations.

	Station	ID Number		
1	McDonald	70802419		
2	Yarragadee	70900513		
3	Greenbelt	71050725		
4	Monument Peak	71100411		
5	Haleakala	72102313		
6	Zimmerwald	78106801		
7	Borowiec	78113802		
8	Mount Stromlo	78259001		
9	Grasse SLR	78353102		
10	Potsdam	78365801		
11	Graz	78393402		
12	Herstmonceux	78403501		
13	Mount Stromlo	78498001		
14	Wettzell	88341001		

472259 normal points of Ajisai, 265282 for Starlette and 133093 for Stella were accepted by the GEODYN. The number of rejected normal points is 2335 for Ajisai and 3729 together for Starlette and Stella. The obtained mean RMS of fit values for the calculated orbits for Ajisai are 1.61 cm and 1.98 cm for 6 and 12-hour interval of empirical acceleration coefficients, respectively. For Starlette and Stella the mean RMS of fit values are 1.15 cm and 1.63 cm, respectively.

Table 3. OLOD IN II – force models and parameters
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Force Model					
Earth gravity field: EIGEN-GRACE02S (Reigber et al., 2005), 45x45 solution for Ajisai and					
75x75 solution for Starlette and Stella					
Earth and ocean tide model: EGM96 (Lemoine et al., 1998)					
Third body gravity: Moon, Sun and all planets – DE200 (Standish, 1990)					
Solar radiation pressure: C _R coefficient <i>a priori</i> 1.13 for Starlette and Stella and 1.0 for Ajisai					
Earth albedo (Pavlis et al., 1998)					
Dynamic polar motion (Pavlis et al., 1998)					
Relativistic correction (Pavlis et al., 1998)					
Constants					
Gravitational constant times the mass of the Earth (GM): 3.986004415x10 ¹⁴ m ³ /s ² (EIGEN-GRACE02S Earth					
gravity field model)					
Speed of light: 299792.458 km/s (McCarthy et al., 2004)					
Semi-major axis of the Earth: 6378.13630 km (EIGEN-GRACE02S Earth gravity field model)					
Inverse of the Earth's flattening: 298.2570 (EIGEN-GRACE02S Earth gravity field model)					
Reference Frame					
Inertial reference system: true of date defined at 0 ^h of the first day of each arc (Pavlis et al., 1998)					
Stations coordinates and stations velocities: ITRF2005 solution, epoch 2000.0 (Altamimi et al., 2007)					
Precession: IAU 1976 (Lieske et al., 1977)					
Nutation: IAU 1980 (Seidelmann 1982)					
Polar motion: C04 IERS (IERS 2001)					
Tidal uplift: Love model H2 = 0.609 , L2 = 0.0852 (Pavlis et al., 1998)					
Pole tide (Pavlis et al., 1998)					
Estimated parameters					
Satellite state vector					
Station geocentric coordinates					
Drag coefficients C _D determined every 15 hours for Ajisai and 12 hours for Starlette, Stella					
Acceleration parameters along-track, cross-track and radial at 6 and 12 hours intervals					
Measurement Model					
Observations; 30 seconds normal points from Eurolas Data Center					
Laser pulse wavelength: 532 nm (Zimmerwald 423 nm)					
Tropospheric refraction: Marini/Murray model (Marini and Murray, 1973)					
Editing criteria:					
$5\sigma \approx \pm 11$ cm for arc (Ajisai), $5\sigma \approx \pm 12$ cm for arc (Starlette, Stella)					
cut-off elevation 10 deg.					
Numerical Integration					
Integration: Cowell's method					
Orbit integration step size: 110 sec for Ajisai and 90 sec for Starlette and Stella					
Arc length: 14 days for Aijsai and 10 days for Starlette and Stella					

3. Determination of SLR station coordinates and velocities

The coordinates and velocities of the following stations were determined: Yarragadee (7090), Greenbelt (7105), Graz (7839) and Herstmonceux (7840). These stations provide the best quality of observation results. Besides these stations provided the largest number of normal points of Ajisai, Starlette and Stella in the mentioned period. The calculated X, Y, Z

geocentric coordinates referenced to the epoch of ITRF2005 were transformed to the topocentric coordinates N, E, Up. These results were related to ITRF2005. In order to determine the movement of station the calculated X, Y, Z geocentric coordinates were referenced to the epoch of each orbital arc. Also in this case the calculated X, Y, Z components were transformed to N, E, Up components. On figures 1-4 the results obtained from Ajisai data were presented in two cases with empirical acceleration coefficients determined every 6 (blue line) and 12 (red line) hours. The left site of each figure shows determined station coordinates in reference to ITRF2005, the right one the movement of each topocentric component for a given epoch of orbital arc. Analogous results were presented for Starlette and Stella on figures 5-8. The gaps on the figures mean the lack or less than 50 number of normal points of the station for the given orbital arc. The RMS of the determined coordinates amount from 14.7 to 20.1 mm for Ajisai and 9.3 to 17.1 mm for Starlette and Stella depending on the number of the empirical acceleration coefficients. In case of Ajisai results the noticable systematical shifts of Up component for each station are visible. Probably they are caused by the inaccurate definite centre of mass correction of Ajisai. This parameter changes within the range of 5 cm (Otsubo et al., 2003).



Figure 1. The topocentric coordinates N, E, Up of the station Yarragadee (7090) obtained from laser observations of Ajisai.



Figure 2. The topocentric coordinates N, E, Up of the station Greenbelt (7105) obtained from laser observations of Ajisai.



Figure 3. The topocentric coordinates N, E, Up of the station Graz (7839) obtained from laser observations of Ajisai.



Figure 4. The topocentric coordinates N, E, Up of the station Herstmonceux (7840) obtained from laser observations of Ajisai.



Figure 5. The topocentric coordinates N, E, Up of the station Yarragadee (7090) obtained from laser observations of Starlette and Stella.



Figure 6. The topocentric coordinates N, E, Up of the station Greenbelt (7105) obtained from laser observations of Starlette and Stella.



Figure 7. The topocentric coordinates N, E, Up of the station Graz (7839) obtained from laser observations of Starlette and Stella.



Figure 8. The topocentric coordinates N, E, Up of the station Herstmonceux (7840) obtained from laser observations of Starlette and Stella.

	V 2D [mm/rok]					
	AJISAI	STARLETTE STELLA	LAGEOS	ITRF2005	NNR NUVEL1A	
Yarragadee	66.5 67.0	65.7 71.5	68.3	68.1	66.6	
Greenbelt	16.1 15.0	16.2 14.7	15.6	15.7	15.4	
Graz	25.3 25.0	26.7 25.8	26.6	26.7	24.8	
Herstmonceux	22.8 23.4	22.7 23.3	23.3	23.4	23.3	

Table 4. The velocities in horizontal plane of the stations determined for Ajisai, Starlette/Stella, LAGEOS, ITRF2005 and geological model NNR-NUVEL1A.

Table 4 shows the velocities of the stations in horizontal plane determined for Ajisai and Starlette and Stella in comparison to the same results obtained for LAGEOS, ITRF2005 and geological model NNR-NUVEL1A (DeMets et al., 1994). In case of LAGEOS the velocity of the station were computed for the period 1999-2004 (Schillak, 2008). For Ajisai and Starlette and Stella two values of the V2D velocity were presented. The upper value is in Table 4 for variant with empirical acceleration coefficients determined every 6 hours. For this solution the differences come from 0.4 to 2.6 mm/year. The bottom is for variant with empirical

acceleration coefficients determined every 12 hours. For this solution the differences don't exceed 1.6 mm/year for Ajisai and 4.9 mm/year for Starlette and Stella. The fastest station is Yarragade moving about 7.0 cm/year in North-East direction. The European stations like Graz and Herstmonceux 2.5 and 2.3 cm/year in Nort-East direction, rspectively. The slowest is Greenbelt moving in North-West direction about 1.5 cm/year.

5. Summary and conclusions

The results presented in this work show, that observations of low satellites such as Ajisai, Starlette and Stella can be used for determination of the SLR station coordinates and velocities. The obtained results permit verification of results derived from the LAGEOS observations are in a good agreement with those determined from LAGEOS observations, for ITRF2005 and geological model NNR-NUVEL1A. The stations' coordinates and velocities were counted for 5-years period. Approximately 0.5% of all normal points of Ajisai were rejected by the orbital program and about 1% for Starlette and Stella. The RMS of fit is on the level 1.61-1.98 cm for Ajisai and 1.15-1.63 cm for Starlette and Stella. The RMS of the calculated coordinates amount from 9.3 to 20.1 mm depending on the number of the empirical acceleration coefficients. The stations are moving in horizontal plane from 1.5 to 7.0 cm/year. These results show, that the laser observations of low satellites could be included to the global solution of the stations' coordinates and velocities.

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